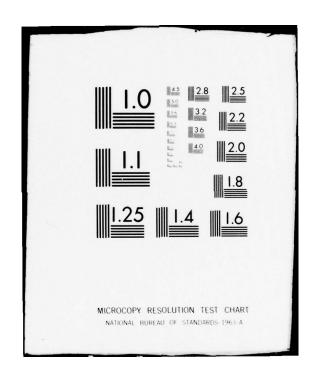
AERONAUTICAL SYSTEMS DIV WRIGHT-PATTERSON AFB OHIO F/G 1/3
A METHOD FOR ADJUSTING MAINTENANCE FORECASTS TO ACCOUNT FOR PLA--ETC(U)
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A METHOD FOR ADJUSTING MAINTENANCE FORECASTS TO ACCOUNT FOR PLANNED AIRCRAFT SORTIE LENGTHS

Lawrence D. Howell, Captain, USAF Engineering Specialties Division Directorate of Equipment Engineering

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DEPUTY FOR ENGINEERING
AERONAUTICAL SYSTEMS DIVISION
AIR FORCE SYSTEMS COMMAND
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LAWRENCE D. HOWELL, Captain, USAF Project Engineer

FOR THE COMMANDER

PAUL E. BECK

Technical Director

Directorate of Equipment Engineering

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Maintenance actions are plotted assists average as	, the failures resulting in

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Then linear regression is used to smooth chance variation and the intercept is related to cyclic or warm-up failure rates while the slope is related to time induced failures.

This technique is used to analyze failure rates for four aircraft types (three military and one civilian). Comparisons are made and it is recommended that this technique be used in forecasting failure rates of developmental military aircraft systems.

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### BACKGROUND

# HIŞTORICAL AIRCRAFT RELIABILITY

Ever since the government began buying large numbers of military aircraft from competing civilian aircraft production corporations the problem of forecasting reliability and maintenance requirements has been an important consideration.

As soon as data became available upon operational aircraft the unscheduled maintenance failure rate was analyzed in terms of maintenance—man-hours-per-flying-hour (MMH) for the aircraft weapon system. Then as new or replacement aircraft were considered, the competing contractors were asked to estimate the MMH of their proposed design. Since the total weapon system cost includes the maintenance costs, the lower the reported MMH the more likely the corporation would be to sell its model. Although companies did perform some demonstration tests (often utilizing highly trained personnel under ideal conditions) the estimates proved to be grossly inaccurate and costs greatly exceeded predictions. The credibility of the contractors' estimates soon waned as more manpower and spare parts were needed than had been made available based upon those estimates.

As a result, spare parts shortages and lack of trained maintenance personnel led to lower mission accomplishment than forecasted and another method of reliability/maintainability forecasting was sought.

As maintenance officers and technicians gained experience in operational aircraft similar to the developmental aircraft, their opinions as to the numbers of men and time required to do each task were sought.

This source tended to be somewhat of a exaggeration since these operational personnel were concerned with having ample people and parts to accomplish the requested missions and thus "looking good." Contractors, predictably, criticized this method as biased and unfair.

In any case both of the previous sources proved to be less than precise and cost forecasts and comparisons continued to be difficult and imprecise. With the advent of computers and the Maintenance Data Collection (MDC) system the use of computer forecasting models became feasible.

The MDC system was developed in the late 1950s as a system of documents or forms to record unscheduled and routine maintenance performed on each aircraft belonging to the Air Force. The other services also developed maintenance collection systems. Under the MDC system, each time maintenance or support is performed on or for an aircraft, the repairmen document the aircraft identification, what tasks were performed, and how many repairmen of each type worked for what time period. This data is then stored on computer tapes for a variety of uses. See Appendix B for a complete description reproduced from AFM 66-1. DEVELOPMENTAL AIRCRAFT MAINTENANCE FORECASTING MODELS

In the mid-sixties the Air Force, in conjunction with the Rand Corporation, developed a series of computer simulation program frameworks (CONCUR, CONVOL, SAMSON II, etc)<sup>3</sup> and later, by contract with CACI, Inc., (LCOM).<sup>4</sup> The Army<sup>5</sup> and Navy<sup>6</sup> also have made use of simulation models to predict maintainability/reliability and mission accomplishment.

The current simulation model used by the Air Force for manpower requirements forecasting is the Logistics Composite Model (LCOM) written in Simscript II.5.7 LCOM is a general aircraft or weapons system model which can be adapted to a particular aircraft or weapon system by incorporating the data concerning that particular aircraft.

The inputs to this model are the daily mission schedules which define: when the aircraft are to fly and for how long; the scheduled maintenance servicing tasks, including the number and time distribution that maintenance repairmen of each specialty are required for each task; and what support resources each task utilizes. In addition the user defines the corrective maintenance networks to include tasks, times, and resources to repair each subsystem when it breaks. Failure clocks defining the frequency with which each subsystem requires corrective maintenance, the initial quantities and resupply times are also user supplied.

With these inputs, along with the initial number of aircraft to be assigned, LCOM simulates the flying operations for the modeled organization for any desired time period. See AFHRL-TR-74-97(II) for a more complete description of LCOM.8 Figure 1 illustrates the simulation technique.

In order to define the failure clocks for the subsystems of developmental aircraft the engineer collects whatever data is available from similar operational aircraft, usually from the MDC system. This data normally comes directly from an air base that operates the comparison aircraft. The AFM 66-1 maintenance data is then processed through a

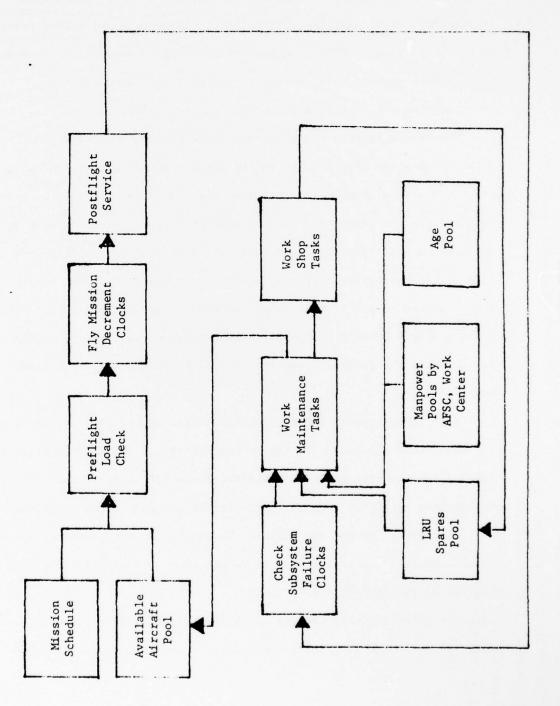


Figure 1. How LCOM Works

series of FORTRAN computer programs written for the CDC 6600 computer to put it in a form that can be used in developing the simulation inputs.

The task times and failure rates are then analyzed by aircraft systems development engineers to make adjustments for differences and improvements in the developmental subsystems. This new data is then used to develop the computer networks of the simulated aircraft.

PROBLEM ENCOUNTERED DURING A COMPARABILITY STUDY

A number of difficulties arise during this process that must be overcome by the simulation design engineer. For instance, data from different bases flying the same aircraft may display different failure rates for the same equipment. Since the LCOM simulation normally bases the forecasted failure distribution upon number of sorties between failures, this parameter is highly sensitive to the air base selected as the comparison location.

One possible solution to this problem is to average the data from several air bases before making the comparability analysis. However, this method does not take into account the possibility that these differences may be due to other than random chance. If, on the other hand, some of the factors causing different failure rates can be measured we can take them into account in future studies and produce a more accurate forecasting model.

One of the factors that possibly affects the failure rate is the average length of the sorties flown. For instance, if the takeoff and climb portion of a sortie produces a failure rate that is different

from the failure rate during the cruise portion of the flight, and perhaps a third failure rate is prevalent during the landing phase, then the average number of sorties between failures is dependent upon the average length of the sorties. This could also be described as the failure rate during warm-up period for the equipment and the steady-state failure rate.

Thus the intent of this study is to determine whether sortie length is a significant factor in making comparability analyses and, if so, what the simulation design engineer can do to account for an expected sortie length for a developmental aircraft that differs from the comparison aircraft.

# PLANNING EXPERIMENTAL DESIGN

## OBJECTIVE

The objective of this study is to provide a technique by which the LCOM simulation engineer can readjust the failure rates of aircraft subsystems to account for a planned difference between sortic length of the developmental aircraft and that of the comparison aircraft.

Since maintenance data is kept by systems, subsystems, and components, we are able to collect and analyze data for each aircraft system. The aircraft parts are tracked by a five-digit work unit code when the first two digits refer to the system, the third digit refers to the subsystem and the last two digits refer to the components. For the purpose of this study, I will group the data by two-digit level. Technical Manual TO-OO-20-2 explains the data collection system in detail. 10

## REQUIREMENTS AND CONSTRAINTS

The requirements for completion of this study are: data from three or more different aircraft that can be associated with sortic length, the use of computer statistical analysis packages, and computer data reduction programs.

The constraints that persist throughout the study include limited time and manpower to do the study. The source of information will be the MDC system and civilian aircraft contractor sources.

Since the prime immediate use of the results is to assist in a modeling developmental cargo airlift aircraft, I have limited the data

collection effort to large cargo, bomber and civil aircraft. However, the techniques described in this report can be used for other models as well.

Several assumptions apply to the planned use of regression to analyze the data. The most important assumption pertains to the warm-up period for each of the systems analyzed in this study. The assumption is that every sortic is of sufficient length to cover the warm-up period and allow the system to reach its steady-state failure rate. Thus, if we discount other effects and assume that cyclic effects will stabilize within the sortic flight time then the average number of maintenance actions per sortic can be related linearly with increasing sortic length.

There is some precedent for using this type of analysis. I would like to point out that Mr. Maurice Shurman from the Boeing Company has demonstrated that field data does support the changing failure rate with time theory. According to this study failure rates seem to stabilize between ten and twenty percent of the time into a sortie for the aircraft with which we are concerned.

Mr. Shurman performs his analysis upon the data grouped by failures within each aircraft type and then develops a common time dependent failure rate equation to predict the failure rate at any time within the sortie. His equation is general and relates to all aircraft and unspecified maintenance requirements.

Kern and Drnas have also studied how operational influences affect reliability. From their report one might gain an insight into some of

the operational factors that may change the failure rates of avionics equipment.  $^{12}$ 

Another assumption is that other factors can be divorced from the effect we are attempting to analyze. Such factors as differences in mission profiles, maintenance concepts (methods and policies), weather (environment), aircraft age, aircrew techniques, utilization rates and any others may also offset failure rates. These factors must be isolated or assumed not to completely hide the effect we are studying.

The third assumption is that the cyclic and flying hour effects can be measured and that the MDC system is accurate enough to show these effects.

#### DATA COLLECTION METHOD

From a commercial airplane company I have obtained maintenance data for a civilian aircraft collected for two different sortie lengths. 13 I have also acquired data from B-52D aircraft from three bases operating in Southeast Asia. 14 These two sets of data were grouped by base.

One group of the civilian planes (Boeing 727s) flew commuter flights only (averaging .566 hours) while the other group flew standard flights (averaging 1.327 hours).

The B-52D aircraft data comes from three bases each flying the same mission profile but requiring different sortic lengths (U-TAPAO - 4 hours, KADENA - 8 hours, ANDERSEN - 11.5 hours).

In addition to these data sources I have obtained from the Air Force Logistics Command (AFLC) 16 computer tapes containing the MDC information for all Air Force C-141A and C-130E aircraft for the period of time June 1976 through May 1977. See Appendix A.

### PLANNED ANALYSIS

The initial analysis plan is to take this data, which describes maintenance performed on aircraft flying the same mission profile but different sortie lengths for several different type aircraft, and then use regression programs to relate the failure rates to flying time and number of sorties. We can think of the total average maintenance actions per sortie as being made up of a constant portion (regression line intercept) which we will call maintenance actions per sortie (MAPS) and a variable portion (regression line slope) which we will call Maintenance Actions Per Flight Hour (MAPFH). See Figure 2.

Some of the causes for these differences may be that, during turn-on, electrical surges cause a short-term failure rate that is not indicative of the steady-state equipment usage failure rate. There is intensive low frequency vibration and shock during taxi and takeoff as well as significant vibration during the flight phases requiring high power settings, such as low altitude and climb-out. During the maximum gross weight take-off conditions there is longer use of maximum power. Also, there is the thermal instability of electronic equipment until the equipment reaches its steady-state temperature.

However, it is believed that after these warm-up or cyclic effects have stabilized, the flight enters the phase of relatively constant failure rates for the cruise portion of the flight.

## CLARIFICATION OF TERMS

Here we are defining a maintenance action to be any unscheduled (not routine) action or repair service performed by one or more

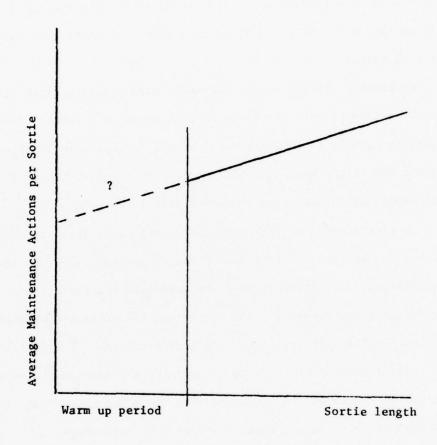


Figure 2. Maintenance Actions (Cyclic and Flying Hour Factors)

maintenance men and documented by a Fcim 349 under the MDC system.

Each action is documented against the proper work unit code and entered into the records regardless of time required to make the repair.

Failure rate and average maintenance actions per sortie are used as parallel terms.

As further clarification, the term "Maintenance Actions Per Sortie (MAPS)" refers only to the failures that occur as a result of the cyclic or warm-up (transient-state) effect and the term "Maintenance Actions Per Flight Hour (MAPFH)" refers only to the failures that are time dependent (steady state) and not related to warm-up.

By plotting the average number of maintenance actions per sortie against the average sortie length for each system, and using regression to smooth the data fluctuations, the estimated maintenance actions per sortie (regression line intercept) and the estimated maintenance actions per flight hour (regression line slope) can be calculated.

This technique can be used to estimate the maintenance actions per sortie (MAPS) and the Maintenance Actions Per Flight Hour (MAPFH) for each of the aircraft types considered in this study.

From this analysis an adjustment factor can then be determined to account for a planned difference in sortic lengths between the developmental and comparison aircraft.

# ANALYSIS AND RESULTS

## DATA REDUCTION TECHNIQUES

The civilian aircraft data is combined by systems to coincide with military designations and is presented in Table 1 along with the B-52D data in terms of average maintenance actions per sortie. This table is computed by dividing the total number of failures for each system, aircraft and sortie length by the total number of sorties flown during the considered time period.

Since the civilian aircraft all flew the same mission profile and the B-52D aircraft all flew the same mission profile, and each source of data is relatively self-consistent as far as weather and maintenance concepts, it is reasonable to use linear regression to test the effect of different sortie lengths upon average numbers of failures for each aircraft type and system.

Since the civilian data yields only two data points per system and the B-52D data only three, I have limited the regression analysis to linear relationships. The results of these regressions and data summary are presented in Tables 2 and 3. Regression line plots appear in Appendix C.

Data for the C-130E and C-141A aircraft was obtained from the MDC input to AFLC and was collected by each aircraft serial number and date. This data was processed through a series of computer programs to provide the total number of maintenance actions in each two-digit work unit code areas for the time periods June 1976 to September 1976 and October 1976 through May 1977. Analysis were made upon each of these

TABLE 1. AVERAGE MAINTENANCE ACTIONS PER SORTIE

System		.566 Hr Civilian 1	1.327 Hr Civilian 2	4 Hr B-52	8 Hr B-52	11.5 Hr B-52
11	Airframe	070.	980.	4.216	7.024	8.602
12	Fuselage	.079	.186	. 592	844.	606.
13	Landing Gear	.041	.058	2.504	2.496	2.772
14	Flt Controls	.007	.015	3.616	2.721	3.289
23	Engines	.045	060.	3.712	7.504	8.522
41	Air Cond	.015	.025	.912	776.	1.403
42	Elec Pwr	.011	.018	1.296	1.848	2.013
77	Lighting	720.	.136	.652	.632	1.909
45	Hvd Pwr	.007	.013	1.396	2.152	2.404
94	Fuel	.003	900.	.868	.624	1.668
47	Oxvgen	.010	.013	.208	.192	.299
51	Instruments	900.	.010	2.932	3.096	3.473
52	Autopilot	700.	900.	.752	. 904	.817
63	UHF Comm	.025	.043	.684	89.	.782
74	Fire Cont	700.	900.	2.48	2.608	5.026
HF + I	UHF + Interphone	.025	.043	1.152	1.296	1.392

LINEAR REGRESSION RESULTS FOR BOEING 727 AND B-52D AIRCRAFT (Data Grouped by Base) TABLE 2.

		B-9	B-52D	Boein	Boeing 727	
System	8	MAPS*	MAPFH**	MAPS	MAPFH	
						1
11	Airframe	2.011	.586	.0057	7090.	
12	Fuselage Compartments	.333	070.	0	.1406	
13	Landing Gear	2.318	.035	.0284	.0223	
14	Flight Controls***	1.419	.163	.001	.0105	
23	Engines	1.499	679.	.0115	.0591	
41	Air Cond	. 584	790.	9200.	.0131	
42	Electrical Power Supply	.963	760.	.0058	.0092	
77	Lighting Systems	.217	.164	.0331	.0775	
45	Hydraulic & Pneumatic Sys	.921	.136	.0025	6200.	
94	Fuel Systems	. 249	.103	8000.	.0039	
47	Oxygen Supply	.141	.0118	.0078	.0039	
51	Instruments	2.608	.0714	.0030	.0053	
52	Autopilot	.7513	.0093	.0025	.0026	
63	UHF Comm	.6135	.0131	.0116	.0237	
14	Fire Cont	.770	.332	.0025	.0026	
UHF +	UHF + Interphone	1.029	.032	.0116	.0237	

\* MAPS = maintenance actions per sortie or cycle \*\* MAPFH = maintenance actions per flight hour \*\*\* Using two sortie lengths from B-52 data (U-Tapao data excluded due to different reporting method)

TABLE 3. AIRCRAFT DATA SUMMARY

	Civilian 1	Civilian 1 Civilian 2	B-52	B-52	B-52
Average Sortie Length	. 566	1.327	7	80	11.5
Total Flight Hours	9539	50294	18,013	15,286	34,918
Total Sorties	16830	38062	4503	11911	3036
Number of Aircraft	9	22	35	20	42
Average Sorties per Aircraft per Month	233.8	144.2	21.4	18.9	12.0
Flight Hours per Aircraft per Month	132.5	190.5	85.8	127.4	138.6

sets of data and upon the one year data combination for each aircraft.

The method of data reduction to obtain the total maintenance actions

for each system is presented in Appendix A.

The data for the C-141A aircraft came from six Air Force bases:

- 1. McGuire AFB NJ with 47 aircraft.
- 2. Travis AFB CA with 41 aircraft.
- 3. Charleston AFB SC with 99 aircraft.
- 4. McCord AFB WA with 40 aircraft.
- 5. Norton AFB CA with 50 aircraft.
- 6. Altus AFB OK with 16 aircraft.

The data for the C-130E aircraft came from ten bases:

- 1. Pope AFB NC with 38 aircraft.
- 2. Elmendorf AFB Alaska with 15 aircraft.
- 3. Little Rock AFB AR with 64 aircraft.
- 4. Mildenhall Air Base England with 16 aircraft.
- 5. Clark AFB Philippine Islands with 18 aircraft.
- 6. Yokota AB Japan with 18 aircraft.
- 7. Andrews AFB MD with 1 aircraft.
- 8. Elmendorf AFB Alaska with 10 aircraft.
- 9. McCord AFB WA with 21 aircraft.
- 10. Langley AFB VA with 6 aircraft.

The average number of maintenance actions per sortie for the C-141A and C-130E aircraft grouped by base are presented in Tables 4 and 5 respectively.

TABLE 4. C-130E DATA GROUPED BY BASE (Jun 76 - May 77) Average Maintenance Actions Per Sorties and Standard Deviations

Mean = Average number of maintenance actions per sortie S.D. = Standard deviation of maintenanct actions per sortie

Ba	Base	Pc 38	Pope 38 acft	Rhein 15	Rhein Mein 15 acft	Litt 64	Little Rock 64 acft	Men 16	Mendehall 16 acft
Sy	System	Mean	S.D.	Mean	S.D.	Mean	S.D.	Mean	S.D.
1					0		0.50		17.
	Il Airframe	479.	275.	. 285	/87.	.400	617.	004.	6/1.
1	12 Fuselage Compartments	.212	.158	.231	.074	.228	.107	.213	760.
	13 Landing Gear	.191	860.	.189	.101	.186	.108	.142	.056
	14 Flight Controls	.146	.100	.125	.048	.093	.043	.081	.038
	22 Engine	.680	.241	.697	.255	.546	.277	667.	.236
	24 Aux Power Plant	.114	.053	980.	.028	.065	.033	090.	770.
1	41 Air Cond	.197	.100	.209	.122	.140	.067	.129	.055
	42 Electrical Power Supply	.132	.056	.144	.067	.081	.045	080.	.021
	Lighting Systems	.151	.078	.136	.083	.137	.077	.111	.070
		.236	.132	. 209	080	.147	.093	.126	.042
	46 Fuel Systems	.320	.155	.310	.118	300	.190	.235	790.
		.048	.037	970.	.031	040.	.022	.041	.026
	49 Misc Utilities	.117	.059	.115	090.	.062	.034	.051	.028
	51 Instruments	.129	.050	.125	670.	.082	.035	.067	.029
	52 Autopilot	.093	070.	.080	.038	.064	.039	.063	.034
	61 Communications	.213	990.	. 209	.134	.183	.077	.165	090.
	65 IFF	.021	.019	.027	.028	.019	.021	.013	.013
	71 Radio Navigation	.143	.058	.141	920.	.164	690.	.147	890.
	72 Radar Navigation	.350	.105	.369	.202	. 269	.114	.224	.056
Av	Average Hrs Per Flight	2.647	.185	2.788	.934	3.538	686.	3.006	.393
To	Total Number of Sorties/Month	230.368	54.268	227.667	61.351	190.953	65.223	215.813	44.619
Al	All Maintenance	4.041	1.537	3.907	1.503	3.131	1.210	2.850	.953
Av	Average Hrs Per Month	51.442	9.859	51.639	8.333	53.618	7.862	55.417	6.775
Av	Average Sorties Per Month	19.574	4.176	19.730	5.075	16.439	5.368	18.712	3.319

TABLE 4. C130E DATA GROUPED BY BASE (Cont'd)
(Jun 76 - May 77)
Average Maintenance Actions Per Sorties and Standard Deviations

Mean = Average number of maintenance actions per sortie S.D. = Standard deviation of maintenance actions per sortie

Base		C1 18	Clark 18 acft	Yol 18	Yokota 18 acft	E1me	Elmendorf 10 acft	Mc( 21	McCord 21 acft	Lar 6	Langley 6 acft
System	ш	Mean	S.D.	Mean	S.D.	Mean	S.D.	Mean	S.D.	Mean	S.D.
==	Airframe	.618	.182	.369	.196	.379	.129	.436	711.	.752	.460
12	Fuselage Comp.	.393	.155	.138	.121	.265	.097	.221	920.	.305	.178
13	Landing Gear	.203	740.	.178	.104	.133	.039	.264	.084	.209	.174
14	Flight Controls	.159	.055	.119	.091	.114	.052	.111	770.	.159	960.
22	_	.731	.140	.384	.173	.419	.128	.501	.148	1.289	. 508
1 24	•	.093	.036	.078	.054	.083	.032	680.	.038	.136	.089
41	Air Cond.	.200	.074	.133	.075	.122	.037	.193	.057	.263	.223
42	Elect. Power Supply	.077	.025	.059	670	.085	.025	.058	.030	.149	
77	Lighting Systems	.142	090.	.108	.122	.194	960.	. 206	090.	.179	.124
45	Hyd. & Pneumatic Sys	.161	.034	.174	.093	960.	670.	.156	890.	.259	.158
94	Fuel Systems	.172	.057	.187	.092	.147	.072	.360	.093	.314	. 200
47	Oxygen Supply	.072	.034	.043	.029	.034	.014	.052	.021	.062	.050
65	Misc Utilities	920.	.025	.045	.025	.057	.024	.065	.029	.169	.136
51	Instruments	.101	.035	.065	.042	.059	.021	.093	.038	.123	.059
52	Autopilot	.101	.051	.038	.022	.050	.023	.095	970.	.057	.025
61	Communications	.189	.052	.139	.061	.155	.084	.246	.062	.162	.110
65	IFF	.027	.039	.017	.012	.018	.013	.031	.025	.037	.025
71	Radio Navigation	.136	.041	.127	.070	.111	970.	.171	.063	.109	.055
72	Radar Navigation	.311	.091	.202	860.	.199	.076	.312	.072	.170	.143
Avera	Average Hrs Per Flight	2.837	.374	2.740	.202	2.013	.546	3.193	.383	2.453	.518
Total	-c	214.444	47.737	218.222	69.798	282.400	55.925	181.238	37.667	218.000	69.587
All M		3.863	. 804	2.534	1.230	2.661	.611	3.570	.693	4.782	2.765
Ave.	nth	51.295	7.460	52.869	11.521	48.420	7.265	50.420	7.542	45.325	7.270
Ave.	Sorties Per Month	18.382	3.564	19.471	7.896	25.093	5.095	15.937	2.762	19.233	4.830

TABLE 5. C-141A DATA GROUPED BY BASE (Jun 76 - May 77) Average Maintenance Actions Per Sorties

System		McGuire 47 acft	Travis 41 acft	Charleston 49 acft	McCord 40 acft	Norton 50 acft	Altus 16 acft
	Airframe	0.632	0.609	0.527	0.363	0.438	1.890
1.7	Fuselage Compartments	0.232	0.159	0.169	0.116	0.130	0.615
13	Landing Gear	0.375	0.307	0.273	0.216	0.221	1.020
17	Flight Controls	0.258	0.284	0.237	0.183	0.176	1.176
2	Engine	0.599	0.526	0.440	0.377	0.441	1.600
- †	Aux Power Plant	0.083	0.059	0.050	0.043	970.0	0.157
-	Air Cond.	0.128	0.112	0.100	0.079	0.069	0.189
75	Electrical Power Supply	0.074	0.078	0.076	0.045	970.0	0.136
20		0.347	0.183	0.224	0.144	0.159	0.483
45	Hydraulic & Pneumatic Sys	0.185	0.221	0.164	0.088	0.089	0.305
9+	Fuel Systems	0.126	0.119	0.129	0.130	0.108	0.206
7.7	Oxygen Supply	0.084	0.050	0.065	0.052	0.039	0.118
5,1	Misc Utilities	0.129	0.104	0.077	0.063	0.079	0.185
51	Instruments	0.124	0.135	0.117	0.101	0.098	0.272
52	Autopilot	0.116	0.112	960.0	0.109	0.088	0.187
61	Communications	0.213	0.225	0.194	0.213	0.168	0.353
65	IFF	0.020	0.015	0.017	0.015	0.010	0.020
7.1	Radio Navigation	0.141	0.134	0.128	0.144	0.110	0.255
7.2	Radar Navigation	0.216	0.214	0.181	0.167	0.134	0.333
Averag	Average Hrs Per Flight	3.846	3.856	3.455	3.319	3.706	3.608
Number	Number of Sorties Per Month	295.532	278.220	331.224	326.450	307.900	174.563
Average	Hrs Der Month	671 79	55, 171	677.85	38.775	24.640	112.000
Averag	Average Sorties Per Month	19.702	13.366	16.388	13.750	5.900	35.500
Averag	Average Hrs Per Month Average Sorties Per Month	64.149	55.171	58.449	38.775	24	24.640

For the C-141A and C-130E, since the data was available by aircraft serial number and the aircraft at each base do not fly the same mission lengths as did the B-52D and Boeing 727, each aircraft is considered separate. Thus we have 244 data points for the C-141A and 215 data points for the C-130E aircraft. See Appendix D for C-130E and C-141A data.

# HANDLING OF MISSION AND ABNORMAL DATA

Data for some of the C-141A and C-130E aircraft was incomplete and so these aircraft are eliminated from further analysis. These missing or incomplete data stem from either undocumented or inaccurate data collection at base level, keypunch or computer tape production errors, or aircraft being grounded for long periods of time. In any case, it is assumed that due to the volume of data received intact, the analysis will not be greatly biased by the elimination of incomplete data from the analysis.

Furthermore, in an attempt to use data from aircraft which fly similar mission profiles I have removed the data from bases which fly only training missions since their usage of aircraft create different failure rates than those that fly airlift missions.

Since these airlift aircraft (C-141A and C-130E) fly global missions and have repairs made at many locations, each of which uses the same Air Force policies and maintenance regulations, the effects of weather, climate, and maintenance locational idiosyncrasies are assumed to be in balance.

TABLE 6. LINEAR REGRESSION RESULTS FOR C-130E AIRCRAFT (Jun 76 - Sep 76)
Average Maintenance Actions per Cycle (Intercept) and Flt Hr (Slope)

(54)		c.v.
016) 111 1		$\mathbb{R}^2$
irept/ and it	stimate ntion coefficient	(S.E.E.)
רארוב וזוורם	S.E.E. = Standard error of estimate C.V. = Coefficient of variation $R^2$ = Linear correlation coefficient	Flt Hr (S.E.E.)
actions per	. = Standard = Coeffici = Linear c	Cycle (S.E.E.)
average maintenance actions per cycle (intercept) and fit in (stope)	S.E.E. C.V. R2	Cycle
		System

	.7	0	.1	8	. 2	.5	9.	. 2	7.	6	6	9.	. 2	8	٣.	2	6	9	9
-	95.	114.0	91.	103	66	109.	102.	112.	105.	87.	84.	118.	118.	92.	120.	74.	225.	83.	74.
-	.024	.001	.024	.011	900.	.001	800.	.011	.001	.033	.021	.001	.010	. 003	000.	.048	.031	.024	.052
	(.028)	(.017)	(.010)	(.007)	(.036)	(.005)	(.011)	(900.)	(000)	(*008)	(.015)	(.003)	(002)	(900')	(:005)	(000)	(:003)	(.007)	(.014)
	790.	008	.022	.011	.041	002	.014	600.	005	.022	.030	.002	.007	.005	000.	.030	800.	.016	.047
	(:095)	(.055)	(.033)	(.025)	(.122)	(.018)	(.037)	(.020)	(.031)	(.027)	(670')	(.011)	(.017)	(.021)	(.017)	(.031)	(.011)	(.024)	(.047)
	. 206	.227	620.	790.	.375	920.	.105	.045	.136	.059	.141	.035	.037	.077	090.	.071	007	.067	.109
	Airframe	Fuselage Compartments	Landing Gear	Flight Controls	Engine	Aux Power Plant	Air Cond	Electrical Power Supply	stems	Hydraulic & Pneumatic Sys	Fuel Systems	Oxygen Supply	Misc Utilities	Instruments	Autopilot	Communications	IFF	Radio Navigation	Radar Navigation
	11	12	13	14	22	1	1,1	75	77	45	76	14	6+	51	52	61	65	71	77

TABLE 7. REGRESSION RESULTS FOR C-130E AIRCRAFT (Oct 76 - May 77)
Average Maintenance Actions per Cycle (Intercept) and Flt Hr (Slope)

		Average Ma	S.E.E.	= Standard = Coefficie	Standard error of estimate Coefficient of variation	Average maintenance actions per Cycle (intercept) and fit in (stope)  S.E.E. = Standard error of estimate  C.V. = Coefficient of variation		
S	System	F	Cycle	, )	Flt Hr	(S.E.E.)	R2	c.v.
1	=	Airframe	600	(360.)	.206	(,031)	.170	63.6
	17	Fuselage Compartments	.061	(.039)	920.	(.012)	.148	53.6
	13	( )	057	(.029)	.095	(000)	.330	50.5
	14	Flight Controls	790.	(.024)	.025	(*008)	870.	67.7
	22	Engine	197	(.085)	.162	(.027)	.144	49.2
	24	Aux Power Plant	.062	(.019)	.013	(900')	.021	73.5
,	41		.034	(.030)	.054	(600.)	.132	0.09
23	42	Electrical Power Supply	.033	(.021)	.027	(.007)	.073	71.5
	77	Systems	.019	(.024)	.050	(008)	.164	57.0
	45	ပ	.071	(.035)	970.	(,011)	.073	66.5
	94	a	019	(.045)	.112	(.015)	.221	56.3
	47	Oxvgen Supply	013	(.010)	.023	(,003)	.184	74.2
	64	Misc Utilities	940.	(.019)	.016	(900')	.032	80.3
	51	Instruments	.021	(,014)	.028	(:002)	.150	24.0
	52	Autonilot	.035	(.013)	.016	(,004)	.070	60.2
	61	Communications	007	(.022)	.075	(.007)	.355	39.4
	65		600.	(*008)	.005	(:003)	.018	133.7
	71	Radio Navigation	.012	(.018)	.052	(900')	.281	42.1
	72	Radar Navigation	.012	(.032)	.102	(.010)	.328	39.1

TABLE 8. LINEAR REGRESSION RESULTS FOR C130E AIRCRAFT

Jun 76 - May 77

Average Maintenance Action Per Cycle (Intercept) and Flight Hour (Slope)

S.E.E. = Standard Error of Estimate C.V. = Coefficient of Variation R<sup>2</sup> = Linear Correlation Coefficient

S	System	E	Cycle	(S.E.E.)	Flt Hr	(S.E.E.)	R <sup>2</sup>	C.V.
1	:	A 4 - 5 - 1	196	(890)	UBU	( 022)	061	6.87
	11	Alrirame	107.	(000.)		(110.)	100.	
	12	Fuselage Compartments	.165	(.036)	.027	(.012)	.025	53.1
	13	Landing Gear	.002	(.024)	.063	(*008)	.243	42.4
	14	Flight Controls	.091	(,019)	600.	(900.)	.010	58.6
	22	Engine	.289	(920)	.101	(,024)	920.	7.97
	24	Aux Power Plant	.083	(.013)	000.	(,004)	000.	55.7
21	41		.092	(.025)	.025	(*008)	970.	52.9
	42	Electrical Power Supply	.071	(.015)	.007	(:002)	.010	59.8
	77	stems	680.	(.023)	.019	(.007)	.031	57.1
	45		060.	(.027)	.027	(000)	.045	57.5
	94		.050	(.039)	920.	(.013)	.148	51.2
	47	Oxygen Supply	.015	(*00.)	.010	(:003)	.075	61.4
	64	Misc Utilities	.059	(.015)	900.	(:002)	600.	68.5
	51	Instruments	640.	(.013)	.015	(,004)	.058	50.4
	52	Autopilot	.047	(.012)	600.	(,004)	.026	57.3
	61	Communications	.048	(.021)	.048	(.007)	.194	38.0
	65	IFF	.014	(900')	.003	(.002)	.007	105.9
	71	Radio Navigation	.021	(.015)	.042	(:002)	.262	37.6
	72	Radar Navigation	.081	(.031)	.067	(.010)	.187	38.7
14	1 W	All Maintenance	1.61	(.371)	. 605	(.120)	.111	37.4

TABLE 9. REGRESSION RESULTS FOR C-141A AIRCRAFT (Jun 76 - Sep 76)
Average Maintenance Actions Per Cycle (Intercept) and Flt Hr (Slope)

ate		cient
estima	variation	coeffic
of	Var	ion
Error	ent of	correlation
Standard	Coefficien	Linear c
H		-
S.E.E.	c.v.	R <sup>2</sup>

C.V.

 $\mathbb{R}^2$ 

(S.E.E.)

Flt Hr

Cycle (S.E.E.)

System

104.3	107.7	7.76	116.6	81.4	103.7	146.2	87.0	75.4	82.7	6.99	186.2	70.4	63.5	63.7	51.1	181.0	8.09	54.5
.001	.004	.003	· 004	.002	. 005	.001	.011	000.	.011	.001	.001	.032	000.	.003	000.	900.	.001	.021
(,074)	(.022)	(.033)	(.036)	(.050)	(.007)	(.016)	(.007)	(.020)	(.014)	(*008)	(.013)	(*008)	(*00.)	(.007)	(.012)	(.003)	(000)	(.012)
034	022	030	037	.036	800.	.010	.012	001	.023	.005	007	.021	.001	900.	.001	700.	005	.027
(.283)	(.084)	(.127)	(.138)	(.192)	(.027)	(.063)	(.028)	(.075)	(.053)	(.032)	(.050)	(.029)	(.031)	(.027)	(.045)	(.012)	(.034)	(.045)
.750	261	.423	.411	.405	.030	.062	.030	.230	090.	.092	980.	.014	.109	.073	. 206	.001	.145	.088
Airframe	Fuselage Compartments	Landing Gear	Flight Controls	Engine	Aux Power Plant	Air Cond	Electrical Power Supply	Lighting Systems	Hydraulic & Pneumatic Sys	Fuel Systems	Oxygen Supply	Misc Utilities	Instruments	Autopilot	Communications	IFF	Radio Navigation	Radar Navigation
11	12	13	14	23	24	41	42	77	45	94	47	67	51	52	61	65	71	72

TABLE 10. LINEAR REGRESSION RESULTS FOR C-141A AIRCRAFT (Oct 76 - May 77)
Average Maintenance Actions per Cycle (Intercept) and Flt Hr (Slope)

		C.V.
		$\mathbb{R}^2$
7.4	timate ttion coefficient	(S.E.E.)
	S.E.E. = Standard error of estimate C.V. = Coefficient of variation R <sup>2</sup> = Linear correlation coefficient	Cycle (S.E.E.) Flt Hr
		(S.E.E.)
ייינייניינייניינייניינייניינייניינייניי	S.E.E. C.V. R2	Cycle
W. C. C.		
		System

.3.					0	
	321	(.321)	.087	(680)	.004	9.98
Compartments .0	050	(106)	770.	(.029)	600.	86.7
	133	(.147)	090.	(.041)	600.	72.2
.10	109	(.219)	.057	(.061)	.004	118.4
.1	182	(.261)	.112	(.072)	.010	76.2
١. ٥	200	(.033)	.021	(.000)	.021	83.5
0.	010	(.035)	.029	(.010)	.036	52.1
Supply .0.	640	(.024)	900.	(.007)	.003	59.1
	.010	(.093)	.065	(.026)	.025	9.59
Hydraulic & Pneumatic Sys0.	024	(990')	.055	(.018)	.036	65.0
	880	(.040)	.015	(.011)	.007	47.8
0.	038	(.023)	800.	(900.)	900.	60.2
0	003	(.035)	.029	(.010)	.037	58.0
ŏ.	190	(.039)	.021	(.011)	.015	49.3
0.	990	(.032)	.015	(600.)	.011	45.5
.10	102	(.048)	.033	(.013)	.025	37.2
0.	003	(000)	700.	(.003)	.010	91.7
0.	054	(.039)	.026	(.011)	.023	45.2
0.	690	(.051)	.036	(.014)	.026	43.7

TABLE 11. LINEAR REGRESSION RESULTS FOR C-141A AIRCRAFT (Jun 76 - May 77)
Average Maintenance Actions Per Cycle (Intercept) and Flight Hour (Slope)

S.E.E. = Standard Error of Estimate C.V. = Coefficient of variation R<sup>2</sup> = Linear Correlation Coefficient

System		Cycle	(S.E.E.)	FIt Hr	(S.E.E.)	R <sup>2</sup>	c.v.
1	Airframo	087	(,121)	.166	(.033)	.100	34.2
12	Fuselage Compartments	040	(,048)	.055	(.013)	.073	43.1
13	Landing Gear	001	(.073)	.077	(.020)	.063	37.8
14	Flight Controls	005	(,063)	790.	(.017)	.057	40.3
23	Engine	107	(,106)	.160	(.029)	.120	32.3
24	Aux Power Plant	033	(,021)	.025	(900')	.073	55.2
41	Air Cond	019	(,026)	.032	(.007)	.081	39.1
42	Electrical Power Supply	.012	(,018)	.014	(.005)	.036	41.4
77	Lighting Systems	079	(,075)	080	(.020)	790.	8.05
45	Hydraulic & Pneumatic Sys	113	(.050)	.072	(.014)	.110	48.7
94	Fuel Systems	.093	(,031)	800.	(*008)	· 004	36.6
47	Oxygen Supply	.001	(.019)	.016	(.005)	.038	48.6
67	Misc Utilities	084	(.025)	.048	(.007)	.178	40.3
51	Instruments	.021	(.023)	.026	(900.)	690.	29.4
52	Autopilot	.048	(.026)	.015	(.007)	.021	35.8
61	Communications	.110	(.037)	.025	(.010)	.027	79.97
65	IFF	.004	(600.)	.003	(.002)	800.	80.8
71	Radio Navigation	.063	(.030)	.018	(.008)	.021	34.0
72	Radar Navigation	.035	(.041)	070.	(.011)	.055	32.5
All Mai	All Maintenance	201	(.619)	.919	(.169)	.116	28.6

TABLE 12. LINEAR REGRESSION RESULTS FOR C-130E AIRCRAFT

		IABLE		12. LINEAR REGRESSION RESULIS FOR C-130E AIRCRAFT (Data Grouped by Base - Jun 76 - Sep 76)	on resulis r ase - Jun 76	ok C-130E AL	NOVAF 1	•
Sy	System		Cycle	(S.E.E.)	Flt Hr	(S.E.E.)	R <sup>2</sup>	c.v.
	:		220	( 322)	063	( 097)	040	53.3
	11		057.	(1775)	000	(100.)		
	12	Fuselage Compartments	.271	(.184)	018	(550.)	.011	0.49
	13	Landing Gear	028	(.151)	.062	(.045)	.157	63.8
	14	C	.080	(*084)	800.	(.025)	.011	57.3
	22		.544	(.547)	.013	(.164)	.001	68.3
	24	Aux Power Plant	.124	(.053)	016	(,016)	.092	53.0
	41		006	(.170)	090.	(.051)	.120	66.1
	75	Electrical Power Supply	.061	(980)	.047	(,026)	.250	68.2
	77	Lighting Systems	.147	(.121)	900	(.036)	.003	69.3
	45	C	.084	(.103)	.016	(.031)	.026	55.6
	94	sms	017	(.249)	.088	(.075)	.123	67.5
	47	Oxygen Supply	.063	(.030)	~.007	(600.)	.063	8.99
		Misc Utilities	.052	(.055)	.003	(.017)	.004	63.6
20		Instruments	.074	(.080)	900.	(.024)	900.	63.4
		Autopilot	.082	(.064)	009	(.019)	.020	86.9
	61	Communications	980.	(.122)	.026	(.037)	.047	52.6
	65	IFF	017	(.025)	.013	(.007)	.234	72.6
	71	Radio Navigation	.112	(,077)	000.	(.023)	000	50.4
	72	Radar Navigation	025	(.215)	.092	(,064)	.168	57.7

	TABL	E 13. LIN (Data	TABLE 13. LINEAR REGRESSION RESULTS FOR C-130E AIRCRAFT (Data Grouped by Base - Oct 76 - May 77)	ON RESULTS F ase - Oct 76	50R C-130E AL 5 - May 77)	RCRAFT	
stem		Cycle	(S.E.E.)	Flt Hr	(S.E.E.)	R <sup>2</sup>	c.v.
1 1	Airframe	289	(.667)	.360	(,226)	.202	51.5
12	Fuselage Compartments	.211	(.149)	.035	(.050)	.045	27.9
13	Landing Gear	.035	(,118)	690.	(040)	.232	29.0
14	Flight Controls	.024	(.107)	.048	(.036)	.151	37.8
22	Engine	.601	(.912)	.085	(.309)	800.	62.7
24	Aux Power Plant	.007	(.311)	640.	(.105)	.022	120.4
41	Air Cond	081	(.242)	.113	(.082)	.159	57.1
42	Electrical Power Supply	.045	(.101)	.028	(.034)	.062	8.97
77	Lighting Systems	.030	(,208)	090.	(020)	.067	9.69
45	Hydraulic & Pheumatic Sys	.053	(.272)	.067	(.092)	.050	64.2
95		037	(.235)	.120	(080)	.184	0.44
47	Oxygen Supply	018	(.113)	.033	(.038)	.070	83.5
64	Misc Utilities	.112	(.113)	001	(.038)	000.	9.09
51	Instruments	.085	(090)	010	(.020)	.022	30.8
52	Autopilot	.061	(.030)	.007	(.010)	.041	21.5
61	Communications	069	(.212)	.110	(.072)	.189	9.67
65	IFF	.031	(.020)	003	(.007)	.015	50.9
71	Radio Navigation	.055	(.068)	.037	(.023)	. 203	24.5
72	Radar Navigation	.077	(.124)	.084	(.042)	.286	22.5

		TABLI	TABLE 14. LI (Data	. LINEAR REGRESSI (Data Grouped by Ba	SSION RESULTS Base - Jun 76	LINEAR REGRESSION RESULTS FOR C-130E AIRCRAFT a Grouped by Base - Jun 76 - May 77)	IRCRAFT	
S	System		Cycle	(S.E.E.)	Flt Hr	(S.E.E.)	R <sup>2</sup>	c.v.
1	=	Airframo	. 584	(,256)	023	(.087)	600.	25.4
	12	Fuselage Compartments	.273	(.141)	006	(.048)	.002	28.2
	13	Landing Gear	960	(.122)	.106	(.042)	.447	29.8
	14		.170	(.051)	016	(.017)	.095	21.2
	22		.747	(.539)	032	(.184)	,004	42.4
	24	Aux Power Plant	.161	(.041)	026	(.014)	.300	23.9
	41	Air Cond	.139	(.101)	.015	(.034)	.023	28.3
	42	Electrical Power Supply	.119	(.072)	007	(.025)	600.	37.3
	77		. 200	(990)	017	(.022)	.065	22.2
	45	-	.277	(.110)	038	(.038)	.116	34.2
	97	ms	.080	(.135)	.063	(970')	.190	26.4
20	47	Oxygen Supply	.072	(.028)	600	(600.)	.105	30.8
		Misc Utilities	.176	(.077)	033	(.026)	.167	7.67
		Instruments	168	(.157)	.100	(.053)	.304	66.2
		Autopilot	.005	(.043)	.024	(.015)	.253	29.4
		Communications	011	(.084)	.073	(.029)	777.	21.8
	65		014	(.028)	.014	(600)	.221	51.8
	71	Radio Navigation	.044	(.024)	.033	(008)	.682	8.7
	72	Radar Navigation	950.	(.133)	.077	(.045)	.262	24.7

TABLE 15. LINEAR REGRESSION RESULTS FOR C-141A AIRCRAFT (Data Grouped by Base - Jun 76 - Sep 76)

S	System	ч	Cycle	(S.E.E.)	Flt Hr	(S.E.E.)	R <sup>2</sup>	c.v.
	11	Airframe	.925	(3.519)	043	(.935)	.001	86.2
	12	Fuselage Compartments	.481	(1.113)	068	(.296)	.013	92.7
	13	Landing Gear	748.	(2.056)	121	(.546)	.012	98.7
	14	$\circ$	.825	(2.105)	124	(.559)	.012	109.8
	23	Engine	.335	(2.460)	.081	(.653)	.004	72.2
	54	Aux Power Plant	026	(.198)	.025	(.052)	.052	26.0
	41	Air Cond	012	(.270)	.032	(.072)	970.	47.2
	42	Electrical Power Supply	700.	(,224)	.020	(.059)	.028	51.9
	77	Lighting Systems	.228	(.828)	.007	(.220)	000.	61.3
	45	Hydraulic & Pneumatic Sys	680.	(.529)	.019	(.141)	.005	6.09
	94	Fuel Systems	.214	(.253)	025	(.067)	.033	39.1
31	47	Oxygen Supply	.167	(.369)	024	(*00.)	.015	91.6
	65	Misc Utilities	065	(.285)	.045	(920)	.081	51.4
	51	Instruments	.157	(.360)	008	(960.)	.002	52.9
	52	Autopilot	.136	(*188)	009	(.050)	800.	34.5
	61	Communications	.350	(.369)	036	(860.)	.032	32.0
	65	IFF	023	(.036)	.010	(.010)	.216	45.0
	71	Radio Navigation	.384	(.343)	064	(.091)	.111	45.2
	72	Radar Navigation	.142	(.431)	.017	(.115)	.005	39.3

TABLE 16. LINEAR RECRESSION RESULTS FOR C-141A AIRCRAFT (Data Grouped by Base - Oct 76 - May 77)

S	System	п	Cycle	(S.E.E.)	Flt Hr	(S.E.E.)	R <sup>2</sup>	c.v.	
22	111 113 113 113 113 113 113 113 113 113	Airframe Fuselage Compartments Landing Gear Flight Controls Flight Controls Bux Power Plant Air Cond Lighting Systems Hydraulic & Pneumatic Sys Fuel Systems Oxygen Supply Misc Utilities Instruments Autopilot Communications IFF Radio Navigation Radar Navigation	470 225 354 115 115 602 585 585 293 293 206 039	(6.825) (2.271) (3.330) (4.856) (5.860) (.586) (.486) (.380) (1.435) (.387) (.387) (.463) (.768) (.768) (.768) (.768) (.768) (.768) (.768)	.350 .135 .217 .079 .079 .072 .019 .243 .216 .000 .034 .032 .032	(1.905) (.634) (.929) (1.355) (1.636) (.164) (.136) (.106) (.107) (.108) (.126) (.129) (.129) (.129) (.129) (.129) (.129) (.129)	. 008 . 0111 . 0113 . 0013 . 002 . 008 . 008 . 006 . 006 . 005 . 005 . 005 . 005	89.4 80.9 80.9 80.9 84.3 74.3 74.3 74.3 74.3 74.3 74.3 74.3 7	

TABLE 17. LINEAR REGRESSION RESULTS FOR C-141A AIRCRAFT (Data Grouped by Base - Jun 76 - May 77)

			(Data	(Data Grouped by Base -	Jun	76 - May 77)		
S	System		Cycle	(S.E.E.)	Flt Hr	(S.E.E.)	R <sup>2</sup>	c.v.
1	11	Airframe	838	(.614)	.372	(.169)	.619	15.8
	12	Fuselage Compartments	235	(.321)	.109	(.088)	.338	26.2
	13	Landing Gear	400	(.419)	.187	(.115)	.467	19.8
	14	Flight Controls	207	(.326)	.119	(.089)	.373	18.7
	23	Engine	660	(.376)	.312	(.103)	.753	10.4
	24	Aux Power Plant	114	(.100)	.047	(.028)	.488	23.7
		Air Cond	108	(.173)	.057	(.048)	.321	23.4
	42	Electrical Power Supply	062	(.130)	.035	(.036)	.239	26.8
		Lighting Systems	383	(.627)	.164	(.172)	.231	39.1
		Hydraulic & Pneumatic Sys	415	(.402)	.155	(.110)	.397	35.4
		Fuel Systems	.198	(.067)	021	(.018)	.299	7.3
33	47	Oxygen Supply	.012	(.148)	.013	(.041)	.031	33.6
	64	Misc Utilities	244	(.121)	.092	(.033)	.720	17.6
	51	Instruments	029	(.108)	070	(.030)	.372	12.4
	52	Autopilot	.053	(.100)	.014	(.027)	.083	12.6
	61	Communications	.160	(.191)	.012	(.052)	.017	12.4
	65	IFF	.012	(.030)	.001	(.008)	900.	25.6
	71	Radio Navigation	.170	(.116)	011	(.032)	.036	11.7
	72	Radar Navigation	069	(.263)	690.	(.072)	.234	19.0

TABLE 18. CORRELATIONS FOR C130E AIRCRAFT Jun 76 - May 77

	יסיד קקושי	Jun 76 - May 77	
System	E	Average Sortie Length	Total Number of Sorties
=	Airframe	. 247	517
12	Fuselage Compartments	.158	403
13	Landing Gear	. 493	769
14	Flight Controls	.100	437
22	Engine	.276	468
24	Aux Power Plant	,00°	345
41	Air Cond	.216	500
42	Electrical Power Supply	.100	359
77	Lighting Systems	.177	397
45	Hydraulic & Penumatic Sys	.211	431
95	Fuel Systems	.385	576
47	Oxygen Supply	.279	491
67	Misc Utilities	.093	345
51	Instruments	.242	467
52	Autopilot	.160	320
61	Communications	.441	589
65	IFF	980.	199
71	Radio Navigation	.512	573
72	Radar Navigation	. 433	525
Avera	Average Sortie Length	1.000	710
Total	Total Number of Sorties	710	1.000

TABLE 19. CORRELATIONS FOR C-141A AIRCRAFT (Jun 76 - May 77)

System		Average Sortie Length	Total Number of Sorties
			737
11	Airframe	.316	1.404
12	Fuselage Compartments	.271	314
13	Landing Gear	.250	396
14	Flight Controls	.240	389
23		.346	478
24	Aux Power Plant	.270	260
41	Air Cond	.284	353
42	Electrical Power Supply	.189	272
77	Lighting Systems	.254	282
45	Hydraulic & Pneumatic Sys	.332	325
94	Fuel Systems	790.	321
47	Oxygen Supply	.194	257
67		.422	418
51	Instruments	.262	332
52	Autopilot	.145	370
61	Communications	.165	401
65	IFF	060.	291
71	Radio Navigation	.146	388
72		.235	348
Verage	Average sortie length	1.000	398
70401		- 398	1.000

### LINEAR REGRESSION RESULTS AND ANALYSIS

Linear regression was performed for each aircraft type and system, and the resultant intercepts, slopes, variance and R<sup>2</sup> are presented in Tables 6 through 11 for C-130E and C-141A aircraft. Regression was also used on the airlift data grouped by base, and the results appear in Tables 12 through 17. Correlations of the data are presented in Tables 18 and 19.

The Coefficient of Variation (C.V.) is the relative variability, written as a fraction or percentage, and is defined as the standard deviation divided by the mean. <sup>15</sup> The squared linear correlation coefficient is used to express the ratio of explained variation to total variation. <sup>16</sup> The standard error of the estimate is a measure of the scatter about the regression curve. <sup>17</sup>

From the Linear Regression Results (Tables 6 through 11) it can be seen that for some systems the maintenance failure rate is somewhat sortic length related and for many others the sortic length does not account for much of the variation in maintenance requirements. In each case, this information is quite useful in determining what, if any, adjustment should be made in the forecasted failure rates of developmental aircraft. Some maintenance relationships to cycles and flight hours are graphically displayed in Appendix C.

As can be seen from the figures in Appendix C and Tables 2 and 6 through 11 the number of maintenance actions for nearly every system is largest for the B-52D. Also, for nearly every system the number of maintenance actions for the civilian aircraft is smallest. These observations hold true for both cyclic and flight hour related failures.

From Tables 12 through 17 it can be seen that the data for the C-130E and C-141A aircraft, when grouped by base, displays a base related effect that may overshadow the sortic length effect in this narrow range.

Whereas the civilian aircraft and the B-52D both flew consistent predetermined sorties of the same length and mission profile, each C-130E and C-141A aircraft fly more varied mission profiles and sortie lengths. Cargo airlift, airdrop, training, passenger, air evacuation, and other missions are flown by the C-130E and C-141A aircraft. Thus, it is not unexpected that there is much more variance in the maintenance requirements for the C-130E and C-141A aircraft.

The data for the C-130E and C-141A aircraft was also grouped by sortie length classes as follows:

For the C-130E (Jun 76 - May 77 data)

Class	Average Sortie Length (Hours)
1	0 - 2.5
2	2.5 - 3.5
3	3.5 - 4.5
4	4.5 - 7.0

For the C-141A (Jun 76 - May 77 data)

Class	Average Sortie Length (Hours)
1	0 - 3.0
2	3.0 - 3.5
3	3.5 - 4.0
4	4.0 - 7.0

The results of the linear regression analysis of these two data sets are presented in Tables 20 and 21 respectively. It is easily observed that the effects of this grouping by sortic length illustrate the cyclic and flying hour factors for each system.

TABLE 20. LINEAR REGRESSION RESULTS FOR C-141A AIRCRAFT (Data grouped by Sortie Lengths)

Jun 76 - May 77

Average maintenance actions per cycle (Intercept) and Flt Hr (Slope)

f ustimate	ariation	on coefficient
Standard error of	Coefficient of variation	Linear correlation
S.E.E. =	C.V. =	R <sup>2</sup> =

System	E	Cycle	(S.E.E.)	Flt Hr	(S.E.E.)	R <sup>2</sup>	C.V.
1	, i. i.	- 182	( 083)	189	(760)	026.	5.0
17	Alliame	701.1	(000)	790	(700)	700	2
12	Fuselage Compartments .	1.0/4	(:013)	.004	(+00+)	774	6.7
13	Landing Gear	059	(.017)	.092	(:002)	766.	1.9
14	Flight Controls	017	(.024)	890.	(.007)	.980	3.2
23		110	(.037)	.162	(.010)	.992	2.3
28	Aux Power Plant	029	(.013)	.024	(,004)	.957	6.7
41	Air Cond	034	(900)	.036	(.002)	. 995	2.0
42	Electrical Power Supply	900	(.014)	.020	(,004)	.928	6.3
77	Lighting Systems	106	(.023)	.087	(900.)	686.	3.3
45	Hydraulic & Pheumatic Sys	130	(.010)	.077	(.003)	.997	2.2
97		980.	(.027)	.011	(*00.)	. 505	6.3
77	Oxygen Supply	016	(.007)	.021	(.002)	.982	3.6
65	Misc Utilities	072	(.020)	.045	(900.)	696.	8.9
51	Instruments	040.	(.025)	.021	(.007)	.820	6.3
52	Autopilot	040	(900.)	.017	(.002)	.981	1.7
19	Communications	.115	(,010)	.024	(.003)	.970	1.5
65	441	002	(.002)	.005	(.001)	.970	4.0
71	Radio Navigation	.067	(.029)	.018	(*008)	.717	7.9
72	Radar Navigation	.013	(.027)	.047	(.008)	.950	4.4

TABLE 21. LINEAR REGRESSION RESULTS FOR C-130E AIRCRAFT (Data grouped by Sortie Lengths)

Jun 76 - May 77

Jun 76 - May 77 Average Maintenance Actions per Cycle (Intercept) and Flt Hr (Slope)

S.E.E. = Standard error of estimate C.V. = Coefficient of variation R<sup>2</sup> = Linear correlation coefficient

System	а	Cycle	(S.E.E.)	Flt Hr	(S.E.E.)	R2	c.v.
111 122 133 144 145 147 147 147 147 147 147 147 147 147 147	Airframe Fuselage Compartments Landing Gear Flight Controls Engine Aux Power Plant Air Cond Electrical Power Supply Lighting Systems Hydraulic & Pneumatic Sys Fuel Systems Oxygen Supply Misc Utilities Instruments Autopilot Communications IFF		(.006) (.008) (.008) (.008) (.009) (.009) (.001) (.001) (.008) (.008) (.008) (.008)		(.002) (.003) (.003) (.004) (.002) (.003) (.003) (.003) (.003) (.003)	. 985 . 986 . 9877 . 940 . 9877 . 987 . 987 . 988 . 989 . 980 . 980 . 980 . 980 . 980 . 980 . 980 . 980 . 980 . 980	2.2.2. 13.4.2. 10.0.2.
71 72	Radio Navigation Radar Navigation	.069	(.017)	.070	(.003)	.995	2.2

### FACTOR ANALYSIS RESULTS AND DISCUSSION

Also, a preliminary factor analysis was considered and a stepwise regression program used to check the relative impact of several factors including the total sorties flown, average sortie length, utilization in terms of flight hours and sorties per month, and the base of assignment. The results of this stepwise regression analysis are presented in Tables 22 and 23. A second order regression was checked but due to a large variance and limited range of sortie lengths, results did not appear to be meaningful.

The stepwise regressions (Tables 22 and 23) illustrate the need for further study in the area of factor analysis. As can be seen from these two tables the average monthly utilization of the aircraft and the total sorties flown seem to provide a much greater explanation of the variation in the average number of maintenance actions per sortie than does the average sortie length for C-130E and C-141A aircraft. However, these figures are quite often the result of maintenance rather than the drivers of it. If the aircraft is down for maintenance often, it cannot be utilized to fly as many missions or sorties as can the aircraft that requires less frequent maintenance.

Also, during peacetime the aircraft are not planned to fly as much and the flight crews may be more insistent upon having repairs accomplished before flight. During wartime or surge efforts the aircraft are often required to be utilized at the maximum level maintenance will permit.

TABLE 22. STEPWISE REGRESSION RESULTS FOR C-130E AIRCRAFT
(Data from Jun 76 - May 77)
Model Used is Average Maintenance Actions = Variable as Follows:

(.520)

(.519)

S

(.519)

B

(.514)

(.392)

H

All Maintenance

TABLE 23. STEPWISE REGRESSION RESULTS FOR C-141A AIRCRAFT (Data from Jun 76 - May 77)
Model Used is Average Maintenance Actions = Variable as Follows:

		L S H B C L	Total sorties Code for home Average utiliz Average utiliz Average sortie	Total sorties for time period Code for home base of aircraft Average utilization in hours per mc Average utilization in sorties per Average sortie length for period	for time period base of aircraft ation in hours pation in sorties length for peri	t per mo s per iod	onth month				
System	E	Step 1	(R <sup>2</sup> )	Step 2	(R <sup>2</sup> )	Step 3	(R <sup>2</sup> )	Step 4	(R <sup>2</sup> )	Step 5	$(\mathbb{R}^2)$
1	Airframe	, v	(,244)	88	(395)	H(S+F)	(.399)	Ţ	(,399)	S	(007)
12	Fuselage Compartments	М	(,237)	S	(308)	, H	(.309)	H(S→L)	(.310)	S	(.310)
13	Landing Gear	В	(.271)	S	(394)	H(S→L)	(.404)	T	(*708)	S	(*08)
14	Flight Controls	S	(.176)	М	(.280)	H(S→L)	(.291)	S	(.298)	I	(.298)
22	Engine	S	(.244)	В	(.340)	₽	(.343)	$\Gamma(S\rightarrow H)$	(.350)	S	(.354)
24	Aux Power Plant	В	(.158)	S	(,208)	н	(.212)	Н	(.212)	,,	(.212)
41	Air Cond	В	(.299)	ĘŦ	(.371)	7	(.372)	S	(.372)	.11	(.378)
42	Electrical Power Supply	В	(,224)	T	(.263)	H(T→L)	(.267)	S	(.276)	Н	(.281)
77	Lighting Systems	В	(.288)	T	(.327)	1	(.328)	Н	(.328)	S	(.328)
45	Hydraulic & Pneumatic Sys	В	(346)	S	(907)	T(S→H)	(.413)	S	(.415)	I	(.416)
95	Fuel Systems	T	(.103)	æ	(.113)	В	(1116)	S(B+T)	(.140)	В	(.143)
47	Oxygen Supply	В	(.203)	T	(.238)	S	(.239)	н	(.239)	IJ	(.247)
65	Misc Utilities	В	(.241)	S	(,384)	T(S→H)	(907)	S	(.411)	H.	(.412)
51	Instruments	В	(.122)	S	(.205)	T(S→H)	(.213)	S	(.222)	Н	(.225)
52	Autopilot	S	(.149)	В	(.178)	H(S→L)	(.190)	H	(.190)	S	(.190)
65	IFF	T	(,085)	В	(.116)	S	(.127)	Н	(.131)	L	(.138)
7.1	Radio Navigation	H	(.150)	В	(.164)	н	(.168)	S	(.169)	IJ	(.183)
72	Radar Navigation	В	(.252)	T	(.326)	Н	(.326)	L	(.326)	S	(.326)

All Maintenance

Thus, it follows that for a planned consistent utilization rate this type approach may be useful, but in practice, utilization is far less certain than the expected sortie lengths.

Also, it need be noted, the stepwise regression analyses of the C-130E and C-141A aircraft, in no case explains more than about half of the variation in maintenance data. Thus, we might look to other factors, random chance, or data error as reasons for much of the variation in the maintenance data.

An Analysis of Variance (ANOVA) was accomplished upon the airlift data grouped by sortie length classification as previously described, total sorties flown, average utilization by flight hours and sorties per month, and by aircraft assigned home base. Total sorties were grouped as follows:

for the C-130E Aircraft (Jun 76 - May 77 data)

### Class

- 1 less than 126 sorties 2 126 to 175 sorties 3 176 to 250 sorties
- 4 more than 250 sorties

for the C-141A Aircraft (Jun 76 - May 77 data)

# Class

1 up to 250 sorties 2 250 to 325 sorties 3 326 to 400 sorties 4 more than 400 sorties

The average monthly utilization by flying hours was grouped as follows:

for the C-130E Aircraft (Jun 76 - May 77 data)

### Class

1	up to 40 hours per month
2	40 to 50 hours per month
3	50 to 60 hours per month
4	more than 60 hours per month

for the C-141A Aircraft (Jun 76 - May 77 data)

## Class

1	up to 80 hours per month
2	80 to 95 hours per month
3	95 to 110 hours per month
4	more than 110 hours per month

The average utilization in number of sorties per month was grouped as follows:

for the C-130E Aircraft (Jun 76 - May 77 data)

### Class

1	less than 10 sorties per month
2	10 to 17 sorties per month
3	17 to 24 sorties per month
4	more than 24 sorties per month

for the C-141A Aircraft (Jun 76 - May 77 data)

### Class

1	less than 23 sorties per month
2	23 to 27 sorties per month
3	27 to 32 sorties per month
4	more than 32 sorties per month

The base of assignment was also used as a classification for the ANOVA research. The results of the ANOVA, presented in Tables 24 and 25, illustrate that there is a significant impact upon maintenance requirements caused by each of the factors that were analyzed. There also was significant interaction between many of the variables.

TABLE 24. RESULTS OF ANOVA FOR C-130E AIRCRAFT

(Data from Jun 76 to May 77)

F Value - Sum of squares between treatments divided by sums of squares within treatments

PR>F - Probability of chance F value occurring given no effect of variable(s)

R<sup>2</sup> - Linear correlation coefficient

חדווכמו כסייבודמוי כסייודרייור	ation
11011011	of variation
11100	Coefficient of
חדוונם	Coeff
	1
4	C.V.

			Sortie	Sortie Length	Total Sorties	orties	Hrs/Month	onth	
(V)	System	ES.	F Value	(PR>F)	F Value	(PR>F)	F Value	(PR>F)	
1									
	11	Airframe	9.17	(,0001)	94.79	(,0001)	65.69	(,0001)	
	12	Fuselage Compartments	6.65	(,0004)	38.44	(.0001)	70.60	(.0001)	
	13	Landing Gear	45.99	(.0001)	124.05	(,0001)	59.82	(.0001)	
	14	Flight Controls	1.17	(.3243)	37.36	(,0001)	46.35	(.0001)	
	22	Engine	11.71	(.0001)	60.91	(,0001)	48.86	(.0001)	
	24	Aux Power Plant	2.14	(0860)	32.84	(.0001)	53.06	(.0001)	
	41	Air Cond	9.04	(,0001)	78.05	(.0001)	72.29	(.0001)	
	42	Electrical Power Supply	1.72	(.1662)	32.23	(,000)	49.39	(.0001)	
	77	Lighting Systems	10.00	(,0001)	47.13	(.0001)	51.03	(.0001)	
	45	Hydraulic & Pneumatic Sys	8.61	(.0001)	61.71	(,0001)	42.28	(.0001)	
	94	Fuel Systems	21.98	(,0001)	43.60	(.0001)	29.01	(.0001)	
45	47	Oxygen Supply	7.68	(.0001)	46.39	(,0001)	29.04	(.0001)	
	64	Misc Utilities	1.08	(3598)	42.82	(.0001)	67.61	(.0001)	
	51	Instruments	7.96	(.0001)	36.84	(,0001)	29.84	(,0001)	
	52	Autopilot	1.38	(.2517)	8.70	(,0001)	6.03	(6000)	
	61	Communications	29.21	(,0001)	59.40	(.0001)	32.97	(.0001)	
	65	IFF	0.65	(.5910)	5.55	(,0016)	87.9	(:000)	
	71	Radio Navigation	20.61	(.0001)	29.16	(,0001)	5.42	(,0018)	
	72	Radar Navigation	30.20	(,0001)	77.97	(.0001)	18.96	(.0001)	
Ţ	otal	Total Maintenance	30.43	(,0001)	154.42	(,0001)	128.84	(.0001)	

TABLE 24. RESULTS OF ANOVA FOR C-130E AIRCRAFT (Cont'd) (Data from Jun 76 to May 77)

		Sortie	Sorties/Month	Assign	Assigned Base				
System	ща	F Value	(PR>F)	F Value	(PR>F)	Model F Value	PR>F	R <sup>2</sup>	C.V.
11	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	67 82	(,0001)	11 27	(10001)	96.7	.0001	.833	29.4
12	Fire lase Compartments	36.48	(,0001)	11.39	(,0001)	4.42	.0001	.817	32.4
13	Landing Gear	118.69	(.0001)	9.98	(.0001)	5.86	.0001	.855	28.2
14	Flight Controls	30.88	(.0001)	7.07	(.0001)	3.64	.0001	.786	38.5
22	Engine	84.20	(.0001)	22.95	(.0001)	5.29	1000.	.842	27.5
24	Aux Power Plant	33.46	(.0001)	17.33	(.0001)	4.79	.0001	.829	34.2
41	Air Cond	89.44	(.0001)	17.36	(.0001)	6.34	.0001	.865	29.0
42	Electrical Power Supply	40.56	(.0001)	15.61	(.0001)	4.05	.0001	.804	37.8
77	Lighting Systems	53.67	(.0001)	7.93	(.0001)	06.4	.0001	.832	33.7
45	Hydraulic & Pneumatic Sys	63.81	(.0001)	16.07	(.0001)	6.83	.0001	.873	29.7
94	Fuel Systems	54.17	(.0001)	8.37	(.0001)	3.10	.0001	.758	39.0
146	Oxygen Supply	53.43	(.0001)	7.35	(.0001)	3.92	.0001	.798	6.04
65	Misc Utilities	54.24	(.0001)	26.70	(.0001)	7.03	.0001	.877	34.1
51	Instruments	48.78	(.0001)	22.88	(.0001)	4.87	.0001	.831	30.1
52	Autopilot	8.79	(.0001)	5.55	(,0001)	1.55	.0132	.610	51.3
61	Communications	74.80	(.0001)	9.52	(.0001)	4.31	.0001	.812	26.2
65	IFF	3.55	(.0171)	2.08	(.0327)	1.47	.0252	.601	8.76
71	Radio Navigation	23.50	(.0001)	2.05	(.0353)	1.74	.0027	.637	37.2
72	Radar Navigation	46.90	(.0001)	11.81	(.0001)	7.06	.0001	804	56.9
Total	Total Maintenance	175.34	(.0001)	23.22	(.0001)	9.58	.0001	906.	17.4

TABLE 25. RESULTS OF ANOVA FOR C-141A AIRCRAFT

(Data from Jun 76 to May 77)

F Value - Sum of squares between treatments divided by sums of squares within treatments PR>F - Probability of chance F value occurring given no effect of variable(s)

R2 - Linear correlation coefficient

C.V. - Coefficient of variation

		Sortie Length	Length	Total Sorties	orties	Hrs/Month	onth	
System	ша	F Value	(PR>F)	F Value	(PR>F)	F Value	(PR>F)	
		•				;		
11	Airframe	14.46	(.0001)	21.82	(.0001)	18.81	(10001)	
12	Fuselage Compartments	10.63	(,0001)	13.39	(10001)	7.36	(.0002)	
13	Landing Gear	11.13	(,0001)	19.02	(,0001)	16.19	(.0001)	
14	Flight Controls	8.66	(,0001)	18.12	(.0001)	16.20	(.0001)	
23	Engine	17.30	(.0001)	24.76	(,0001)	14.00	(.0001)	
24	Aux Power Plant	8.22	(.0001)	5.40	(.0017)	2.44	(.0662)	
41	Air Cond	12.44	(.0001)	14.26	(.0001)	6.62	(,0004)	
42	Electrical Power Supply	7.55	(,0001)	11.26	(,0001)	7.83	(.0001)	
77	Lighting Systems	11.26	(.0001)	9.39	(,0001)	3.61	(.0151)	
57	Hydraulic & Pneumatic Sys	18.19	(.0001)	14.86	(,0001)	6.23	(9000')	
94		2.74	(.0450)	16.13	(.0001)	14.43	(.0001)	
47	Oxygen Supply	8.31	(,0001)	9.73	(,0001)	4.39	(.0058)	
67	Misc Utilities	25.14	(.0001)	23.13	(.0001)	9.36	(.0001)	
51	Instruments	8.20	(,0001)	11.49	(,0001)	6.51	(:000)	
52	Autopilot	3.28	(.0230)	11.98	(.0001)	15.08	(.0001)	
61	Communications	2.60	(.0539)	17.39	(.0001)	10.25	(.0001)	
65	IFF	1.46	(,2266)	9.00	(.0001)	7.03	(:0003)	
71	Radio Navigation	4.14	(*0018)	17.23	(.0001)	12.03	(.0001)	
72	Radar Navigation	10.76	(.0001)	12.04	(.0001)	7.11	(.0002)	
Total	Total Maintenance	23.61	(.0001)	37.19	(,0001)	23.91	(,0001)	

TABLE 25. RESULTS OF ANOVA FOR C-141A AIRCRAFT (Cont'd) (Data from Jun 76 to May 77)

		Sorties	Sorties/Month	Assign	Assigned Base				
System	шe	F Value	(PR>F)	F Value	(PR>F)	Model F Value	PR>F	R <sup>2</sup>	C.V.
11	Airīrame	31.71	(,0001)	30.51	(.0001)	3.07	.0001	769.	26.2
12	Fuselage Compartments	17.13	(,0001)	30.48	(,0001)	2.73	.0001	899.	33.9
13	Landing Gear	25.58	(,0001)	32.08	(,0001)	3.18	.0001	.701	28.1
14	Flight Controls	21.92	(,0001)	20.78	(.0001)	3.08	.0001	. 695	30.2
23	Engine	28.81	(,0001)	22.58	(.0001)	2.95	.0001	.685	25.4
24	Aux Power Plant	7.37	(,0002)	15.0	(.0001)	1.72	.0021	.559	50.1
41	Air Cond	14.38	(,0001)	31.78	(,0001)	3.05	.0001	.692	29.8
42	Electrical Power Supply	8.73	(.0001)	32.44	(.0001)	2.91	.0001	.682	31.2
77	Lighting Systems	12.04	(.0001)	99.97	(.0001)	3.14	.0001	669.	37.9
45	Hydraulic & Pneumatic Sys	19.35	(,0001)	53.38	(.0001)	3.50	.0001	.721	35.9
97	Fuel Systems	18.18	(,0001)	3.25	(.0140)	2.47	.0001	979.	28.7
17 4	Oxygen Supply	8.91	(.0001)	27.59	(.0001)	2.49	.0001	879.	38.7
67	Misc Utilities	31.36	(,0001)	33.83	(.0001)	2.86	.0001	629.	33.1
51	Instruments	11.87	(,0001)	12.06	(.0001)	1.88	,000	. 58	25.9
52	Autopilot	15.83	(.0001)	6.81	(.0001)	1.99	.0001	. 596	30.3
61	Communications	14.32	(,0001)	10.01	(,0001)	1.67	.0033	.552	23.6
65	7:14	5.15	(.0023)	5.79	(.0003)	2.35	.0001	.635	64.5
71	Racio Navigation	13.00	(.0001)	6.47	(.0001)	2.16	.0001	.615	28.0
72	Radar Navigation	15.70	(.0001)	24.52	(.0001)	2.49	.0001	879.	26.1
Total	Total Maintenance	44.58	(.0001)	45.68	(.0001)	4.39	.0001	.764	19.2

# PROBLEMS AND QUALIFICATIONS

### DATA PROBLEMS

In reducing the data a number of problems were encountered with the usage of the MDC system. Not only was there often missing data, but there were apparent inconsistencies in how some items were documented. Many maintenance actions may not have been documented at all, some appear to have been grouped differently when reported from various bases, and some may have been in error. In any case a review of the MDC system has been initiated and a more accurate method of obtaining data will undoubtedly be of assistance in future research.

#### DATA NOT AVAILABLE

If, in the future, failure data could be documented according to the time within the sortie the failure occurred, then a more accurate analysis of flight phase failure rate might be undertaken.

Differences between ages of the aircraft, number of landings and crew techniques were not available for this study. However, it might be noted that each of the bases is assigned aircraft of varying ages, and all crews are trained at a common location using a standardized methodology and that crew members and maintenance personnel are often reassigned to other bases having the same type aircraft.

### NUMBER OF AIRCRAFT TYPES AND DATA TIME PERIODS

Another limiting factor is the time and number of aircraft types studied. Data for different periods of time might be considered as well as other aircraft types in order that a more complete study might be undertaken in the future.

### CONCLUSIONS AND RECOMMENDATION

#### USAGE OF RESULTS

From the correlations (Tables 18 and 19) and linear regression results (Tables 2, and 6 through 17) it can be concluded that there is, to varying degrees, a cyclic and a flight hour failure factor involved in aircraft reliability. In order to use the information gained from this study the engineer need only compute the percent of maintenance that is cyclic or sortic related (PSR) for each sortic length in question. See Figure 3.

Since the processed MDC data gives the engineer the mean number of sorties between maintenance actions (MSBMA1) for the comparison aircraft with a known sortie length  $(S_1)$ , the average cyclic or sortie related failure constant can be computed as follows:

MAPS = 
$$(\frac{1}{MSBMA_1})$$
 (PSR<sub>1</sub>) = a constant

where MAPS = maintenance actions per sortie associated with the cycle only  $(\frac{1}{\text{MSBMA}_1})$  = average maintenance actions per sortie of length  $S_1$  PSR<sub>1</sub> = percent of maintenance that is sortie related for sortie length  $S_1$  (from Tables 26 through 29 or as computed by the engineer).

So we see that the mean number of sorties between maintenance actions for the developmental aircraft (MSBMA $_2$ ) can be calculated from the known information

now MAPS = 
$$(\frac{1}{MSBMA_2})$$
 (PSR<sub>2</sub>) = known (from above)  
so solving the equation for MSBMA<sub>2</sub>  
gives MSBMA<sub>2</sub> =  $\frac{PSR_2}{MAPS}$  = MSBMA<sub>1</sub> ( $\frac{PSR_2}{PSR_1}$ )

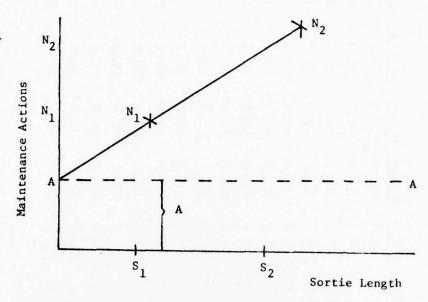


Figure 3. Average Maintenance Actions vs Sortie Lengths

 $\mathbf{N}_{\mathbf{i}}$  = total average maintenance actions per sortie length  $\mathbf{i}$ 

i = 1, 2, 3, . . .

A = average number of failures per cycle only (regression line intercept) percent sortic related (PSR) =  $\frac{A}{N_i}$ 

PERCENT OF MAINTENANCE ACTIONS THAT ARE SORTIE RELATED (by sortie length using B-52 regressed data only) TABLE 26.

	Sortie											
System	1 Hr	2 Hrs	3 Hrs	4 Hrs	5 Hrs	6 Hrs	7 Hrs	8 Hrs	9 Hrs	10 Hrs	11 Hrs	12 Hrs
11 Airframe	77.4	63.2	53.4	46.2	40.7	36.4	32.9	30.0	27.6	25.6	23.8	22.2
12 Fuselage	89.3	9.08	73.5	67.5	62.5	58.1	54.3	51.0	48.1	45.4	43.1	41.0
13 Landing Gear	98.5	97.1	95.7	94.3	93.0	91.7	90.4	89.2	88.0	86.9	85.8	84.7
14 Flz. Controls	89.7	81.4	74.5	9.89	63.6	59.3	55.6	52.2	49.3	46.7	44.3	42.2
23 Engines	8.69	53.6	43.5	36.6	31.6	27.8	24.8	22.4	20.4	18.8	17.4	16.1
41 Air Cond.	90.1	82.0	75.3	69.5	64.6	60.3	9.99	53.3	50.3	47.7	45.3	43.2
12 Elec. Pwr.	8.06	83.2	76.8	71.3	66.5	62.3	58.6	55.4	52.5	49.8	47.4	45.3
14 Lighting	57.0	39.8	30.6	24.9	20.9	18.1	15.9	14.2	12.8	11.7	10.7	6.6
15 Hyd. Pwr.	87.1	77.2	69.3	62.9	57.5	53.0	49.2	45.8	42.3	40.4	38.1	36.1
16 Fuel	70.7	54.7	44.6	37.7	32.6	28.7	25.7	23.2	21.2	19.5	18.0	16.8
17 Oxygen	92.3	85.7	6.62	74.9	70.5	9.99	63.1	59.9	57.0	54.4	52.1	49.9
il Inst.	97.3	94.8	92.4	90.1	88.0	85.9	83.9	82.0	80.2	78.5	6.97	75.3
32 Autopilot	98.8	9.76	96.4	95.3	94.2	93.1	92.0	91.0	0.06	89.0	88.0	87.1
33 UHF Comm.	97.9	6.36	94.0	92.1	90.4	88.6	87.0	85.4	83.9	82.4	81.0	9.62
'4 Fire Cont.	6.69	53.7	43.6	36.7	31.7	27.9	24.9	22.5	20.5	18.8	17.4	16.2
HF + Interphone	97.0	94.1	91.5	88.9	86.5	84.3	82.1	80.1	78.1	76.3	74.5	72.8

TABLE 27. PERCENT OF MAINTENANCE ACTIONS THAT ARE SORTIE RELATED (by sortie length using civilian regressed data only)

	12 Hrs	8.	0	9.6	.87	1.6	4.6	5.0	3.4	.0	1.7	14.3	4.5	7.4	3.9	7.4	3.9
	11 Hrs	.85	0	10.4	6.	1.8	5.0	5.4	3.7		1.8	15.4	4.9	8.0	4.3	8.0	4.3
	10 Hrs	6.	0	11.3	1.0	1.9	5.5	5.9	4.1	3.1	2.0	16.7	5.4	8.8	4.7	8.8	4.7
	9 Hrs	1.0	0	12.4	1.2	2.2	6.1	6.5	4.5	3.4	2.2	18.2	5.9	6.7	5.2	7.6	5.2
	8 Hrs	1.2	0	13.7	1.3	2.4	8.9	7.3	5.1 4.5	3.8	2.5	20.0	9.9	10.7	5.8	10.7	5.8
									5.8								
	1								9.9								
									7.9								
									9.6								
	3 Hrs	3.0	0	29.8	3.4	6.4	16.2	17.4	12.5	9.5	6.4	40.0	15.9	24.3	14.0	24.3	14.0
	2 Hrs	4.5	0	38.9	5.0	9.6	22.5	24.0	17.6	13.7	9.3	50.0	22.1	32.5	19.7	32.5	19.7
Sortie Length	1 Hr	8.6	0	56.0	9.5	19.1	36.7	38.7	29.9	24.0	17.0	66.7	36.1	49.0	32.9	49.0	32.9
	System	11 Airframe	12 Fuselage	13 Landing Gear	14 Flt. Controls	23 Engines	41 Air Cond.	42 Elec. Pwr.	44 Lighting	45 Hyd. Pwr.	46 Fuel	47 Oxygen	51 Inst.	52 Autopilot	53 UHF Comm.	74 Fire Cont.	UHF + Interphone

TABLE 28. PERCENT OF MAINTENANCE ACTIONS THAT ARE SORTIE (CYCLIC) RELATED (by sortie length using C-130E regressed data)

								S	Sortie Length	Length			
S	System	E	1 hr	2 hr	1 hr 2 hr 3 hr	4 hr	5 hr	6 hr	7 hr	8 hr	9 hr	10 hr	
1													
	11	Airframe	76.5	62.0	52.1		39.5	35.2	31.8	29.0	26.6	24.6	
	12	Fuselage Compartments	85.9	75.3	67.1	7.09	55.0	50.4	9.97		40.4	37.9	
	13	Landing Gear	3.1	1.6	1.05	. 79			.45	04.	.35	.32	
	14	Flight Controls	91.0	83.5	77.1	71.7			59.1		53.0	50.3	
	22	Engine	74.1	58.9	48.8	41.7		32.3	29.0	26.3	24.1	22.2	
	24	Aux Power Plant	8.66	7.66	99.1	8.86		98.2	6.76	97.7	97.4	97.1	
	41	Air Cond	78.6	8.49	55.1	78.0		38.0	34.5	31.5	29.0	26.9	
	42	Electrical Power Supply	91.0	83.5	77.2	71.7		62.8	59.2	55.9	53.0	50.4	
	77	Lighting Systems	82.4	70.1	61.0	54.0		43.8	40.1	36.9	34.2	31.9	
	45	Hydraulic & Pneumatic Sys	6.97	62.5	52.6	45.5		35.7	32.3	29.4	27.0	25.0	
5	97	Fuel Systems	39.7	24.8	18.0	14.1		6.6	8.6	7.6	8.9	6.2	
4	14	Oxygen Supply	0.09	42.9	33.3	27.3		20.0	17.6	15.8	14.3	13.0	
	65	9 Misc Utilities	8.06	83.1	9.9/	71.1		62.1	58.4	55.1	52.2	9.65	
	51	Instruments	9.97	62.0	52.1	45.0		35.3	31.8	29.0	26.6	24.6	
	52	Autopilot	83.9	72.3	63.5	9.99		46.5	42.7	39.5	36.7	34.3	
	61	Communications	50.0	33.3	25.0	20.0		14.3	12.5	11.1	10.0	9.1	
	65	IFF	82.4	70.0	6.09	53.8		43.8	0.04	36.8	34.1	31.8	
	71	Radio Navigation	33.3	20.0	14.3	11.1	9.1	7.7	9.9	5.9	5.3	8.4	
	72	Radar Navigation	54.7	37.7	28.7	23.2		16.8	14.7	13.1	11.8	10.8	

TABLE 29. PERCENT OF MAINTENANCE ACTIONS THAT ARE SORTIE (CYCLIC) RELATED (By Sortie Length Using C-141A Regressed Data)

							Sortie	Length					
S,	System	ше	1 hr	2 hr	3 hr	4 hr	5 hr	6 hr	7 hr	8 hr	9 hr	10 hr	
	11	Airframe		0		0	0		0	0	0	0	
	12	Fuselage Compartments		0		0	0		0	0	0	0	
	13	Landing Gear		0		0	0		0	0	0	0	
	14	Flight Controls		0		0	0		0	0	0	0	
	23	Engine		0		0	0		0	0	0	0	
	24	Aux Power Plant		0		0	0		0	0	0	0	
	41			0		0	0		0	0	0	0	
	42	Electrical Power Supply	46.2	30.0	22.2	17.6	14.6	12.5	10.9	6.7	8.7	7.9	
	77			0		0	0		0	0	0	0	
	45	-		0		0	0		0	0	0	0	
	94	Fuel Systems		85.3		74.4	6.69		62.4	59.5	26.4	53.8	
_	47	Oxygen Supply		7.7		1.2	1.0		0.7	9.0	9.0	0.5	
	67	Misc Utilities		0		0	0		0	0	0	0	
	51	Instruments		28.8		16.8	13.9		10.3	9.5	8.2	7.5	
	52	Autopilot		74.4		50.8	43.8		34.4	31.1	28.3	26.0	
	61	Communications		8.89		52.4	8.94		38.6	35.5	32.8	30.6	
	65	IFF		37.3		22.9	19.2		14.5	12.9	11.7	10.6	
	11	Radio Navigation		63.6		46.7	41.2		33.3	30.4	28.0	25.9	
	72	Radar Navigation		30.4		17.9	14.4		11.1	6.6	8.9	8.0	

In this manner the engineer can take data from an operational aircraft flying a known average sortie length, and adjust the mean number of sorties between maintenance actions, or failure rate, to account for a planned different sortie length when making maintenance forecasts for a developmental aircraft.

Another method that can be used to forecast the failure rate, when using the data from this report as the comparison basis, is to use the intercept of the appropriate aircraft system regressed data added to the slope of this data times the number of hours in the expected sortie length of the developmental aircraft.

This second technique can only be used when using the actual data within this study whereas the first technique can be used, when we assume the percent sortic related (PSR) the same for other data sources as for one of the aircraft in this study, for many different comparison studies.

For instance, in forecasting maintenance requirements for a new Short Takeoff and Landing (STOL) aircraft, the Advanced Medium STOL Transport (AMST) many of the parts are the same or similar to those of the C-141A and C-130E aircraft. However, the planned sortic length for this new aircraft is about one hour. Thus either technique described above can be utilized in this case.

Some other parts of the developmental AMST are more like those of the C-5A. And so, if we make the bold assumption that maintenance in the relative areas of the C-5A are equally sortic related (has the same PSR) as those of the C-141A, we can use the first technique utilizing the PSRs from the C-141A and the MDC data from the C-5A.

\* Thus there are several situations which might occur. To illustrate by using some calculations of the AMST forecasted maintenance failure rate:

Situation 1: The electrical system (System 42) of the AMST is quite similar to that of the C-141A and so no reliability adjustment is necessary. However, the average sortic length associated with the C-141A data is 4 hours. The average planned sortic length of the AMST is 1 hour. The mean sortics between maintenance actions (MSBMA<sub>1</sub>) for the C-141A is 18. So the mean sortics between maintenance actions (MSBMA<sub>2</sub>) for the AMST can be calculated as follows:

$$MSBMA_2 = \frac{PSR_2}{PSR_1} \quad MSBMA_1$$

 $MSBMA_1 = 18$  from MDC data for C-141A

PSR<sub>2</sub> = 46.2 from Table 29 sortie length 1 hour

PSR<sub>1</sub> = 17.6 from Table 29 sortie length 4 hours

thus

$$MSBMA_2 = \frac{46.2}{17.2}$$
 (18) = 48

Situation 2: The Pitch Trim (WUC 14D) from the AMST is similar to that of the C-141A and so no adjustment is necessary to account for reliability. From Table 29 we see that the flight controls are not sortic related (PSR = 0) and so we go to Table 11 to find that the average number of failures per flight hour is .064. From the second

computational technique and recalling that the planned sortie length for the AMST is 1 hour,

$$MSBMA_2 = \frac{1}{maintenance actions per sortie} = \frac{1}{.064} = 16$$

Situation 3: The Fuel Controls (System 46J) of the AMST are to be like those of the C-5A. We must now assume the percent sortic related (PSR) for maintenance actions on the C-5A to be the same as those of the C-141A. Then from the first technique the MSBMA<sub>1</sub> for the C-5A is 45 from MDC data. The C-5A flew an average sortic length of 5 hours and the AMST is planned for 1 hour sorties. Then by utilizing Table 29,

$$MSBMA_2 = \frac{PSR_2}{PSR_1}$$
 (MSBMA<sub>1</sub>) =  $\frac{92.1}{69.9}$  (45) = 59.

# RECOMMENDATIONS FOR DATA COLLECTION AND FURTHER STUDY

Although further study into the factors that affect maintenance requirements is recommended, the use of the data for the most similar aircraft, as presented in this report, can be of benefit to the simulation engineer.

As emphasized earlier, there is a need for accurate maintenance data that includes the time into the sortie that each failure occurs. This data need be collected by aircraft serial number and include all relevant factors such as data, phase of flight, airframe age, age of failed part, utilization, historic mission summary, weather, where maintenance work was accomplished, etc. With a more complete data set a factor analysis may prove quite useful in future failure forecasting.

In the meantime it is recommended that MDC data be analyzed, using the techniques developed in this study, for other time periods and also for other aircraft, to further refine the results of this study.

# APPENDIX A

Data Reduction Technique for C-130E and C-141 Aircraft

The C-130E and C-141A data arrived from AFLC on sixteen 9-track tapes which listed the maintenance data in cronological order of all maintenance actions performed by type aircraft. First, these tapes were converted to 7-track tapes to speed processing on the CDC-6600 computer system.

Next, the relevant data was extracted and grouped by four digit aircraft tail numbers (last four digits of aircraft serial number). Then the data was grouped by aircraft system for each individual aircraft. Finally, the data was punched on IBM cards to be used in the Statistical Analysis System (SAS) to analyze the failure data. See Figure A-1. Computer programs and collection notes follow.

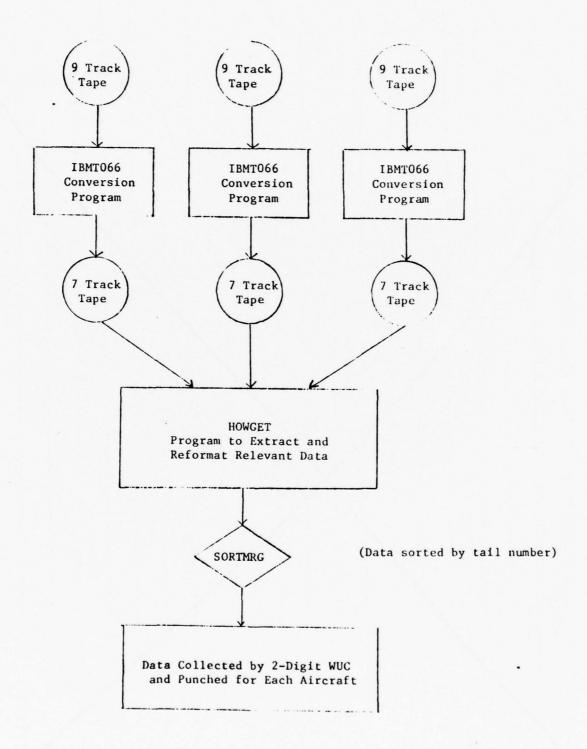


Figure A-1. Data Collection Method

T dats	
FT40052,05GIM,56521 9IK-7TK TO  PUNS OFF PEEL  S740052	
Z&L), SICSA, TIEDD, TOISDD, CHEDDDD, WIZ, FTWDDS VSR   ZADES XOUGET = 917097.  PAUSE DEERAIDZ, A TROP JOR HHEN IZD = 9UNS OF ECOUST TAPE 8, S, NORING.  PAUSE, TAPE 7 S A MULTI-RESL FILE.  VSN, TAPT = NOE 118 Y, SING, L= ATMOSD 1 TO PAUSE.  I POLICE.  A TACCH, 1947065 POWELLIAW, CY = 1, MR = 1.  I EMIDGE.  OILIA  OILIA	

ZIAJ,STCSA,TISOO,IOISOO,CM50000,MT2. E740052,BEGIN,56531 9TK-7TK X TO L STEP I VSN,TAPE8=X00867=01B007.
PAUSE.OPERATOR...DROP JOB WHEN TAPE8 RUNS OFF REL...
PAUSE.TAPE8,S.NORING.
PAUSE.TAPE8,S.NORING.
LABEL,TAPE7 IS A MULTI-REEL FILE.
LABEL,TAPE7,W,D=HY,RING,L=AFM66RITOMMFORM4. E740052
ATTACH,IBMT066,HOWELLIBM,CY=1,MR=1.
IBMT066.
RETURN,TAPE8,TAPE7.

C141A

Z1ZA,T2O,1050,MT1,STCSA,DBSO1. E740052,BEGIN,56531 ATTACH,PROG,IBMHOWELL,CY=33,SN=ASDEN,MR=1. LABEL,TAPE8,R,D=HY,L=C141AHOWELLDATA,VSN=L03810,NORING. LGO,PL=4000. RETURN,PROG,TAPE8.

65

Z1CA,STCSA,T350,10950,CM65000,MT2,DBS00. E740052,BEGIN,56531 C141ACREATE III E740052 LABEL, TAPE24, R, D=HY, NORING, L=C141ABYJCNOCTMAY. COPYCF, TAPE24, TAPE1. FILE, TAPE1, FO-SQ, BT-C, RT-Z, FL-150. ATTACH, COMBINE, COMBINE, ID-E740052, CY-2. REWIND(TAPE1, TAPE2)
ATTACH, COLLECT, COLLECT, ID=E740052, CY=1.
LDSET, PRESET=ZERO.COLLECT. ATTACH, ADJUST, ADJUST, ID=E740052, CY=2. ATTACH, REPORT, REPORT, ID=E740052, CY=2. LDSET, PRESET=ZERO. REWIND, TAPE8, TAPE9, TAPE10, TAPE12. FILE, TAPE12, FO=SQ, BT=C, RT=Z, FL=150. FILE, TAPE3, FO=SQ, BT=C, RT=Z, FL=33. REWIND(TAPE1,TAPE2) FILE,TAPE2,FO~SQ,BT=C,RT=Z,FL=41. LDSET(FILES=TAPE1/TAPE2) REWIND(TAPE1,TAPE2) FILE,TAPE2,FO=SQ,BT=C,RT=Z,FL=33. LDSET(FILES=TAPE1/TAPE2) VSN , TAPE24=L06119/L06120. LDSET(FILES=TAPE3/TAPE12) REWIND, TAPE3, TAPE12. REWIND, TAPE1, TAPE2. REWIND (TAPEI, TAPE2) LDSET, PRESET=ZERO. LDSET, PRESET=ZERO. RETURN, COLLECT. RETURN, COMBINE. RETURN, REPORT. RETURN, ADJUST. REWIND, TAPEL. REWIND, TAPE1. REWIND, TAPE4. LIMIT, 4000. SORTMRG(6C) SORTMRG(6C) SORTMRG(6C) SORTMRG(6C) COMBINE. ADJUST. REPORT.

ATTACH, TAPE17, DATABNK2, ID=E740052, CY=1.
LABEL, TAPE19, W, D=HY, L=C141AHOWELLDATA, VSN=L03810, RING.
ATTACH, WUCMG, WUCMG, ID=E740052, CY=1. REWIND, TAPE1, TAPE2. FILE, TAPE1, FO=SQ, BT=C, RT=Z, FL=150. ATTACH, COMBINE, COMBINE, ID=E740052, CY=2. ATTACH, THRELVL, THRELVL, ID=E740052, CY=1. ATTACH, COLLECT, COLLECT, ID=E740052, CY=1. LDSET, PRESET=ZERO. ATTACH, ADJUST, ADJUST, ID=E740052, CY=2. REWIND(TAPE1,TAPE2) ATTACH, REPORT, REPORT, ID=E740052, CY=2. REWIND(TAPE1, TAPE2) FILE, TAPE2, FO=SQ, BT=C,RT=Z,FL=41. LDSET(FILES=TAPE1/TAPE2) FILE, TAPE2, FO=SQ, BT=C, RT=Z, FL=33. LDSET(FILES=TAPE1/TAPE2) COPYCF, TAPE24, TAPE10. REWIND (TAPE1, TAPE2) REWIND (TAPE1, TAPE2) LDSET, PRESET=ZERO. LDSET, PRESET=ZERO. LDSET, PRESET=ZERO. LDSET, PRESET=ZERO. LDSET, PRESET=ZERO. RETURN, COLLECT. RETURN, THRELVL. RETURN, COMBINE. RETURN, REPORT. RETURN, ADJUST. REWIND, TAPE1. REWIND, TAPE1. REWIND, TAPE4. RETURN, WUCMG. SORTMRG(6C) SORTMRG(6C) SORTMRG(6C) THRELVL. COMBINE. COLLECT. ADJUST. REPORT. WUCMG.

3.

.

REWIND, TAPE1, TAPE2. FILE, TAPE1, FO=SQ, BT=G, RT=Z, FL≈150. ATTACH, COMBINE, COMBINE, 1D=E740052, CY=2. REWIND(TAPE1, TAPE2) ATTACH, COLLECT, COLLECT, ID=E740052, CY=1. LDSET, PRESET=ZERO. ATTACH, THRELVL, THRELVL, ID=E740052, CY=1. ATTACH, ADJUST, ADJUST, ID=£740052, CY=2. LDSET, PRESET≈ZERO. REWIND(TAPE1, TAPE2) ATTACH, REPORT, REPORT, ID=E740052, CY=2. FILE, TAPE12, FO=SQ, BT=C, RT=Z, FL=150. FILE, TAPE3, FO=SQ, BT=C, RT=Z, FL=33. LDSET(FILES=TAPE3/TAPE12) REWIND, TAPE17. ATTACH, WUCMG, WUCMG, ID=E740052, CY=1. REWIND(TAPE1, TAPE2) FILE, TAPE2, FO-SQ, BT=C, RT=Z, FL=33. LDSET(FILES=TAPE1/TAPE2) REWIND(TAPE1,TAPE2) FILE,TAPE2,FO=SQ,BT=C,RT=Z,FL=41. LDSET(FILES=TAPE1/TAPE2) REWIND, TAPE8, TAPE9, TAPE10, TAPE12. COPYCF, TAPE24, TAPE10. REWIND, TAPE3, TAPE12. LDSET, PRESET=ZERO. LDSET, PRESET=ZERO. LDSET, PRESET=ZERO. RETURN, THRELVL. RETURN, COMBINE. RETURN, COLLECT. RETURN, ADJUST. REWIND, TAPE1. REWIND, TAPE1. RETURN, WUCMG. REWIND, TAPE4. SORTMRG(6C) SORTMRG(6C) SORTMRG(6C) SORTMRG(6C) COLLECT. COMBINE. ADJUST. WUCMG.

ATTACH, COLLECT, COLLECT, ID=E740052, CY=1. ATTACH, THRELVL, THRELVL, ID=E740052, CY=1. FILE, TAPE1, FO=SQ, BT=C, RT=Z, FL=150. ATTACH, COMBINE, COMBINE, ID=R740052, CY=2. ATTACH, ADJUST, ADJUST, ID=E740052, CY=2. REWIND, TAPE8, TAPE9, TAPE10, TAPE12. FILE, TAPE12, FO=SQ, BT=C, RT=Z, FL=150. FILE, TAPE3, FO=SQ, BT≈C, RT=Z, FL=33. LDSET(FILES=TAPE3/TAPE12) REWIND, TAPE17. ATTACH, WUCMG, WUCMG, ID=E740052, CY=1. LDSET, PRESET=ZERO. REWIND(TAPE1,TAPE2) FILE,TAPE2,FO=SQ,BT=C,RT=Z,FL=33. LDSET(FILES=TAPE1/TAPE2) COPYCF, TAPE24, TAPE10. REWIND, TAPE3, TAPE12. REWIND, TAPE1, TAPE2. REWIND (TAPE1, TAPE2) RETURN, COLLECT. REWIND (TAPE1, TAPE2) LDSET, PRESET=ZERO. LDSET, PRESET=ZERO. LDSET, PRESET=ZERO. LDSET, PRESET=ZERO. LDSET, PRESET=2ERO. RETURN, COMBINE. RETURN, THRELVL. REWIND, TAPE1. RETURN, ADJUST. RETURN, REPORT. REWIND, TAPEI. RETURN, WUCMG. REWIND, TAPE4. SORTMRG(6C) SORTMRG(6C) SORTMRG(6C) COMBINE. THRELVL. ADJUST. REPORT. WUCMG.

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SORT(1,1,45,,4)
FILE(TAPE1,S,D,,R,N)
FILE(TAPE4,0,D,,R,N)
SEQ(37,ABCDEFGHLJKLMNOPQRSTUVWXYZO123456789)
KEY(A,C,1,29)
RECORD(1,U,45)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                             FILE(TAPE1, S, D, R, N)
FILE(TAPE2, O, D, R, N)
SEQ(37, ABCDEFGHIJKLMNOPQRSTUVWXYZ0123456789)
KEY(A, C, 1, 17)
RECORD(1, U, 33)
                                                                                                                                                                                                                                                                                                                                                                                                    ATTACH, THRELVL, THRELVL, ID=E740052, CY=1.
LDSET, PRESET=ZERO.
                                                         REWIND(TAPE1,TAPE2)
ATTACH,REPORT,REPORT,ID=E740052,CY=2.
LDSET,PRESET=ZERO.
                                                                                                                                                               REWIND, TAPE8, TAPE9, TAPE10, TAPE12.
FILE, TAPE12, FO=SQ, BT=C, RT=Z, FL=150.
FILE, TAPE3, FO=SQ, BT=C, RT=Z, FL=33.
LDSET(FILES=TAPE3/TAPE12)
                                                                                                                                                                                                                                                                                                                 ATTACH, WUCMG, WUCMG, ID=E740052, CY=1.
LDSET, PRESET=ZERO.
FILE, TAPE2, FO=SQ, BT=C, RT=Z, FL=41.
LDSET(FILES=TAPE1/TAPE2)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                 LAFSCDUMMYAFSC ASSIGNED TO RUN
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                      RETURN, TAPES, TAPE9, TAPE10.
RETURN, TAPE19.
                                                                                                                                                                                                                                                                         REWIND, TAPE3, TAPE12.
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                               800 800
                                                                                                                                                                                                                                                                                                                                                                                                                                                                   RETURN, THRELVL.
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                            SORT(1,1,33,,4)
                                                                                                                                                                                                                                                                                                   REWIND, TAPE17.
                                                                                                                                                RETURN, REPORT.
                                                                                                                                                                                                                                                                                                                                                                                  RETURN, WUCMG.
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                 TRANSF(Z1ZA)
                                       SORTMRG(6C)
                                                                                                                                                                                                                                                           SORTMRG(6C)
                                                                                                                                                                                                                                                                                                                                                                                                                                                THRELVL.
                                                                                                                            REPORT.
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                               800
                                                                                                                                                                                                                                                                                                                                                            WUCMG.
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FILE(TAPE1,S,D,,R,N)
FILE(TAPE2,O,D,,R,N)
SEQ(63 ,ABCDEFGHIJKLMNOPQRSTUVWXYZ0123456789+-*/()$= ,.# ()X"_16.?<>-a\chin :)
KEY(A,C,1,15)
RECORD(1,U,41)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                         SORT(1,1,33,,4)
FILE(TAPE12,S,D,,R,N)
FILE(TAPE3,O,D,,R,N)
SEQ(63, ABCDEFGHIJKLMNOPQRSTUVWXYZ0123456789+-*/()$= ,,#(\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2},\frac{1}{2
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                            SORT(1,1,41,4)
FILE(TAPE1,S,D,,R,N)
FILE(TAPE2,O,D,,R,N)
SEQ(63 ,ABCDEFGHIJKLMNOPQRSTUVWXYZ0123456789+-*/()$= ,.#(?\".16.?<>5\^5)\".:)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                   SORT(1,1,33,4)
FILE(TAPE1,S,D,,R,N)
FILE(TAPE2,0,D,,R,N)
SEQ(37,ABCDEFGHIJKLMNOPQRSTUVWXYZO123456789)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                               SORT(1,1,45,4)
FILE(TAPE1,S,D,,R,N)
FILE(TAPE4,O,D,,R,N)
SEQ(37,ABCDEFGHIJKLMNOPQRSTUVWXYZO123456789)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                1AFSCDUMMYAFSC ASSIGNED TO RUN
SORT(1,1,41,,4)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                     800 800 800
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                         KEY(A,C,1,15)
RECORD(I,U,41)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                     KEY(A,C,1,29)
RECORD(1,U,45)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                   KEY(A,C,1,17)
RECORD(1,U,33)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                  C141A
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                 END
                                                                                                                                                                                                                                                                                                                                                                                                                                                            END
```

```
SORT(1,1,33,4)
FILE(TAPE12,S,D,,R,N)
FILE(TAPE3,O,D,,R,N)
SEQ(63,ABCDEFGHIJKLMNOPQRSTUVWXYZO123456789+-*/()$= ,.# ()%"_1&'?<>$\pi^*()$."
RECORD(1,U,33)
SORT(1,1,33,,4)
FILE(TAPE12,S,D,,R,N)
FILE(TAPE3,O,D,,R,N)
SEQ(63, ABCDEFGHIJKLMNOPQRSTUVWXYZO123456789+-*/()$= ,,# ()?"_16'?<>3\^:)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                              FILE(TAPE1,S,D,,R,N)
FILE(TAPE4,O,D,,R,N)
SEQ(37,ABCDEFGHJKLMNOPQRSTUVWXYZO123456789)
KEY(A,G,1,29)
RECORD(1,U,45)
                                                                                                                                                                                                                                                                                                                                                         SORT(1,1,33,,4)
FILE(TAPE1,S,D,,R,N)
FILE(TAPE2,O,D,,R,N)
SEQ(37,ABCDEFGHIJKLMNOPQRSTUVWXYZO123456789)
                                                                                                                                                                                                                                                                                                                          LAFSCDUMMYAFSC ASSIGNED TO RUN
                                                                                                                                                                                                                                                                                                            800 800 800
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                               SORT(1,1,41,,4)
                                                                                                                                                                       SORT(1,1,45,,4)
                                                                   KEY(A,C,1,5)
RECORD(I,U,33)
                                                                                                                                                                                                                                                                                                                                                                                                                             KEY(A,C,1,17)
RECORD(I,U,33)
                                                                                                                                       C141A
                                                                                                                                                                                                                                                                            END
                                                                                                       END
                                                                                                                                                                                                                                                                                                                                                                                                                                                                  END
```

# C130E

20F ES DOSGE 0180 W 001  AC 130E: 76 JUN-76 SEP 77172  SP ASDIENESA	X03906
WP AFB 7-2701 Z AZ74  OCS 3 7080	L04548/L04135
20F ES DOSGE 0180 W 002  AC 130E 76 JUN- 76 SEP 77172  SP ASDIENESA  WPAFB 7-270L Z AZ80  OCS 3 7080	X03907 V L04122
L04548/L04135/L04122 =	L04220
20F ES DOSGE 0180 U 00L  C130E 760CT-77 MAY 77171  SP ASDIENESA  WPAFB 7-2679 Z AC66  0C53 7080	X03902 L01518/L01549

73

20F ES DOSGE C	01B0 U 002
C130E 760CT-77	
SP ASDIENE	The state of the s
WPAFB 7-26	79 Z AD3/ L01647/L02574
ocs 3 70	080
20F ES DOSGE O	0180 U 003
CIBUE 760CT-77	
SP ASDIE	ME3A V
WPAFB 7-26	179 Z AD91 LO2599/LO285
OCS 3 70	
A Trans Section Co.	
20F ES DOSGE	0180 U 004
	77 MAY 77171 X 03905
SP ASDIENE	EZY
WPAFB 7-26	79 Z BF54 L02877
ocs 3 7	
10111111025	
LU 164 1   LO 25 1	4/L02599/L02857/L02877
promote the production of the production of the	
Section to assert and the section of	
	74

#### - STANDARD

# C14/A

30F ES DOSGE 0180 U 00L  AC141A 76 JUN-76 SEP 77172  SP ASDIENESA	↑ X 0378T
WPAFB 7-2699 Z AZ19  OCS 3 7080	MO0100/M0010T
30F ES DOSGE 0180 U 002 ACIHIA 76 JUN-76 SEP 77/72	X 0 3 9 0 0
SP ASD/EXESA WPAFB 7-2699 Z AZ62 0053 7080	406L0Z
30F ES DOSGE 0180 W 003  AC141A 76 JUN-76 SEP 77172  SP ASD ENESA  WPAFB 7-2699 Z AZ63  OCS 3 7080	TO 0704   TO 0702
75	

# <u>C1418</u>

30F ES DOSGE 0180 U OOL CI4IA 760CT-77MAY 77171	X 02619
SP ASDIENESA	
WPAFB 7-2679 Z AD97	LO 6106
ocs3 7080	A CONTRACTOR OF THE PROPERTY O
30F ES DOSGE 0180 U 002	X 03100
C141A 76 OCT - 77 HAY 77171	
SP ASDIENESA	
WPAFB 7-2679 7 AEO6	Y06T08
OCS 3 7080	
30F ES DOSGE 0180 U 003	X 0 3 3 8 9
C141A 76 OCT-77 HAY 77171	
SP ASDIENES A	
WPAFB 7-2679 Z AE09	7007TO/70 9777
ocs 3 7080	
30F ES DOSGE 01BO U 004	X 0 3 5 4 7
C141A 760CT-77 HAY 77171	
TSP ASDIENESA	
WPAFIS 7-2679 = BN67	LO 6112/LO6113
OCS 3 7080	NO OILN NOOLLS
7000	

X03558
L06114/L06112
X0357L
roe116/roe112
X00867
MO6118

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/ '	130	_
	100	

INPUT TAPL	OUTPUT THPE	OUTPUT TAPE	
JUH	76 - SEI	076	
X03906	L04548	LO 4135	
X03907	L04122		
DCT	76 - MA	177	
X 03902	LO 1518	L01547	
X03903	L01647	L02574	
X 03904	L03599	L02857	
X 0 3935	L02877		
TAPES WITH D	ARY (CIBOE)		
101518			
L01549			
LO2574			
L02857			
L03626	PHASE II		
L06124	PHASE I		
			78

#### C141A

	C 141 H		
	INPUT TAPE	DUTRYT TAPE	CUTPUT TAFE
	JUNTG	- SEP 70	6
1/3	X03181	106100	406101
213	X 33 700	206102	
3/3	X0390L	206104	LO 6105
	OCTT	6 - MAY 7	1
1/7	X02619	206106	
2/7	X03100	406108	
3/7	X03389	206110	206111
417	X03547	206112	406/13
517	X03558	206114	L06115
617	X 03571	206116	206117
7/7	X 00867	L06118	
	L06100 x L06101 x L06102 x L06103 x L06104 x L06105 x L06106 x	106108X 106110X 106111X 106112X 106112X 106113X 106114X 106115X	LOGIIS - LOGIIS - LOGIIO - LOGIII - LOGIII - LOGIII - LOGIII -

APE USED	DATE GEATED	TAIL NR	TAIL NR	TAIL NR	TAIL NK	+ SAYN NA'.
65948	12 OCT 77	0495	0496	0497	0498	CA - N/A
Lo3845	15 OCT 77	0499	0500	050L	0502	CA - 4
103996	15 OCT 77	0503	0504	0510	0512	CB-8
104359	15 <b>≪</b> T77	0513	0514	0515	0517	cc - 12
L05949	150CT77	0518	0519	0520	0521	CD - 16
Lo5287	17 00777	0523	0524	0525	0526	CA -20
65299	17 OCT 77	0527	0529	0530	0531	CB - 24
L05344	17 OCT 77	0532	0533	0534	0535	cc - 28
65948	17 OCT 77	0537	0538	0539	0540	co -32
L03845	18 OCT 77	0541	0542	0543	0544	CA - 36
10 3996	18 OCT 77	0549	0550	0554	0532	cB - 40
L04359	18 OCT 77	0553	0554	0555	0556	cc - 44
105949	18 OCT 77	0557	0559	0560	0561	CD - 48
105287	19 OCT TI	0562	0564	0565	0566	CA - 52
105299	19 oct 77	0567	0568	0569	0570	cB- 56
105344	19 oct 77	0571	0572	05-4	0934	ec - 60
105948	19 OCT 77	0935	0936	0937	0938	CD- 64
L03845	20 oct 77	0939	0940	0941	0942	CA - 68
L03996	20 OCT 77	0943	0944	0945	0946	CB- 72

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WY USE D	DULL CHAILD	MIL A'X	THIL NR	TAIL GK	IAIL NX	DECK USEB
104359	20 OCT 77	0947	0948	0949	0950	cc - 76
105949	20 OCT 77	0951	1259	1260	126 L	CD - 80
105287	21 OCT 77	1262	1263	1264	1265	CA - 84
105299	21 Oct 77	1266	1267	1268	1269	CB - 88
105344	21 OCT 77	1270	127L	1272	1273	cc - 92
L05948	21 ∝ 77	1274	1275	1276	1288	CD- 96
L03845	22 OCT 77	1289	1290	1291	1292	CA - 100
L03996	22 OCT 77	1293	1294	1295	1296	CB -104
104359	22 OCT 77	1298	1299	1784	1786	ce- 108
L05949	22 OCT 77	1787	1788	1789	1790	CD-112
105287	26 OCT 77	1791	1792	1793	1794	CA - 116
105299	26 OCT 77	1795	1798	1799	18aL	CB-120
L05344	26 OCT 77	1803	1804	1806	1807	CC-124
ь 5948	26 OCT 77	1808	1809	1810	1811	CD-128
Lo3845	26 OCT 77	1812	1816	1817	1818	CA - 132
L03996	26 OCT 77	1819	181_	1820	1821	CB- 136
LO 4359	26 OCT 77	1822	1823	1824	1825	cc - 140
L05949	26 OCT 77	1826	1827	1828	1829	CD- 144
Lo 5287	270777	1830	1832	1833	1834	CE-148

				1
NK USED DATE CHEVILD THE WAY	THIL NA	MIL N'K	TAIL NK	SKIP UR
105299 27 OT 77 1835	1836	1837	1838	CF -152
195344 270cT 17 1839	1842	1843	1844	CG -156
105948 270CT 77 1846	1847	1848	1849	CH - 160
103845 27 00777 1850	1857	1852	1855	CA - 164
103996 2700777 1856	1857	1858	1859	CB - 168
104359 210CT 77 1860	1862	1863	1864	cc - 172
125949 27 OCT 77 1866	2358	2359	2360	CD - 176
105287 28 OCT 17 2361	2362	2363	2364	CE - 180
LOS 299 28 OCT 77 2365	2366	2367	2368	CF - 184
105344 28 OCT 77 2369	2370	2371	2372	CG - 188
105948 28 00777 2373	6566	6579	6580	CH -192
103845 28 OCT 77 658L	6582	6583	7680	CA - 196
193996 28 OCT 77 7681	7764	7765	7766	CB -200
104359 28 OCTTI 7767	7768	7769	7770	CC -204
105949 28 OCT 77 777L	7773	7776	7777	CD -208
105287 29 OCT 77 7778	7779	7781	7782	CE- 212
195299 29 OCT 77 7783	7784	7785	7786	CF 216
105344 29 OCT 17 7787	7788	7790	7791	CG - 220
105948 290CT77 1792	1193 <sub>81</sub>	1794	7795	CH-224

· = Number	DATE GLATEL	PAIL MS.	TAIL NY.	7A71 NA	TAIL NR.	DECKUSED +
105948	21 Dec 77	#229 7794	7771 N.K. 1 1230 7795	#231 7796	A232 7799	CG - 228
1			1 #234 7800			
104933	21 PEC77	#1237 7805	1238 7806	#239 7807	#240 7808	CI - 236
104938	21 DEC 77	#241 7809	A #242 1811	#243 7812	#244 78/3	CJ -240
			#246 7815	1	1	1
			#250 7819			
			- 4254 7823			
105287	21 DEC 71	H257 7820	H258 7828	#259 7829	# 260 7830	co-256
			1 #262 7832			
			85 AZLL 7836		1	,
	t c		9 #270 7840	The second secon		
			15 #274 7846			
LO 4933	2100077	4277 784	9 #278 7850	H279 7851	#280 7852	cI -276
L04938	21 Dec 77	#281 785	3 #282 7854	#283 7856	#284 7857	CJ-280
103845	22 DEC 77	#285 785	18 4286 7859	#287 7860	#288 7861	CA - 284
LO 3996	22 Dec 77	#289 786	3 #290 7864	#291 7865	#292 7866	CB - 288
104359	22 Dec 77	NA 786	7 1214 7868	#215 7869	#2× 7871	CC - 292
(1) 18'	1 de 11	· 41 151	12 min 15/4	+214 1876	1300 7877	CD-296
105299	22 DEC 7	7 #34 787	9 #302 7880	#303 788L	#304 7882	CE - 300

	1		!			
		AL MY		1	TAIL NR.	LAIP N-MUER
05344	22 Dec 77	1305 7883	H306 7884	1307 7885	4308 7887	CF - 304
105948	22 Dec 77	N309 7888	#310 7889	H311 7890	#312 789L	c6 - 300
105949	22 DEC 77	#313 7892	1893	#315 7894	#316 7895	CH - 312
104933	22 Dec 77	1317 7896	new 7897	134 7898	130 7899	CI - 316
L04938	22 Dec 77	1321 78-4	1302 7-1-	1323 7-42	P24 7-46	c5 -320
10 3845	22 Dec 77	H325 7-66	1326 8240	1327 9810	H318 9811	CA - 324
Lo 39%	22 DEC 77	1929 9812	130 9813	7814 VES	1332 9815	CB - 328
10 4359	23 DEC 77	#333 9816	H334 9817	7995 9990	H33 9999	cc - 332
105281	22 DE 77	#337 9-99	W338 -806	#359 -828	#340 - 831	co - 336
L05299	22 Dec 77	#341 - 837	1342 _85L			CE - 340
					•	
			•		•	
				1		1

#### C. 1111 ( CC 1 10 11/17/11)

IN NUMBER WITE CREATED TAIL HIR	TAIL NA	TAIL MR	TAIL UR.	DICK USED Y
L05344 22 DECTT #1 7013	#2 000L	73 0002	A4 0003	CF - 0
LO 3845 23 DEC 77 15 0004	#6 0005	#7 0006	#8 0007	CA - 4
L03996 23 PEC 77 119 0009	#10 0010	411 OOLL	#12 0012	CB - 8
L04359 23 DEC 77 #13 0013	#14 0015	415 0017	#16 0018	cc - 12
LOS287 23 DEC 77 AIT 0019	#18 0020	#19 0021	#20 0022	CD = 16
LOSA99 23 DEC 77 #21 0023	#12 0024	#23 0025	#24 0026	CE - 20
LO5344 23 DEC 77 #25 0027	#26 0029	427 0030	#28 0031	CF - 24
105948 28 DEC 77 #29 0128	#30 0129	1131 0130	132 0132	ca - 28
L05949 23 DEC 77 133 0133	#34 0137	N35 0138	436 0139	CH + 32
L04933 23 DEC 77 137 0140	#38 014L	139 0142	140 0143	CI -36
L04938 23 DEC 77 141 0145	My2 0146	#45 0147	MY4 0148	CJ - 40
L03810 28 DEC 77 #45 0149	## 0150	847 0151	#48 0153	CA 44
L03996 28 DEC 77 #49 0154	1150 0155	#ss 0156	#52 0157	CB 48
LO 4359 27 DEC 77 #53 0158	1154 0159	H55-0160	#52 0161	cc ~ 52
LO 5287 27 DEC 77 #57 0162	1158 0164	#59 0165	#60 0166	CD - 5%
LO 5299 27 DEC 77 16 0170	#62 0171	#63 0/72	#4 0173.	CE - 60
195344 27 DEC 77 #15 0176	#66 0177	#67 0178	#68 0179	CF - 64
105948 27 DEC 77 #69 0180	A70 0181	471 0182	*72 0/83	CG - 68
105949 27 DEC 77 #73 0184	#74 0185	115-0188	#76 0189	CH - 72

The state P. Darre II are A 77 11 141	-17 1/.)	-TA () ( ( )	73.11 1/10	DICK USED +
104933 27 DEC 77 #17 0190	478 0191	TAIL NA. #79 0192	180 0193	CT - 76
14938 27 DEC 77 #8 0194	# 0197	183 0/98	184 0200	CT -80
13810 27DEC 77 185 0201		187 0206		
103 996 27 Dec 77 119 0208	190 0216	#91 0217	492 0219	CB '- 88
124359 28 DEC 77 193 0221	#4 0223	195 0224	#96 0225	cc - 92
195287 28 Dec 77 #91 0226	198 0227	199 0228	1100 0229	CD = 96
105299 27 DEC 77 #101 0230	#102 0231	#103 0232	BAY 0233	CE -/00
105344 27 DEC 77 #105 0234	#106 0235	#107 0236	#108 0237	CF -104
15948 27 DECT7 #109 0238	#110 0239	*111 0240	#112 0241	CG -108
105949 27 DEC 77 #113 0242			1	
104933 27 DEC TT #117 0246	#118 0247	#119 0248	#120 0249	CI -116
104938 27 DECTT 144 0250	#122 0251	M123 0252	#44 0253	CJ = 120
103810 28 DEC77 #115 0254	mac 0255	#127 0256	nus 0257	CA - 124
123996 28 DEC 77 May 0258	#130 0259	MBI 0260	#BY 0261	CB = 128
104359 28 DEC 77 #133 0262	P134 0263	1135 0264	MX 0265	CC -132
10 5287 28 Dec 77 #137 0268	W138 0271	#139 0275	440 0276	. CD - 136
165299 28 DEC 77 #141 0277	My2 0278	#143 0280	#14 0609	CE - 140
10 5344 28 Dec 77 #145 0611	MHL 0614	#147 0616	1148 O617	CF -144
105948 28 DECTT #149 0619	#150 0620	4151 0621	#152 0627	- CG = 148

E NUMBER	DATE GLAVED TAIL MR	TAIL NA	TAIL NA	TAIL NR	DICK USED +
105949	28 DEC 77 #153 0623	#154 0625	*155 0626	#156 0627	CH -152
	28 DEC 77 #157 0628				
104938	28 DEC 77 +141 0635	#162 0636	#163 0637	ML4 0638	CJ - 160
103810	30 DEC 77 #165 0639	mu 0640	#167 0642	nus 0643	CA - 164
43996	29 DEC 77 1119 0644	#170 0645	N171 0646	More 0647	CB = 168
	28 DEC 77 #173 0648				1
105287	29 DEC 77 #177 0653	#178 2775	#179 2T16	#180 2777	CD -176
195299	28 DEC 77 1181 2778	#182 2779	#183 4610	M184 4611	CE = 180
105344	28 DEC 77 #185 4612	#186 4613	W87 4615	MISS 4618	CF -184
105948	28 DEC 77 #189 4624	#140 4629	#191 4630	*m 4631	CG -188
105949	29 DEC 77 #193 4633	#174 4639	1195 4642	** 4644	CH -192
L04933	29 De 77 1197 4647	#198 465L	#199 5217	#200 5218	CI -196
104938	28 DEC 77 1001 5220	1202 5222	#2°3 5265	1204 5266	CJ -200
103810	29 DE 77 125 5267	N206 5269	#209 5270	11208 5272	CA - 204
103996	29 DEC 77 #209 5273	#210 5279	Hall 6126	#22 613L	CB -208
104359	29 DEC77 #213 6134	HALY 6135	#215 6/36	Hall 6137.	cc -2/2
105287	29 DEC77 #217 6144	#218 6152	1219 6162	Maso 6163	CD-2/6
105299	29 DEC 77 #24 6164	#222 6/66	#223 6/67	M224 6/68	CE - 220
105344	29 DET 77 #25 6169	#224 6174 86	man 6175	#228 6186	CF - 224

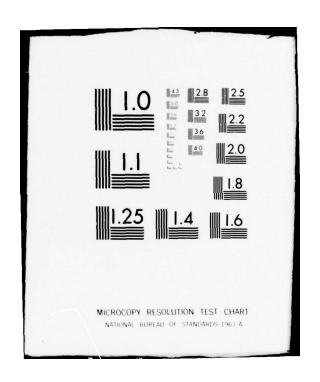
			DICK HOLDS
105948 29 DE 77 4239	6187 430 6195	7AIL AR 7AIL NR 1232 6199	CG -228
			1
105949 29 DET 77 *28	6202 May 6203	#235 6204 #28 6207	CH -232
104933 29 DE 77 1007	1		
104938 29 DETT #41	7016 4242 7028	1243 7944 1124 7945	CJ - 240
103810 30 PEC 77 MASS			1
103996 30 DEC 77 #249	7950 Naso 7957	#251 7952 #252 7953	CB -248
104359 29 DEC 77 1253			
'05287 29 DEC 77 #267			1
'05299 29 DEC 77 17261	8078 #262 8079	4213 8080 H264 808L	CE - 260
105344 29 DEC 77 10265	8082 424 8083	127 8084 HX8 8085	CF - 264
05948 29 DEC 77 0269	8086 +270 8087	1271 8088 HOT - 8089	CG-268
105949 29 DEC 77 #273	8090 AZH 9090	#275 9397 #276 9398	CH 4272
'04933 29 DEC 717 #277	9399 4278 9400	11279 940L 11280 9402	CI - 276
194938 30 DEC 77 +281	9403 #282 9404	1283 9405 1284 9406	CJ - 200
'13810 29 DECTT #285	9408 4286 9409	#287 9410 #288 9411	CA - 284
103996 29 DEC 77 #289	9412 #290 9413	#291 9414 #292 9990	CB-288
'0 4359 29 DECTT #29	<b>9999</b> 87		cc -292

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### 7 TRACK TAPE CREATE (STEP II)

11 Thepe	Intal Wif	LOUIS TONE	Input Trek	Corper Tys	DITTILIT TOPS	anjur
C13	0E - J	LN76 7	HRU SE	P76		
1-04548	L04135	104122		L04220		
C13	DE - De	T 76 77	RUL MAY	77		
1.01518	1.01549	L01647	L02574			
L02579	102857	L02877		L06119		
					1	
<u>C1</u>	HA - J	UN 76-	THRU SE	P76		
.06100 L06105	L0610L	106102	L06104	ž		
200103				! ! !	:	
				1		
CI	41A - C	CT 76	THRU H	MY 77	•	
106106			r06112		L06119	
LU6116	100711	10 6778			106120	
						-
			88	V 700		





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30F ES DOSGE 0180 W OOL	
C141A 76 OCT- 77 MAY 77171	VA21-10
SP ASD/ENESA 27 cm	X02619
NPAFB 7-2679 Z AD97	103,848/203381
OCS 3 7080 10880	3466 Alacho 168 Bloc
30F ES DOSGE 0180 U 002	
C141A 76 OCT-77 HAY 77171	VATIAN
SP ASDIENESA	X03100
WPAFB 7-2679 Z AEOG	L02951 / L00812
OCS 3 7080 04675	3197 Mach, 4376 Coche
30F ES DOSGE 0180 W 003	And the second s
CIMIR TO DOT- 17 HOU -	VATTOO
C141A 76 OCT-77 HAY 77171	X03389
SP ASDIENESA	
UPAFB 7-2679 Z AE09	
OCS3 7080 0451'i	
24 - 2	
30F ES DUSGE 0180 11 004	
C14/A 76 OCT - 77 MAY 77171	X03547
T SP ASD/ENESA	10934
OPAFB 7-2679 Z BN67	
OCS 3 7080 01130	
89	

## CI41A Page 2 OF 2 PAGES FIREM NUMBER = E740052

	30F ES DOSGE 0180 U 005
X03558	C141A. 76 OCT-77 HAY 77171
X03558	T SP ASD/ENESA
70000	WPAFB 7-2679 Z BN90
	OCS 3 7080
	30F ES DOSGE 0180 U 006
MARCAI	C141A 76 OCT - 77 MAY 77171
X035 11	T SP ASD/ENESA
1400	30F ES DOSGE 0180 W 006.  C141A 76 OCT - 77 MAY 77171  T SP ASD/ENESA  WPAFB 7-2679 Z BF 48  OCS 3 7000
	OCS 3 7080
	245 56 2475 44
	301- ES DOSGE 0180 W 007
X AAOL 7	C141A 76 OCT-77 MAY 77171
110000	SP ASD/ENESA
	30F ES DOSGE 0180 W 007.  C141A 76 OCT-77 HAY 77171  SP ASD/ENESA  WPAFB 7-2679 Z BF49
	OCS 3 7080

## AC141A- PAge 1 OF 1 PA BLEM NUMBER = E740052

30F ES DOSGE 0180 W <u>OOL</u> AC141A 76 JUN-76 SEP 77172 SP ASD/ENESA 2'/ 'UPAFB 7-2699 Z AZ19 OCS 3 7080 0486	X03181
30F ES DOSGE 0180 U 002 AC141A 76 JUN-76 SEP 77172 SP ASD/ENESA 'UPAFB 7-2699 Z AZ62 OCS3 7080 50791	X03900
30F ES DOSGE 0180 W 003 AC141A 76 JUN-76 SEP 77172 SP ASD/ENESA WPAFB 7-2699 Z AZ63 -0C53 7080 03993	X0390L
EQCLAS DATA	

# C.130E Page L OF 1 PROBLEY NUMBER = E740052

	20F ES DOSGE 0180 U 001
X03902	C130E: 76 OCT - 77 MAY 77171
X1340K	SP ASD/ENESA
11001	1PAFB 7-2679 Z AC66
	OC\$ 3 7080 00787
	LOF ES DOSGE OIBO U OOR
X03903	CIBOE 76 OCT-77 MAY 77171 SP ASDIENESA
VOJ IOS	SP ASD/ENESA
	1PAFB 7-2679 Z AD31
	ocs 3 7080 30492
	20F ES DOSGE 01BO W 003
X03904	C130E 76 OCT-77 MAY 77171
NUDIUT	SP ASD/ENESA
	UPAFB 7-2679 Z AD91
	ocs 3 7080 31380
	20F ES DOSGE 0180 U 004
V NAGAK	C130E 76 OCT-77 MAY 77171
X03905	T SP ASD/ENESA
	WPAFB 7-2679 Z BF54
	OCS3 7080 19559

## AC130E Page LOF L PROLLETT Number = E740052

20F ES DOSGE 0180 W 00L  AC130E 76 JUN - 76 SEP 77172  SP ASD/ENESA CT 3  VPAFB 7-270L Z AZ74  OCS 3 7080 03776	X03906 204548/204135-X
10F ES DOSGE 0180 W 002 AC130E 76 JUN-76 SEP 77172 SP ASD/ENESA 7 VPAFB 7-2701 Z AZ80 0C53 7080 15123	X03907

### CIOUR WUN-SEP 16)

NK CUED	MITE CHENTED	THIL NR	TAIL NK	THIL NR	THILNE	ELEK UZEBA SKIP NR
W3845	31 OCT 77	7196	7799	1800	7803	CA - 228
L03996	31 OCT 77	7804	7805	7806	1807	CB -232
104359	31 OCT 77	7808	7809	7811	1812	ce-236
105949	31 OCT 77	1813	7814	1815	7816	CD- 240
105287	1 Nov 77	7817	7818	7819	7820	CE - 244
105299	1 HOV77	1821	1822	1823	7824	CF -248
105344	1 Hov77	7825	1826	7828	7829	CG-252
105948	1 Nov77	782_	7830	783L	7832	CH - 256
103845	1 Nov 77	7833	7834	7835	7836	CA - 260
103996	1 NOV 77	1837	7838	7839	7840	CB - 264
104359	1 Nov 77	7841	7842	1845	1846	cc - 268
LO 5949	L HON 77	7847	7848	7849	7850	CD- 272
105287	2 Nov 77	7851	7852	7853	7854	CE- 276
105299	2N0V77	1856	7857	7858	7859	CF - 280
	2 16477	7860	7861	7863	7884	CG-284
	2 Nov 77	7865	7866	7867	1868	CH- 288
	2 Nov77	7869	7871	7872	1874	CA - 292
	2 NOV 77	7876	7877	1879	7880	CB- 296
	2 NOVIT	7881	788294	7883	7884	cc - 300

### C130E (JUN - SEP 76)

" NA JAZ	PAIR KINIL	THE WA	mi Ne	THIL NX	THIL NK	SKIP IIK
. 15949	2 NOV 77	7885	7887	7888	1889	00 -304
· 25287	3 Nov 77	1890	1891	7892	1893	CE - 308
05299	3 NOV 77	7894	7895	7896	1897	CF - 312
15344	3 NOV 77	7898	7899	8240	9810	CG- 316
15948	3 NOV 77	9811	9812	9813	9814	CH - 320
3845	4 NOV 77	9815	9816	9817	9990	CA - 324
3996	4 NOV T7	9999	€872	_	-	CB -328
	C130E	( TUN 4 NOV	-SEP 76 77	> COMPA	ETED	
•			95			

# C141A (JUN-SEP76)

Le us Han	no. (2					Doze III.
L05299	5 NOV 77	OOOL	000Z	0003	0004	CE - 0
L05344	5 Nev 77	0005	0006	0007	0008	CF - 4
L05948	5 Nov 77	0009	0010	0011	0012	ca - 8
L05949 .		0013	0015	0017	0018	CH - 12
L03845	5 Nov 77	0019	0020	0021	0022	CA - 16
L03996	5 NOV 77	0023	0024	0025	0026	CB - 20
L04359	5 NEV 77	0027	0029	0030	003 L	cc - 24
L05287	5 Nov 77	0/28	0/29	0130	0132	CD - 28
L05299	7 Nov 77	0133	0138	0139	0140	CE - 32
L05344	7 Nov 77	0141	0142	0143	0145	CF - 36
L05948		0146	0147	0148	0149	CG - 40
L05949	7 16477	0150	0151	0153	0154	CH - 44
LO 3845	7 Nov 77	0155	0156	0157	0158	CA - 48.
L03996	7 Nov 77	0159	0160	0161	0164	CB - 52.
L04359	7 NOV 77	0165	0166	0170	0/7/	cc - 56.
L05287	7 NOV 77	0172	0173	0176	0177	CD- 60.
105299		0178	0179	0180	0181	CE- 64
LO 5344	8 NOV 77	0182	0183	0184	0185	CF- 68
LU5948	8 NOV. 77	0188	0189 96	0190	0191	(G - 72

Noweth	DAIL (KLAILD	TAIL N'S	TAIL NE	TAIL N'S	INIL NA	SKIP NR.
5949	8 NOV 77	0192	0193	0194	0197	CH - 76
3845	9 NOV 77	0198	0200	0201	0205	CA - 80
3996	9 NOV 77	0206	0207	0208	0216	CB - 84
4359	9 NOV 77	0219	0221	0223	0224	cc - 88
5287	9 NOV 77	0225	0226	0227	0228	CD - 92
25299	10 NOV77	0229	0230	0231	0232	CE- 96
3 <i>5344</i>	10 Nov 77	0233	0234	0235	0236	CF - 100
15948	10 Nov 77	0237	0238	0239	0240	CG - 104
15949	10 NOV 77	0241	0242	0243	0244	CH - 108
03845	11 NOV 77	0245	0246	0247	0248	CA - 112
3996	11 Nov 77	0249	0250	0251	0252	CB- 116
14359	11 Nov 77	0253	0254	0255	0256	cc - 120
05287	11 NOV 77	0257	0258	0259	0260	CD- 124
05299	12 164 77	0261	0262	0263	0264	CE - 128
. 05344	12 NOV 77	0265	0268	0271	0275	CF - 132
. 55948	12 NOV 77	0276	0277	0278	0280	CG - 136
. 05949	12Nov77	0609	0614	0616	0617	CH- 140
. 03845	14 NOV 77	0619	0620	0621	0622	CA- 144
03996	14 NOV 77	0623	0625 97	0626	0627	CB- 148

1 Nomber 104359	DATE CHEATED	0628	0632	7012 ACC 0634	1411 NR. 0635	DECR' 456 5 + SKIP NR. CC - 152
	14 Nov77	0636	0637	0638	0639	CD- 156
105299	14 NW 77	0640	0642	0643	0645	CE - 160
LO 5344	14 Nov 77	0646	0647	0648	0649	CF - 164
	14 Nov 77	0650	0652	0653	2775	CG- 168
10 5949	14 NOV77	2776	2777	2778	2779	CH - 172
L03845	15 Nov 77	4610	4611	4612	4613	CA - 176
103996	15 Nov 77	4615	4618	4624	4629	CB - 180
104359	15 Nov 77	4630	4631	4633	4644	cc - 184
105287	15 Nov 77	4652	5217	5218	5220	CD - 188
105299	15 Nov77	5222	5266	5267	5269	CE- 192
L05344	15 Nov 77	5270	5272	5273	5279	CF - 196
L05948	15 Nov 77	6126	6131	6/34	6/35	CG - 200
	15 Nov 77		6137	6144	6152	CH - 204
L03845	15 Nov 77	6162	6/63	6164	6167	CA - 208
103996	15 Nov 77	6168	6169	6174	675	CB - 21:
L04359	15 Nov77	6186	6187	6/93	6194	cc - 216
105287	15 Nov 77	6195	6196	6199	6202	CD - 221
205299	15 NOV77	6203	6204	6205	6209	CE - 224

15344 15 Nov 77	WIL MS	7016	7028	TAIL NR	DECK USED +
	7014	7016	7028	7944	CF - 228
,5948 15 NOV TT	7945	7946	7947	7948	CG -232
25949 15 NOV 77	7949	7950	795Z	7952	CH - 236
13845 16 NOV77	1953	7954	7955	7956	CA - 240
13996 16 NOV77	7957	7958	1959	8075	CB - 244
14359 16 NOV 77	8076	8078	8079	8080	cc - 248
5287 16 NOV77	8087	8082	8083	3084	CD- 252
5299 17 NOV77	8085	8086	8087	8088	CE -256
5344 17 NOV 77	8089	8090	9397	9398	CF-260
5948 17 NOV77	9399	9400	9401	9402	CG-264
15949 17 Nov 17	9403	9404	9405	9406	CH-268
03845 17 NOV 77	9408	9409	9410	9411	CA - 272
13996 17 NOV.77	9412	9413	9414	9990	CB - 276
15287 17 Nov77	9999				CD-280

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103996	16 DEC 77	#9 0503	#10 0504	#11 0570	#12 0512	CB - 8
L04359	16 DEC 77	#13 0513	#14 0514	#15- 0575	#11. 0517	cc - 12
		#17 0518				
		#21 0523				1
L05344	16 DEC77	N25 0527	#26 0529	N27 0530	#28 0531	CF - 24
105948	16 DEC 77	#29 0532	#30 0533	#31 0534	#32 0535	CG - 28
105949	16 DEC 77	#33 0537	H34 0538	#35 0539	M36 0540	CH - 32
LO 4933	16 DEC 77	#37 0541	#38 0542	139 0543	N40 0544	CI - 36
104938	16 DEC 77	#41 0549	H42 0550	#43 055L	#44 0552	cJ - 40
L03845	16 DEC 77	N45 0553	N46 0554	#47 0555	148 0556	CA - 44
L03996	16 DEC 77	#49 0557	ASO 0559	#51 0560	N5-2 056 L	CB - 48
L04359	16 DEC 77	#53 0562	#54 0564	NSS 0565	#56 056	cc - 52 .
L05287	16 DEC 77	#57 0567	A58 0568	1159 0569	#60 0570	CD - 56
L05299	16 DEC 77	#61 0571	#62 0572	#63 05-4	#4 0934.	CE - 60
L05344	16 Dec 77	#65 0935	#66 0936	#67 0937	#68 0938	CF - 64
L05948	16 DEC 77	#69 6939	#20 0940	#71 0941	#72 0942	CG - 68
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03996						CB - 128
04359	19 Dec 77	#133 1810	#134 1811	#135 1812	#136 1816	CC - 132
05287	19 DEC 77	#137 1817	#138 1818	#139 1819	#140 1820	CD - 136
05299	19 DEC77	#141 /821	HM2 1822	#143 1823	#144 /824	CE - 140
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			#178 1866			
105299	20 DEC 77	#181 2359	#182 2360	#183 2361	#184 2362	CE - 180
			#186 2364	1		
			#190 2368	1		
			#194 2372			
			HAS 6580			
			#202 7680			1
			#206 7766			
103996	21 Dec 77	#209 7769	*20 7770	#211 7771	Fa12 7773	CB- 208
604359	21 DEC 77	#213 7776	M24 7777	1215 7778	Have 1779	cc - 212
LO 5287	21 DEC 77	#217 7781	#218 7782	1219 7783	#220 7784	CD- 216
105299	21 Dec 77	#221 7785	#222 7786	#223 7787	#224 7788	CE- 220
L05344	21 DEC 77	#225 7790	#226 <b>779</b> 1	#227 7792	#221 7793	CF- 224

10

STATE OF SECTION

APPENDIX B

Selected Reference Material

## Chapter 9

## MAINTENANCE EXPERIENCE DATA (AFM 66-1)

- 9-1. PURPOSE. The purpose of this chapter is to outline the Maintenance Data Collection (MDC) system established by AFR 66-14 and AFM 66-1. The MDC is the primary source for Air Force reliability and maintainability data; therefore, basic understanding of its objectives, uses, and limitations is essential to R & M data users.
- 9-2. OBJECTIVES. The Maintenance Data Collection system was designed primarily as a base level production credit and management information system. The objectives are to provide maintenance managers with information about the production accomplished by the assigned maintenance personnel; and to identify the equipment on which work was accomplished, why the work was required, and the action required to complete the job. The MDC system identifies maintenance requirements and problem areas so that appropriate management action can be taken to effectively support and meet the established operational requirements. In addition, the MDC system is designed to provide data to AFLC for maintenance engineering and logistics management. Selected data is also provided to the major commands and HQ USAF in accordance with AFMs 66-267 and 66-271.
- 9-3. SCOPE. The MDC system is applicable to all functions outlined in AFM 66-1, and requires that all maintenance actions involving direct labor expenditures be recorded and reported in this system unless exempted by TO 00-20-2. The system is applicable to the life cycle of aircraft, missiles, ground communications, electronics, and meteorological equipment, and related end items beginning with operational test and evaluation as described in AFR 80-14. This includes compatible data reporting on contractor maintained equipment and maintenance accomplished in depot facilities.
- 9-4 DOCUMENTATION CONCEPT. The AFTO Forms 346, 349, and 350 are used as source documents for the maintenance data collection system.
- a. Recording Concept procedures are divided into two basic categories identified as on-equipment and off-equipment maintenance documentation.
- (1) Maintenance actions accomplished on complete end items of equipment (aircraft, missiles, removed engines, ground communications-electronics-meteorological (CEM), trainers, Aerospace Ground Equipment (AGE) and nuclear weapons) are identified as on-equipment work. This primarily consists of support general tasks, inspections, removal and replacement of components, fix-in-place maintenance actions, and modifications.
- (2) In-shop maintenance actions involving intermediate level maintenance on removed components is identified as off-equipment maintenance. This primarily consists of bench check, repair or modification of components and assemblies, and nondestructive inspection.

- (3) If maintenance is done on components that are removed or removed and replaced to facilitate maintenance in the same room or one immediately adjacent to the end item; this is recorded as onequipment maintenance. If the individual that removed the component has to leave the immediate area (defined as out-of-sight), an AFTO Form 350 will be prepared to identify the status of the removed component. In this regard, when personnel from one workcenter remove an item and send it to personnel with a different workcenter code for maintenance, the latter workcenter will record it as off-equipment maintenance.
- (4) Due to management requirements, unique procedures exist for engines. All maintenance accomplished on gas turbine and reciprocating engines installed in aircraft, missiles, or AGE will be recorded as on-equipment maintenance. Removal and replacement of gas turbine and reciprocating engines for aircraft, missiles, or AGE will be recorded as on-equipment maintenance with the engine treated as a component. Shop work on all removed gas turbine engines and aircraft reciprocating engines will be treated as end item maintenance with on-equipment and off-equipment recording concepts applying. (TO 00-20-2-4). Shop work on reciprocating engines removed from AGE will be treated as component maintenance and the off-equipment maintenance concept will apply.
- (5) Each workcenter participating in a job will record maintenance actions and labor expenditures. The documentation responsibility rests with the senior representative from the workcenter. These documents will be returned to the workcenter supervisor who will check for accuracy and completeness prior to submission for processing.

#### b. Data Forms:

- (1) Use of the AFTO Form 349. The AFTO Form 349, "Maintenance Data Collection Record." was designed with sufficient flexibility for use by the majority of organizations in recording maintenance actions on various types of equipment. Recording and data collection procedures pertaining to this form are outlined in the 00-20-2-series technical orders.
- (a) For on-equipment work the primary entries required on the AFTO Form 349 are block 1 (Job Control Number), block 2 (Workcenter), block 3 (ID Number), block 6 (Time, as applicable), and columns B through K. For in-shop engine work, primary entries are required in blocks 1 and 2, block 3 (Engine ID) and in columns B through K. For off-equipment work on removed components, primary entries are required in blocks 1, 2, and block 3 or 5; block 19 (Federal Supply Class (FSC)), block 20 (Part Number), and columns B through K.
- (b) Up to five related on-equipment maintenance actions covered by a single job control number against a single ID number, and accomplished by a single workcenter may be reported on a single copy

of the AFTO Form 349. If more action lines are required, another AFTO Form 349 containing the same job control number, ID number and workcenter code is completed and the actions continued. This recording procedure also applies to off-equipment actions; however, on-equipment and off-equipment actions will not be combined on a single copy of the AFTO Form 349. The four items could be reported by a single line entry if the job control number, work unit, action taken, how malfunctioned and when discovered codes are all the same and a unit count of four is entered. Similarly the AFTO Forms 350 prepared for shop processing of the four black boxes may reflect a quantity of more than one only if the job control number, work unit code, federal supply class and part number are the same. If these elements are different, a separate AFTO Form 350 must be prepared for each item. Serially controlled and time change items (with an asterisk in the work unit code manual) must be recorded on an individual basis, (for example, only one item per AFTO Forms 349 and 350).

- (c) The AFTO Form 349 can be used for identification of both the end item of equipment and a component for engine change actions, for weapon systems and equipment that are managed under the Advanced Configuration Management System (ACMS), for time change items, for special reporting on tires, and for reporting off-equipment maintenance actions
- (2) Use of the AFTO Form 350. The AFTO Form 350, "Reparable Item Processing Tag," is a two-part perforated form that is attached to components that are removed from equipment end items and serves as an identification and status tag. Another important aspect of this form is that it serves as a source document pertaining to Repaired This Station (RTS), Not Repaired This Station (NRTS), and condemnation actions for the supply system. This information is input to the base supply computer to identify stockage requirements. Information pertaining to RTS, NRTS, and condemnations is also forwarded through the supply system to AFLC as factors for computing the world-wide spares requirements. Recording procedures for the AFTO Form 350 are outlined in the 00-20-2-series technical orders.
- (3) Use of the AFTO Form 346, "Maintenance Data Collection Production and Scheduling Record." The AFTO Form 346 is used for scheduling the calibration of Precision Measuring Equipment (PME) and for recording all maintenance on precision measurement equipment for input to the MDC system. The AFTO Form 346 may also be used for scheduling calendar maintenance requirements on any equipment within the maintenance complex. Note that this pertains only to calendar requirements. Scheduling procedures pertaining to the AFTO Form 346 are outlined in AFR 66-267. Maintenance recording procedures for the AFTO Form 346 on PME are outlined in TO 00-20-10-6

#### c. Data Elements:

(1) Job Control Number (JCN). The JCN consists of seven characters, the first three are the Julian date and the last four are a unique job number for that date. This provides a means to tie together all on- and off-equipment actions taken, man-hours

expended, and parts consumed to satisfy a maintenance requirement whether it be a discrepancy, an inspection, or a TCTO. Every action taken that is related to a job, regardless of workcenter, time or place, will carry the same job control number that was originally assigned to the job. This procedure is necessary to permit control of all related actions, and to provide the capability to tie them together in data systems to identify the total job for analysis purposes.

- (2) Workcenter Code. The workcenter code consists of five characters that identify organizational elements to which maintenance personnel are assigned, or locations to which they may be dispatched. Standard workcenter codes are used by all organizations engaged in the maintenance functions outlined in AFM 66-1. In general, the code entered in the workcenter block of the AFTO Form 349 indicates the workcenter of the individual doing the work and not necessarily where the work is accomplished.
- (3) Identification (ID) Number. The ID number consists of six characters, and is used to identify equipment on which work was performed or from which an item was removed. The first character of the ID number is the first character of the aworkcenter code. The second character of the ID number is the first character (prefix) of the equipment classification code such as A for aircraft, B for Ground Radar or M for Ground Launched Missiles (AFM 300-4, ADE MA-156-XI). The last four characters of the ID number normally are the last four positions of the equipment serial number. Detailed procedures for assigning ID numbers are contained in AFM 66-267.
- (4) Equipment Classification Code. The equipment classification code consists of three characters, and is assigned to identify aircraft, missiles, ground communications, electronics, and meteorological equipment, AGE, trainers, engines, ground launched missile real property installed equipment, munitions, and precision measurement equipment. Codes are also assigned for research and development and shop work. Most of the equipment classification codes are assigned to specific equipment such as LGM-30B missiles. Some of the codes are assigned by category of equipment or work such as non-registered AGE and shop work that is not identified to a weapon or support system. The authorized equipment classification codes are contained in TO 00-20-2.
- (5) Type Maintenance Code. The type maintenance code consists of one character and is used to identify the type of work that was accomplished such as scheduled or unscheduled maintenance. Type maintenance codes are listed in each work unit code manual for individual types of equipment. A complete list of authorized type maintenance codes is contained in AFM 300-4, volume XI.
- (6) Work Unit Code. The work unit code consists of five characters, and is used to identify the system, subsystem, and component on which maintenance is required or on which maintenance was accomplished. These codes are published in work unit code manuals for each weapon and support system and in code manuals by type of equipment for selected ground CEM, trainers, AGE, munitions, PME,

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and shop work. A limited number of work unit codes are assigned in a special category to identify tasks of a general nature such as equipment servicing, cleaning, inspection, storage, ground safety, record keeping, weapons handling, and repetitive shop tasks. Although they are work unit codes, they are identified as "Support General Codes." The first two positions of the work unit codes for aircraft, ground radar, and missiles are standard system codes. They identify functional systems such as flight control system, codes antenna system, or launch control system. The first two positions of the work unit codes for support equipment identify types of equipment, such as ground powered generators, or end items of equipment, such as a trainer. The first position of support general codes begin with a zero; and this is standard in all work unit code manuals. The third and fourth positions of the work unit code identify subsystem or major assembly. The fifth position of the work unit code normally identifies reparable

- (7) Units Completed. The work unit code in combination with an action taken code is used to describe a "unit of work." An entry of one or more units completed must also be made in the UNITS block of the data collection form in order to show a completed action. An example of a unit of work would be a work unit code for an antenna, with an action take, code for removed and replaced, and a unit count of one, for example, one antenna removed and replaced. By using additional codes to identify the end item, the type of maintenance being accomplished, when the maintenance requirement was discovered, how the item malfunctioned, and the time expended in accomplishing the work, and the production credit system also provides information essential for maintenance and materiel management.
- (8) Action Taken Code. The action taken code consists of one character used to identify the maintenance action that was taken, such as remove and replace. Action taken codes are standard for all equipment and are listed in all work unit code manuals. A complete list of authorized action taken codes is contained in AFM 300-4, volume XI.
- (9) When Discovered Code. The when discovered code consists of one character and is used to identify when a defect or maintenance requirement was discovered, such as during a quality control inspection. When discovered codes are listed in each work unit code manual for individual types of equipment. A complete list of authorized when discovered codes is contained in AFM 300-4, volume XI. Only that portion of the when discovered code definition that applies to equipment listed in the work unit code manual is to be used. For example, when discovered code D, In-Flight-No Abort/During AGE Operation, would be listed in the AGE work unit code manual as D, During AGE Operation.
- (10) How Malfunctioned Code. The how malfunctioned code consists of three characters and is used to identify how the equipment malfunctioned, such as cracked. To provide maximum utility, these codes are also used to identify time compliance technical order status requirements, or to show that a maintenance action did not result from a defect. A complete list of authorized how malfunctioned codes is contained in AFM 300-4, volume XI, in both

alphabetical (definition) and numerical (code) sequence. Only those how malfunctioned codes that are applicable will be listed in each work unit code manual. For example, how malfunctioned codes applicable only to a solid rocket missile will not be listed in a ground CEM work unit code manual.

Note: Due to the nature of support type work, the recording of action taken, when discovered, and how malfunctioned codes is not required with support general work unit codes.

- 9-5. The foregoing paragraphs of this chapter describe the MDCS objectives and reporting concept as related to the base maintenance environment. In order to provide AFLC data on maintenance events as they occur worldwide, most of the data documented at AF bases under the TO 00-20-2 series are submitted to HQ AFLC for use in logistic support and related engineering decisions. These data are received and processed centrally at HQ AFLC in the DO56 Product Performance System. This data system not only receives and output reports containing Reliability and Maintainability (R and M) factors within its established computer programs but also services other interfacing data systems with source data. Some of the interfacing data systems also output reports containing R and M factors individually unique to their established computer program controls. Figures 9-1 through 9-18 illustrate the data flow from point of origin through the DO56 major system processes and to other interfacing systems which are driven by the same source data. The following pages of this chapter explain some of the erms used in the DO56 and samples of output eports containing R and M factors; however, for a full understanding of system capabilities refer to AFLCM 66-15 and 171-45,
- 9-6. Definitions of R and M parameters and terms used in the DO56 data system:

a. Type How Malfunctioned Codes.

- (1) Type 1-These codes indicate that the item no longer can meet the minimum specified performance requirement due to its own internal failure pattern.
- (2) Type 2-These codes indicate that the item can no longer meet the specified performance requirement due to some induced condition and not due to its own internal failure pattern.
- (3) Type 6-These codes indicate maintenance resources were expended due to policy, modifications, items location, cannibalization and other no defect conditions existing at the time maintenance was accomplished.
- b. Failure occurrences. The computer definition of a failure occurrence related to a Work Unit Code is: "any Type 1 How Malfunctioned code reported in combination with an action taken indicating repair, adjustment or item replacement and one or more inits produced.
- c. Quantity per Application (QPA). This is the quantity of identical installed items on a single unit of equipment that are reportable under the same work unit code.

## MAINTENANCE DATA COLLECTION SYSTEM

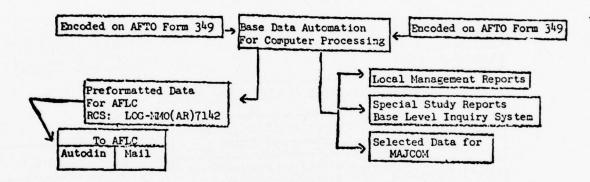
### REPORTING AT AF BASES

### "ON" EQUIPMENT

Corrective and Preventive Maintenance at or on the Aircraft or Equipment

## "OFF" EQUIPMENT

Corrective and Preventive
Maintenance Accomplished in
Repair Shops on Items Removed
From Aircraft or Equipment



rigure 9-1

## AFLC DOS6 WEEKLY CONFUTER PROCESSES

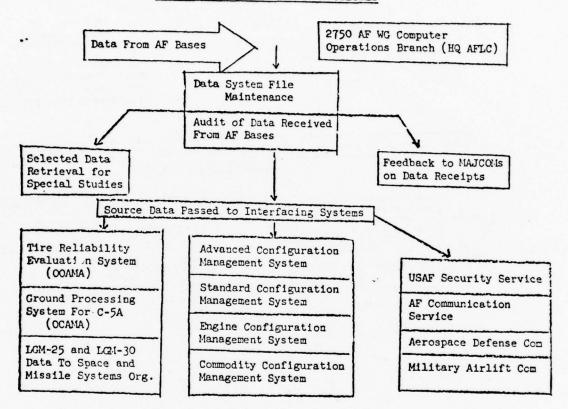


Figure 9-2

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## AFLC DOS6 MONTHLY COMPUTER PROCESSES

Data From DO56 Weekly Processes

Source Data to Interfacing Systems

Logistic Cost Ranking System Operating at Sacramento ALC

Flight Safety Prediction Technique Operating at San Antonio ALC

Analytical Interval Determination For Depot Level Maintenance

Department of Defense Contractors

Data From Interfacing Systems

Part Number to Stock Number Identification, Unit Price and AF Mgt Activity From Federal Cataloging System

Identification of the Item
Manager Technician and Division
Within the AFLC Air Materiel Area

Flying Hours, Inventory, Sorties and Landing From the Aircraft Status Reporting System

Accident, Incident and Emergency Unsatisfactory Materiel Report Data From the AFLC AF Materiel Safety Office

DO56 Output Reports

DO56 Data System Evaluation Reports

Reports Designed for Evaluation of Hardware and Maintenance Performance Related to Individual Weapons and Equipment

Reports Designed for Evaluation of Hardware and Maintenance Performance Related to Recoverable Items

Precision Measurement Equipment Calibration Interval Analysis

Selected Data Retrieval for Special Studies

tigure 9-3

- d. Use Factor  $(K_1)$ . This is a ratio of actual use time of individual Work Unit Codes to flying hours.
- e. Mean Time Between Failure Occurrence (MTBF).

 $MTBF \approx \frac{\text{End Item Operating Time} \cdot \times K_1 \times QPA}{\text{Quantity of Failures}}$ 

\*End item operating time is determined as follows:

For aircraft—active aircraft inventory flying time from AFM 65-110.

For other equipment—Active inventory flying time from AFM 65-110.

f. Mean Time Between Maintenance Occurrence (MTBM).

MTBM = End Item Operating Time\* × QPA

Quantity of Data Maintenance Occurrences\*
\*All types of actions described in paragraph
9-6a.

- g. Action Limit (AL). This is a form of failure limit expressed in MTBF (hours) and used in the computer program to compare current failure rates with past history for the same item.
- h. Failure Limit. This is the acceptable quantity of failures of an item for a 30-day period. It is assigned by the system manager and used in the computer program to compare current period failures with past history for the same item.
- 9-7. DO56 OUTPUT REPORTS. Selected reports containing R and M related data are identified and briefly described in the following subparagraphs.
- a. Selected Work Unit Code. Control Identifier, RCS: LOG-MMO(AR)7166. This report provides summarized information on Work Unit Codes within a weapon for the current reporting period that breeched either the Action Limit or Failure Limit; had Emergency Unsatisfactory Materiel Reporting; were high man-hour consumers or were high corrosion repair man-hour consumers. This report is used as a management reference to identify items that may warrant detail study and evaluation. Sample report Figure 9-4.
- b. Detail Maintenance Actions for Selected Work Unit Codes, RCS: LOG-MMO(AR)7167. This report provides one to twelve months of "on" equipment information on Work Unit Codes within a weapon for detail studies. It is available only on special inquiry and can be limited in data presentation by selective retrieval options. (Sample Report Fig. 9-5.)
- c. Detail Shop Actions for Selected Work Unit Codes, RCS: LOG-MMO(AR)7168. This report is a companion report to paragraph 9-7b and provides detail information from supporting repair shops on reparable items removed from a weapon. It also displays parts replaced during shop repair. (Sample Report Fig. 9-6.)
- d. Summarized Maintenance Actions for Selected Work Unit Codes: RCS: LOG-MMO(AR)7169. This report provides the same type of information as described in paragraphs 9-7b and c but with lesser detail. It is produced when the Action or Failure Limit is breeched and also by special inquiry using

selective retrieval routines. (Sample Report Fig. 9-7.)

- e. Maintenance Actions, Man-hours and Aborts by Work Unit Code, RCS: LOG-MMO(AR)7170. This report provides six months of selected information by month on every reportable Work Unit Code assigned to a particular weapon or equipment. This information includes aborts, failures, maintenance actions, MTBF, MTBM and man-hours. Both "on" and "off" equipment data are considered for display in this report (except for some types of AGE, trainers and munitions). (Sample Report Fig. 9-8.)
- f. Aborts and Degraded Alerts, RCS: LOG-MMO(AR)7171. This report provides current month detail information on Work Unit Codes and part numbers causing aborts, mission failures and degraded alerts. For ground equipment, this report identifies items causing equipment downtime. (Sample Report Fig. 9-9.)
- g. Materiel Safety Deficiency Report, RCS: LOG-MMO(M)7178. This report provides twelve months of selected information for Work Unit Codes applicable to a Mission Design Series aircraft that have been reported as contributing to an accident or incident or have been the subject of an Emergency Unsatisfactory Materiel Report. Any of the above events having occurred within the past twelve months and recorded in the DO56 system, drives the computer to display failure rate, trending and predictive maintenance experience data in this report as well as the quantity of hazard conditions reported. (Sample Report Fig. 9-10.)
- h. Work Unit Code Corrosion Summary, RCS: LOG-MMO(AR)7179. This report provides three months of information on a weapon or equipment identifying Work Unit Codes, number of units, manhours and labor cost for corrective maintenance due to corrosion. The 25 Work Unit Codes incurring the highest corrosion repair cost are rank ordered and displayed separately in the report for ease of identification. (Sample Report Fig. 9-11.)
- i. System, Subsystem Corrosion Summary, RCS: LOG-MMO(AR)7180. This report is produced as a comparison report to h above using the same three months of corrosion repair data except that the information is summarized to system/subsystem level and base location. (Sample Report Fig. 9-12.)
- j. System, Subsystem, Work Unit Code Failure Summary, RCS: LOG-MMO(AR)7183. This report provides twelve months of information related to current quarter experience for systems, selected subsystems and Work Unit Codes on an aircraft. The data displayed is rank ordered by system, subsystem within system and Work Unit Code within subsystem based on the quantity of failures incurred. Information displayed includes the quantity of failures, MTBF, and a ratio of current quarter to the last twelve months experience. (Sample Report Fig. 9-13.)
- k. Failure Rate Data for Selected Work Unit Codes, RCS: LOG-MMO(AR)7181. This report provides twelve months of information quarterly when the Action Limit is breeched and also by special in-

quiry using selective retrieval routines. Information displayed includes current quarter, previous quarter and 12-month MTBF, quarter to 12 month ratio and data groupings by when discovered, action taken, how malfunctioned and base location. (Sample Report Fig. 9-14.)

l. Maintenance Man-hours per Flying Hour by Weapon, Command and System, RCS: LOG-MMO(AR)7185. This report provides 12 months of information updated quarterly. The data displayed and the related computations are as indicated in the report title. (Sample Report Fig. 9-15.)

m. Maintainability Reliability Summary, RCS: LOG-MMO(AR)7220. This special inquiry report provides 12 months of information on Work Unit Codes within an aircraft. Information displayed includes failure, maintenance action, abort and manhour rates as well as the most predominate malfunction modes. (Sample Report Fig. 9-16.)

n. Selected Part Number Action Summary, RCS: LOG-MMO(AR)7188. This report provides 12 months of information on a recoverable line item of supply (part number worldwide) regardless of its installed use environment. It is produced when the computed failure limit for an item (Federal Stock Number) is breeched and also by special inquiry. The information displayed does not reflect maintenance required while installed in a weapon or equipment. It is limited to "off" equipment (shop and depot) repairs. (Sample report Fig. 9-17.)

o. Maintenance Actions for Selected FIIN Numbers, RCS: LOG-MMO(AR)7189. This report provides six months of "ofl" equipment (shop and depot repair) information on a recoverable item. Information displayed includes quantities of maintenance actions, malfunction modes, and base location. It is produced when the computed failure limit for an item (Federal Stock Number) is breeched and also by special inquiry.

p. Parts Replaced During Field or Depot Repair, RCS: LOG-MMO(AR)7190. This report is produced on the same criteria as above displaying six months of parts replaced during repair of items identified in the RCS: LOG-MMO(AR)7189. Information also displays quantity and reason for replacement. (Sample report Fig. 9-18.)

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### T.O.1F-111E-06

COO	TABLE OF CONTENTS	7345	CORE	TABLE OF CONTENTS	PARE	2005	ACTION TAKEN CODES DESCRIPTION
	PRETACE	1-01		INSTRUMENTATION		4	BENCH CHECKED AND REPAIRED THIS CODE WILL BE ENTERED WHEN
	TARLE OF CONTENTS	H-01	51	INSTRUMENTS, GENERAL	51-01		BINCH CHECK AND REPAIR OF ANY ONE
	TYPE MAINTENANCE CODES	IN 01	52	AUTOPILOT	52-01		TIME (ALSO SEE CODE F)
	TYPE MAINTENANCE CODES	IV-01	55	MALFUNCTION ANALYSIS AND RECORDING EQUIPMENT			BENCH CHECKED SERVICEABLE
	ACTION TAKEN CODES	V-01		RECORDING EQUIPMENT	55-01		(NO REPAIR REQUIRED) THIS CODE WILL BE ENTERED WHEN THE ITEM IS
	WHEN DISCOVERED CODES	VI 01		COMMUNICATIONS EQUIPMENT			BENCH CHECKED AND NO REPAIR WAS REQUIRED.
	HOW MALFUNCTIONED CODES		61	HE COMMUNICATIONS	61 01	t	DENCH CHECKED DIPAID DESIRATE
	ALPHARETICAL LISTING	AH-01	63	UHF COMMUNICATIONS	63-01		THIS CODE WILL BE ENTERED WHEN BENCH CHECK IS ACCOMPLISHED AND
	HOW MALFUNCTIONED CODES FOR HIGH POWER TUBES	VIII-01	64	INTERPHONE SYSTEMS	64-01		REPAIR ACTION IS DEFERRED. (SEE
	HOW MALFUNCTIONED CODES		65	IFF SYSTEMS	65 01		
	FOR COMPUTER OR PROGRAM- TYPE EQUIPMENT	IX-01	69	MISCELLANEOUS COMMUNICATIONS		.0	DENCH CHECKED TRANSFERRED TO ANOTHER BASE OR UNIT
	HOW MALFUNCTIONED CODES			EQUIPMENT	69-01		ITEM IS RENCH CHECKED AT A CODWADO
	NUMERICAL LISTING	X-01		MAVIGATION, FIRE CONTROL.			OPERATING BASE DISPERSED OPERATING BASE OR ENROUTE BASE AND
	AIRCRAFT SUPPORT, GENERAL (EXCEPT 03000 AND 04000)	XI 01		WEAPONS DELIVERY, ECH. PHOTO			IS FOUND UNSERVICEABLE AND TRANSFERRED TO A MAIN OPERATING
	A!RCRAFT SUPPORT, GENERAL	~ 01	71	RADIO NAVIGATION	71-01		COM WILL NOT BE USED FOR ITEMS
	(03000)	03 01	13	BOMBING NAVIGATION	73-01		BASE OR HOME BASE FOR REPAIR THIS COME WILL HOT BE USED FOR ITEMS RETURNED TO A DEPOT FOR OVERHAUL THIS CODE WILL ALSO BE USED WHEN PME OR OTHER EQUIPMENT IS SENT TO
	AIRCRAFT SUPPORT, GENERAL		14	FIRE CONTROL SYSTEM	74 01		
	(04000)	04 01	75	WEAPONS DELIVERY	75-01		CHECK CALIBRATION, OR REPAIR AND IS TO BE RETURNED, AND FOR ITEMS FORWARDED TO CONTRACTORS ON BASE
	AIRCRAFT, BASIC		16	ELECTRONIC COUNTERMEASURE	76-01		FORWARDED TO CONTRACTORS ON BASE LEVEL CONTRACTS
- 11	AIRFRAME	11 01	11	PHOTOGRAPHIC/RECONNAISSANCE	77.01		BENCH CHECKED HRTS (NOT REPAIRABLE
13	LANDING GEAR	13.01		MISCELLANEOUS EQUIPMENT			THIS STATION REPAIR NOT
14	FLIGHT CONTROL	14 01					AUTHORIZED) THIS CODE WILL BE ENTERED WHEN THE SHOP IS NOT AUTHORIZED TO ACCOM-
16	ESCAPE CAPSULE	16 01		PERSONNEL AND MISCELLANEOUS	96-01		PLISH THE REPAIR. THIS CODE SHALL
				EXPLOSIVE DEVICES AND			PLISH THE REPAIR THIS CODE SHALL ONLY BE USED WHEN THE REPAIR REQUIRED TO RETURN AN ITEM TO
	PROPULSION SYSTEM			COMPONENTS	97.01		SERVICEABLE STATUS IS SPECIFICALLY PROHIBITED BY CURRENT TECHNICAL DIRECTIVES THIS CODE SHALL NOT BE
73	TURBO IET POWER PLANT	23 01					DIRECTIVES THIS CODE SHALL NOT BE
	UTILITIES			ACTION TAKEN CODES			USED DUE TO LACK OF AUTHORITY FOR EQUIPMENT, TOOLS, FACILITIES, SKILLS PARTS, OR TECHNICAL DATA.
41	AIR CONDITIONING PRESSUR-		COD	DESCRIPTION			
	IZATION AND SURFACE ICE		•	THIS CODE WILL BE ENTERED WHE	N	1	BENCH CHECKED HRTS LACK OF EQUIPMENT, TOOLS, OR FACILITIES
.,	ELECTRICAL POWER SUPPLY	41 01		BENCH CHECK AND REPAIR OF AN	Y ONE		THIS CODE WILL BE ENTERED WHEN THE
	FIGHTING SAZIEM	47-01		TIME (ALSO SEE CODE F)			REPAIR IS ATTHORIZED BUT CANNOT BE ACCOMPLISHED DUE TO LACK OF EQUIP-MENT TOOLS OR FACILITIES THIS
45	HYDRAULIC AND PNEUMATIC	44 01		BENCH CHECKED SERVICEABLE (NO REPAIR REQUIRED) THIS CODE			CODE SHALL BE USED WITHOUT REGARD AS TO WHITHER THE EQUIPMENT TOOLS, OR FACULTIES ARE AUTHOR.
•,	POWER SUPPLY	45 01		WILL BE ENTERED WHEN THE ITEM	1 15		TOOLS, OR FACILITIES ARE AUTHOR-
46	FUFL SYSTEM	46 01		PEQUIRED AND NO REFAIR	ma3		
47	OXYGEN SYSTEM	47 01	C	BENCH CHECKED BEPAIR DEFERRE	0	,	DENCH CHECKED HRTS LACK OF TECHNICAL SKILLS
49	MISCELLAMEOUS UTILITIES	49.01		THIS CODE WILL BE ENTERED WHI BENCH CHECK IS ACCOMPLISHED	AND		THIS CODE WILL BE ENTERED WHEN REPAIR CANNOT BE ACCOMPLISHED DUE
				REPAIR ACTION IS DEFERRED. (SEE			TO LACK OF TECHNICALLY QUALIFIED PEOPLE
							DENCH CHECKED HRTS LACK OF PARTS
							THIS CODE WILL BE ENTERED WHEN PARTS ARE NOT AVAILABLE TO ACCOM-
							PLISH REPAIR
						5	BENCH CHECKED BRTS SHOP BACKLOS
							THIS CODE WILL BE ENTERED WHEN REPAIR CANNOT BE ACCOMPLISHED DUE TO EXCESSIVE SHOP BACKLOG.
						•	BENCH CHECKED HRTS LACK OF TECHNICAL DATA
							REPAIR CANNOT BE ACCOMPLISHED DUE
							TO LACK OF MAINTENANCE MANUALS.
							TAMED REPAIR PROCEDURES AND
						,	DENCH CHECKED MATS EXCESS TO BASE

BERCH CHECKED HRTS EXCESS TO DASE REQUIREMENTS
THIS CODE WILL RE ENTERED WHEN REPAIR WILL NOT BE SCHEDULED FOR

ACTION TAREN CODES (CONT)
CODE DESCRIPTION (CONT)
DENCH CHECKED WAIS EXCESS TO
REQUIREMENTS (CONT)

SHOP REPAIR DUE TO ITEM BEING EXCESS TO BASE REQUIREMENTS

- BARCH CHECKED RETURNED TO BEPOT RETURNED TO DEPOT BY DIRECTION OF SYSTEM MANAGER (SM) OR TIEM MANAGER (BM) OR THE MANAGER (BM) OR BASE (EVER BEPAIR ARE AUTHORIZED TOR BASE (EVER BEPAIR ARE DIRECTED TO BE RETURNED TO DEPOT FACILITIES BY SPECIFIC WRITTEN OF VERBAL COMMUNICATION FROM THE MO OF VERBAL COMMUNICATION FACINITIES FOR MODIFICATION IN ACCORDANCE WITH A TIME COMPLIANCE TECHNICAL ORDER (TCTO) OR AS UR EXHIBITS
- BERCH CHECATO COMPENSED
  HIS CODE WILL BE ENTERED WHEN THE
  ITEM CANNOT BE REPAIRED AND IS TO
  BE PROCESSED FOR COMPENSATION,
  RECLAMATION OR SALVAGE HIS CODE
  WILL ALSO BE USED WHEN A "CON
  DEMNED. CONDITION IS DISCOVERED
  DURING FIELD MAINTERNANCE DISASSEMBLY OR REPAIR.
- INITIAL INSTALLATION
  THIS CODE WILL BE USED FOR INSTALLATION ACTIONS THAT ARE NOT RELATED TO A PREVIOUS REMOVAL ACTION SUCH AS INSTALLATION OF AND ITEM TO REMEDY A SHIP SHORT CONDITION HIS CODE WILL BE USED ONLY FOR EQUIPMENT MANAGED UNDER THE ADVANCE CONTICURATION MANAGEMENT SYSTEM REFERENCE TO SO 20-2-3, AND 00-70-2-2-7. MUST BE USED WITH HOW MALCODE 799.
- THIS CODE WILL NOT BE USED TO CODE
  "THIS CODE WILL NOT BE USED TO CODE
  "ON EQUIPMENT" WORK IF ANOTHER
  CODE WILL APPLY WHEN IT IS USED
  IN A SHOP RAVINDAMENT. THIS CODE
  WILL DENTE REFAR AS A SEPARATE
  UNIT OF WORK AFTER A BENCH CHECK
  SHOP REPAIR NULLUGES THE TOTAL REPAIR MAIN HOURS AND INCLUDES CLEANING. DISASSEMBLY INSPECTION
  ADJUSTMENT BEASSEMBLY AND
  LUBRICATION OF MINOR COMPONENTS
  INCIDENT TO THE REPAIR WHEN THESE
  SERVICES ARE PERFORMED BY THE SAME
  WORK CENTER. FOR PRECISION
  MEASUREWENTS QUIPMENT. THIS CODE
  WILL BE USED ONLY WHEN CALIBRATION
  OF THE REPAIRED TIEM IS REQUIRED
  USEE CODE G).
- SEE CODE G).

  REPAIR ARBJOR REPLACEMENT OF BINDE PARTS, MACROWARE AND SOFTGOODS (SEAS, GASAETS, ELECTRICAL COMMECTORS, ITTINGS, TUBING, MOSE WRING FASTIKINES, VIRRATION (SOLATORS BRACKETS, ETC.)

  WORK UNIT CODES DO NOT COVER MOST NON REPAIRABLE TIEMS, THREFORE WHEN ITEMS SUCH AS THOSE IDENTIFED ABOVE ARE REPAIRED OR REPLACED THIS ACTION TARKE CODE WILL BE USED WHEN THIS ACTION TARKE CODE IN USED THE WORK UNIT CODE WILL DERTIFY THE ASSEMBLY RELATED TO PARTS BEING REPAIRED OR REPLACED FOR EXAMPLE. If AN ELECTRICAL COMMECTOR WAS REPAIRED OR REPLACED FOR EXAMPLE, If AN ELECTRICAL COMMECTOR WAS REPAIRED ON TRANSMITTER HOW DO A RADIO TRANSMITTER THE WORK UNIT CODE ON THE TRANSMITTER WOULD BE USED OR REPAIRED THIS ACTION TARKE CODE FOR PRECISION MEASUREMENT EQUIPMENT HIS ACTION TARKE CODE FOR PARTS THAT DO NOT REPUIRE CALIBRATION OF THE REPAIRED ITEM (SEE CODE F).

CODE DESCRIPTION (CONT)

- EQUIPMENT CHECKED NO REPAIR REQUIRED (FOR "ON EQUIPMENT" WORK ONLY)

  INS CODE WILL BE USED FOR ALL DISCREPANCIES WHICH ARE CHECKED AND FOUND TO REQUIRE NO FURTHER MAINTERANCE ACTION THIS CODE WILL BE USED ONLY IF ITS DETAINETLY DETERMINED THAT A REPORTED DEFICIENCY DOES NOT ENSIT OR CAMPOT BE DUPLICATED. MUST BE USED WITH HOW MAI CODE 799, 812, OR 948.
- CALIBRATES NO ABJUSTMENT REQUIRED US: THIS CODE WHIN AN ILEM IS CALIBRATED AND FOUND SERVICE ABLE WITHOUT NEED FOR MOLISTMENT. OR IS FOUND TO BE IN TOLERANCE BUT IS ADJUSTED MERELY OF PEAR OR MAXIMIZE THE READING. IT THE ITEM REQUIRES ADJUSTMENT TO ACTUALLY MEET CALIBRATION STANDARDS ON TO BRING IN TOLERANCE USE CODE K.
- CALIBRATED ADMISTMENT REQUIRED

  USE THIS CODE WHEN AN ITEM MUST BE
  ADUSTED TO BRING IT IN TOLERANCE

  ON MEET CALIBRATION STANDARDS IF
  THE ITEM WAS REPAIRED OR REEDS
  PAIR IN ADDITION TO CALIBRATION
  AND ADJUSTMENT, USE CODE F.
  - ABJUST
    INCLUDES ADJUSTMENTS NECESSARY FOR
    SAFETY AND PROPER TUNCTIONING OF
    EQUIPMENT SUCH AS ADJUST. BLEED.
    BALANCE, FIG. FIT REFOULD STATION, OR
    RESEAT, POSTIDON, PEPSTITION, OR
    CIRCUIT BREAFER FOR USE WHEN A
    DISCREPANCY OR CONDITION IS
    CORRECTED BY THESE TYPES OF
    ACTIONS OF THE FORTING COMPOINTIO TO ASSEMBLY ALSO REQUIRES
    REPLACEMENT OF BITS AND PIECES AS
    WELL AS ADJUSTMENT ENTER THE
    APPROPRIATE REPAIR ACTION TAKEN
    CODE INSTEAD OF L.
- DISASSEMBLE
  THIS CODE WILL BE ENTERED FOR DISASSEMBLY ACTION WHEN THE COMPLETE
  MAINTENANCE JOB IS BROATN INTO
  PARTS AND REPORTED AS SUCH. DO NOT
  USE FOR ON EQUIPMENT WORK.
- # ASSEMBLE
  THIS CODE WILL BE ENTERED FOR
  ASSEMBLY ACTION WHEN THE COMPLETE
  MAINTENANCE JOB IS BROKEN INTO
  PARTS AND REPORTED AS SUCH DO NOT
  USE FOR OR-EQUIPMENT WORK.
  - BEMOVED

    IHIS CODE WILL BE ENTERED WHEN AN ITEM IS REMOVED AND ONLY THE REMOVED AND ONLY THE REMOVAL IS TO BE ACCOUNTED FOR IN THIS INSTANCE DELIVED OR ADDITIONAL ACTIONS WILL BE ACCOUNTED FOR SEPARATELY (ALSO SEE CODES OR S. T. AND U.) DO NOT USE FOR OFF-EQUIPMENT WORK
  - INSTALLED
    THIS CODE WILL BE ENTERED WHEN AN
    ITEM IS INSTALLED AND ONLY THE
    INSTALLATION ACTION IS TO BE
    ACCOUNTED FOR (ALSO SEE COOLS E.
    P.R. S. T. AND U.) DO NOT USE FOR
    OFF EQUIPMENT WORK.
- BEWAYE AND REPLACE
  THIS CODE WILL BE ENTERED WHEN AN
  ITEM IS REMOVED AND ANOTHER LINE
  ITEM IS MISTALLED (ALSO SEE CODES
  1 AND U.) DO HOT USE FOR OFF.
  EQUIPMENT WORK
- BEMOVE AND REINSTALL
  THIS CODE WILL BE ENTERED WHEN AN
  ITEM IS REMOVED AND THE SAME ITEM
  REINSTALLED. (ALSO SEE CODES T AND

ACTION TAKEN CODES (CONT)
CODE DESCRIPTION (CONT)
S REMOVE AND REINSTALL (CODE)

U) DO NOT USE FOR OFF EQUIPMENT WORK MUST BE USED WITH HOW MAL CODE 800, 804 OR 805.

- THE WAYER FOR CAMMIDALIZATION
  THIS CODE WILL BE INTERED WHEN A
  COMPONENT IS CANNINALIZED THE
  WORK UNIT CODE WILL SOFWILLY THE
  COMPONENT BEING CAMMIDALIZED DO
  NOT USE THIS CODE FOR OFF
  EQUIPMENT WORK MUST BE USED WITH
  HOW MAL CODE 799.
- B REPLACED AFTER CARRIBALIZATION
  THIS CODE WILL BE ENTRED WHEN A
  COMPORENT IS REPLACED AFTER
  CAMMIDALIZATION DO NOT USE THIS
  CODE 10R OFF EQUIPMENT WORK MUST
  BE USED WITH MOW MAI COGE 799
  - CLEAN
    INIS CODE WILL BE ENTERED WHEN
    CLEANING IS ACCOMPLISHED TO
    CORRECT DISCREPANCY AND/OR WHEN
    CLEANING IS NOT ACCOUNTED FOR AS
    PART OF A REPAIR ACTION SUCH AS
    CODE F INCLUDES WASHING ACTO
    BASH BUFFING SAMP BACKING
    DECREASING, DECONTAMINATION FTC
    CLEANING AND WASHING OF COMPLETE
    ITEMS SUCH AS GROUND EQUIFIMENT
    VEHICLES, MISSIGES OF AIRPLANES
    SHOULD BE RECORDED BY UTILITING
    SUPPORT GENERAL CODES
- I TEST INSPECTION SERVICE
  IMIS COOR WILL BE ENTIRED WHEN AN
  ITEM IS TESTED OR INSPECTED OR
  SERVICED TOTAL THAN BENCH CHECKS
  AND NO REPAIR IS REQUIRED THIS
  COOL DOES NOT INCLUDE SERVICING OR
  IMPECTION CHARGEBARE TO SUPPORT
  GEMERAL WORK UNIT CODES
- TROUBLESHOOT

  FAIRE THIS CODE WHEN TIME EXPENDED
  IN LOCATING A DISCREPANCY IS GREAT

  FROUGH TO WARRANT SEPARATING THE

  FROUBLESHOOT TIME FROM THE REPAIR

  TIME USE OF THIS CODE

  FROM THE USE OF THIS OR TWO

  SEPARATE LIME EXTRICS OR TWO

  SEPARATE FORMS, OWE OF OR THE

  FROUBLESHOOT FINES SEPARATE FROM

  THE REPAIR PHASE WHIN RECORDING THE

  FROM THASE WHIN RECORDING THE

  FRAME THASE WHIN RECORDING THE

  FRAME THASE WHIN RECORDING THE

  FRAME THASE WHIN FECTIONAL THASE

  OF THE DISCREPANCY SHOULD BE

  RECORDED UTILIZING THE WORK WHIT

  ODE OF THE DISCREPANCY SHOULD BE

  RECORDED UTILIZING THE WORK WHIT

  ODE OF THE DISCREPANCY SHOULD BE

  RECORDED UTILIZING THE WORK WHIT

  ODE OF THE DISCREPANCY STEELE

  JE SYSTEM DO NOT USE FOR OFF-
- EQUIPMENT WOPK

  PRIMARY AND PAINING TREATING PRIMARY AND PAINING OF CORRODED ITEMS THIS CODE SHOULD ALWAYS BE USED WHEN ACTUALTY TREATING CORRODED ITEMS EITHER ON FOURMENT OR IN THE SHOP THE WORL UNIT CODE SHOULD IDENTIFY THE HEM THAILS CORRODED USE SUPPOSE GENERAL CODE FOR PAINING OR CORROSION PREVENTIVE TREATMENT FROM 10 AM TIEM BECOMING CORRODED

#### T.O.1F-111E PS

## TYPE MAINTENANCE CODES

TYPE MAINTENANCE CODES FOR AIR CRAFT, ORONES, INSTALLED ENGINES AND RELATED MOBILE TRAINING SEIS (MTS) AND RESIDENT TRAINING EQUIPMENT (RTE). ENGINE SHOP CODES ARE INCLUDED FOLLOWING THIS LIST OF CODES

#### SERVICE .

INCLUDES ALL UNITS OF WORK ASSOCI MOVEMENT OF EQUIPMENT.

#### UNSCHEDULED MAINTENANCE .

INCLUDES ALL UNITS OF WORK ACCOMPLISHED BETWEEN SCHEDULED INSPECTIONS EXCEPT AS PROVIDED IN PRECEDING CODE A, AND EXCLUDING ACCOMPLISHMENTS OF ICTO'S

#### BASIC POSTFLIENT OR C THRUFLIGHT INSPECTION

INCLUDES 'IL UNITS OF WORK ACCOMPLISHED DURING ALL PHASES OF THE BASIC POSTFLIGHT OR THRU-FLIGHT INSPECTION.

#### PREFLIGHT INSPECTION 0

INCLUDES ALL UNITS OF WORK ACCOMPLISHED DURING ALL PHASES OF A PREFLICHT INSPECTION FOR MOBILE TRAINING SEIS AND RESIDENT TRAIN ING EQUIPMENT THIS INCLUDES ALL UNITS OF WORK ACCOMPLISHED DURING SCHEDULED INSPECTIONS SUCH AS DAILY, SAFETY, AND SERVICING IN-SPECTION, EXCLUDING PERIODIC INSPECTIONS

## CALIBRATION OF OPERATIONAL EQUIP MENT (NON PME) BY OWNING OR ASSISTING WORK CENTER

EXCLUDES CALIBRATION ACTIONS BY PME CALIBRATING WORK CENTERS (SEE 1.0. 00-25 06-4-1 FOR TYPE MAINTENANCE CODES FOR PME).

#### PERIODIC PHASES OR MAJOR INSPECTION

INCLUDES ALL UNITS OF WORK PHASES OF PERIODIC, PHASED OR MAJOR INSPECTIONS, EXCLUDING ACCOMPLISHMENT OF TCTO'S.

#### FORWARD SUPPORT SPARES .

INCLUDES ALL UNITS OF WORK PER FORMED BY ALL ACTIVITIES IN RECORDING IN-SHOP MAINTENANCE ACTIONS ON MAC FORWARD SUPPORT SPARES. EXCLUDING ACCOMPLISHMENT OF TCTO'S

#### DEPOT MAINTENANCE .

INCLUDES ALL UNITS OF WORK ACCOMPLISHED WHEN DEPOT MAINTE NAME OR REHABILITATION IS PERFORMED, REGARDLESS OF LOCATION. EXCLUDES ACCOMPLISHMENT OF TOTO'S.

#### SPECIAL INSPECTION .

INCLUDES ALL UNITS OF WORK ACCOMPLISHED DURING ALL PHASES OF SPECIAL INSPECTIONS. EXCLUDING ACCOMPLISHMENT OF TOTO'S. INCLUDES ALL FUNCTIONAL CHECK FLIGHTS

#### TIEE COMPLIANCE TECHNICAL ORDER (TCTO)

INCLUDES ACCOMPLISHMENT OF ALL

#### AIRCRAFT TRANSIENT MAINTENANCE

INCLUDES ALL UNITS OF WORK ACCOMPLISHED ON/OR FOR TRANSIENT AIRCRAFT. EXCLUDING ACCOMPLISHMENT

#### HOTE

SEE TO. 00 25 06-2-1 FOR OFF EQUIPMENT SHOP TYPE MAINTENANCE

#### WHEN DISCOVERED CODES CODE DESCRIPTION

- BEFORE FLIGHT ABORT AIR CREW
- BFFORE FLIGHT-NO ABORT-AIR CREW.
- W FLICHT ABORT
- IN FLIGHT NO ABORT
- AFTER FLIGHT . AIR CREW
- BETWEEN FLIGHTS-GROUND CREW
- GROUND ALERT NOT DEGRADED G
- BASIC POSTFLIGHT INSPECTION
- PREFLICHT INSPECTION.
- DURING TRAINING OR MAINTENANCE ON EQUIPMENT UTILIZED IN A TRAINING ENVIRONMENT (USE ONLY FOR CLASS II TRAINING EQUIPMENT). THIS CODE SHOULD BE USED WHEN RECORDING MAINTENANCE OR DISCREPANCIES ON CLASS II TRAINERS.
- PHASED INSPECTION
- GROUND ALERT DEGRADED
- FUNCTIONAL CHECK FLIGHT.
- SPECIAL INSPECTION 0
- QUALITY CONTROL CHECK
- S DEPOT LEVEL MAINTENANCE.
- DURING SCHEDULED CALIBRATION.
- NON DESTRUCTIVE INSPECTION. IN 11 CLUDES OPTICAL PENETRANT, MAG-NETIC PARTICLE, RADIOGRAPHIC, EDDY CURRENT, ULTRASONIC, SPECTROMETRIC OIL ANALYSIS, ETC.
- DURING UNSCHEDULED CALIBRATION.
- IN SHOP REPAIR AND/OR DISASSEMBLY FOR MAINTENANCE.
- ENGINE TEST STAND OPERATION.
- UPON RECEIPT OR WITHDRAWAL FROM
- SUPPLY STOCKS
  DURING OPERATION OF MALFUNCTION
  ANALYSIS AND RECORDING EQUIPMENT
  OR SUBSEQUENT DATA ANALYSIS
  - CORROSION CONTROL INSPECTION.

# THIS PAGE IS BEST QUALITY PRACTICABLE FROM COPY FURNISHED TO DDC

Table B-1. How Malfunctioned Codes - Numerical Listing

8	· Gan,		Fuel Mossle Coking	700	Manifold Pressure Beyond	782	Tire Iread Area Defective -	•17	Impending Failure or La-
8	Open Pilament or Tube Cir-	279	Spray Pattern Defective		-		Use Cut, Delasinated, Punc-		tent Defect Indicated by
8	Colt	240					tured. Worn, etc., 11		Hon-destructive inspec-
8	The Car of Laistion	-		3	-		Tire Sidewall Danged or		codes and a
800	Moley	8 5	Grounded Electrically	.04	Mon-Co Lodicas (no Canada		Defective		
.00	Ricrophonic	2 5	Poretien Object Damege		tic Beans University	786	Tire bead Area Damaged or		
010	Poor or Incorrect Focus	716	Slow Acceleration	615	Shores		Defective		Activation (Use Code 386
020	Worn. Chafed or Frayed	315	RPH Fluctuation or Incor-	617	Sulfidetion	785	Tire Inside Surface		If ites was lost in
025	Capacitance Incorrect		1000	619	Shimy Excessive		Desegred or Defective	:	(11Cht)
020	Conductance Incorrect	317	Hot Start	6:2	Met / Condensat ton	793	No Detect - ICIO NIC Ne-		Does not theate, Lock or
	Current Incorrect		Excessive Na	625			10 11dans seed to parter	***	Secret or Services
	Fluctuates, Unstable or	7	Temperature Incorrect				Supply	633	Overheated Cathode Stem
170		150	Insulation Breakdown		Accountion incorrect	101	1000	9.38	Power Output 3:p
	Department and or prints		Metal on Magnetic Plug		placement for Resulted			• 39	
8	Migh Voltage Standing Maye		Internal tallure		Blas Voltage Incorrect		plied with		
	<b>L</b> t10		Person linksom		Pretted (Ternel Battery.	798		9:2	
	Flee-dut	181	Leaking - Internal or fr-		Fire Dringingner, Etc.)		der Not Applicable-Equip-	;	**************************************
0.0			teme!	635	Sensitivity Incorrect		ment to be Replaced, 40c1-		Data Error
080	Burned Out or Defective		Liquid Lock	970	Time Delay Incorrect		Tied of Not installed		Incorrect of to Print Our
	Comp. Meter or Indicating	·	Lock on Relfunction		State of the state	90			For
980	Increase Mandillas		Maintenance Action Due to		AIR IN System		Removed and/or Rein-		No Defect Operator Frror
0.00	Incorrect Carn	101	Lost in right Occurrence	652	Automatic Align Time		stalled to Facilitate	676	
	Rusaltined - Weel Halves.		Oli Consumption facessive		Excessive		-	***	Defect
	Electronic Parts, etc.	•10	Leck of . or Improper	653	Ground Speed Error Excessive	80	No Defect - Technical	936	Abnormal Function of Com-
8	No fain or Enission		Labrication		-	802			
	Long or Design House		External Power Source	635			Technical Order Com-		ment.
	Multa Serens Rivers		Wicked		-		plience		No Dieplay
	Fasteners, Clamps, or Other	9	Oren Logic			803	To Defect - Removed for		Incorrect Display
	Comon Hardware		Oscillating	443	Dierarie Meseurescop French	804	Time Change		dundant Equipment
8	Rissing Bolts. Huts. Sereve.		Out of Lalance				Scheduled Maintenance or		Mich Anode Current
	Rivets, Fasteners, Clasps.		Overspeed	650			Modification	962	Las Pover - Electronic
108	Broken, Faulty or Masine		Bushing Worn or Desaged				No Defectation Otherwise	963	Broken Filament/Cathode
			Circuit Protector	000	Stripped		Coded Electrical, Lydrau-	;	Terminal
	Burst or Ruptured	*15	ingine Failed to Start	567			Only eld. Lines. Wether-		AF Window Sucting Broken or
2:	Cut		Keway or Spline	649			ICE L'INGER and Mt.es		Cracked
	Deterlorated	į	Danked or Worn		(Reversion Process)		Servicing & Corponent Se-	968	Dioding
	Improper		Turbine baseged - Reason		Vibration Excessive		Toved Disconnected and or		Cannot Resonate Input Cavit
	Veiu	503	Sudden Stop		Audio Faulty		Pariliate Creat Asing	971	Cracked Cathode Bushing
135	Binding, Stuck or Jamed	513	Compressor Stall		Audio and Video Fauity		Parce.	972	Damaged Input Probe
162	Engine Removed, Excessive	518	Improper Routing		Sync Absent or Incorrect		so Defect-Indicated Iv-	973	Damaged Jutput Probe
9		250	Pitted		Faulty Tape - Program or		fect Caused by Associat-		Does Not Trace Juning Curve
	Leunch Comes		Tressure incorrect		Checkout Program		ed touthment "sifunction		Freezenty Instability
9	Contacts (Connection Delec-	0,0	Published		Checkout		Table location	98;	Frozen Tuning Mechanisa
	1100	553	Does not Meet Specifica-	104	Administrative Condemation		Unto Processo.	983	Grid to Cathode Snort
167	Torque Incorrect		tion. Drawing, or Other	710	Bearing Fallure or Faulty	834	a Plus Incerrect		Grid to Plate Snorr
	Incorrect Voltage		Contormance Requirements		Spread alanco,	970			Interreption
111	Fuel Flow Incorrect		Men discovered Code V)				THE MINE OF THE PARTY OF	986	Nigh Modulator Inverse
	Compression Low	361	Unable to Adjust to	719	Broken or Fraved Sonding or	877	Transporterion Damage		Input Pulse Distortion
. 061	Crecked		Lieits				Weather Danage	986	Loss of Vacuum
500	Accidental faplosion of, or	267	Resistance Incorrect		מנופון נייורקביים בייביי		Lead Broken	000	to focus current
	Damege from, Omboard Muni-		Scope Presentation in-				forestitions	166	Out of Rand Frequency
330	Dirty. Contactnated or	\$85	Sheared			010	Chipped		Jutour Pulse Distortion
	Saturated by foreign	865	Auto Rotation MM In-		Decoupled	911	Techaicel Order Not		of fred Dr. Attenued
***	Macerial	•	correct (Migh/Low)		Battle Damage		Complised with - 1270		Distorted
	tion - Specific Reson		Correct	748			or Pieced in sore in		RF Feed - Thru Completely
	Unknown	109	Detonation	•	correct		Error.	***	Interrupted
200.	. Improper or Faulty Main-		Failed or Designed Due to		Sin Rice of Comutator	916	Espending of Incipient	: :	Af Window Burned
341			Welfunction of Associated		Fallure		Spectrometric 311		
. 552	No Outmit / Inches	109	Oil in Induction System	780	Bent, Burmled, Collapsed,		Analysis		
					Dented Distorted or				
					'Visied				

Typical Avionics Work Unit Codes for UHF Communications

1.0. 16-1116-06			SWITCH, STABILIZATION AMPLIFIER, SCAN MONITOD DAILET		SYNCHUDALZER-TRANSMITTER				EQUIPMENT RACK, ELECTRICAL		4		NOC .		9			CCMPUTER, FLIGHT VECTOR			CENEDATOR SELF-15ST			DEMODULATOR STICK POSITION		ATTACK RAUAR AN/APO-144	•	2		OVERLOAD 1 ASSY				
	UNIT CODE	73000 73KE 0	73KEH *3KEJ	73KE9	73KFU	73KFA	73KFC	73KF9	73KG0	2000	73840	73KHB	73KH9	73KKO		73KKA	73KKC	73KKD	73KKE	73KKG	73KKH	73KK	73KAL	73KKM	73KK9	13400	73440	73444	73VAB	73VAC	73446	TRAF	73VAG	73443
Terrain Following hadar		8048	AMPENNA-RECEIVER, AS-23 30/APG-128 AMPLIFIER, OUAL IF AMPLIFIER, AUTOMATIC FREQUENCY	CONTROL VIDEO PROCESSOR	PHASE DETECTOR VIDEO SUBASSY, (PN 582946-1)	SWITCH, MAVEGUIDE POWER SUPPLY, KLYSTRON	OSCILLATOR, SELF-TEST	RECEIVER DUPLEXER	KLYSTRON & CONNECTOR ASSY	CATAC TOO MIX GOOD TO COMPANY		PANEL ASSY. FRONT	INDICATOR ASST NO. 1	INDICATOR ASSY NO. 3	ASS. 1941	SELECTOR ASST. RANGE	SWEEP DRIVER AND VOLTAGE	REGULATOR ASSY (A9)	ELECTRON TUBE AND DEFLECTION	ASSY	NOC		CONTROL. RADAR SET. C-7510/APQ-128	01006 ASSY	DON	AMPLIFIER, POWER SUPPLY,	AMPLIFIER. ELEVATION	AMPLIFIER, AZIMUTH	PROGRAMMEN, SCAN	POWER SUPPLY, PLUS 15 VDC	SWITCH ASSY, POWER	DIODE & CAPACITOR		
ŭ	UNIT	73000	73K80 73K8A 73K8B	73×8C	73K80 73K8E	73881	73KBM	73x8P	73×85		משני	73KCA	73860	73KCE	73KCF	73KCG	73KCH		73KCK		73866	1747	73×00	73KOB	73KD9	73KE0	73KFA	73KEB	73KEC	73KED	73KEF	73KEG		
and Terral		UNE COMMUNICATIONS RECEIVER-TRANSMITTER RT-749/ARC-	109 POWER SUPPLY 0635 AMPLIFIES AF 0629	ACITER, 0630 R. SIGNAL DATA, 0628		ARCELVER, ANDIO-GUAND 0634	MODULE, MUDULATOR, 0631		GITAL DATA COMPARATOR	MOUNT MT-3322/ARC-109	100	CCNTROL ARC-109	NOC		ANTELNA SELECTOR CABOBLANC	ANTOLINA CONFR	821/A	PANEL ASST. ANTENNA SELECT	NOC															
	#08 x	63000	63444	634AE 634AF	03446	63443	03440	63441	6 34AV	5344#	63444	63450	63409		63ACA	6340	63ACD	63ACE	63469															

### APPENDIX C

Average Number of Maintenance Actions per Sortie versus Sortie Length for Selected Systems

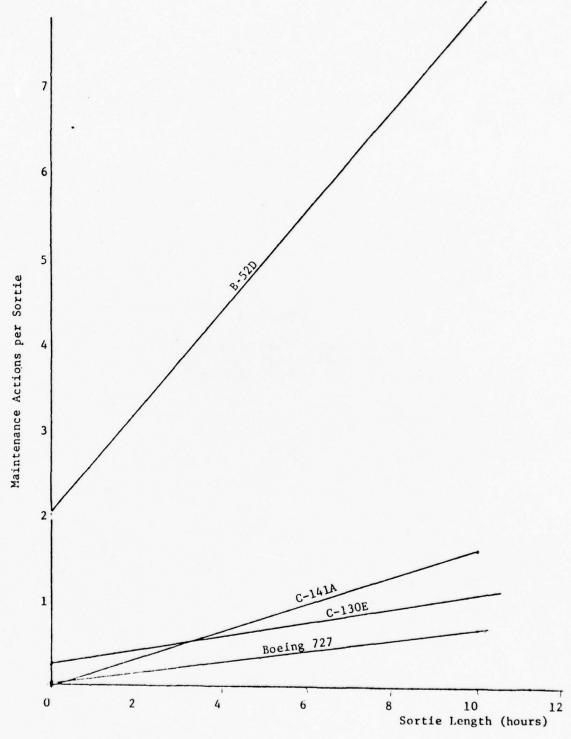


Figure C-1. System 11- Airframe

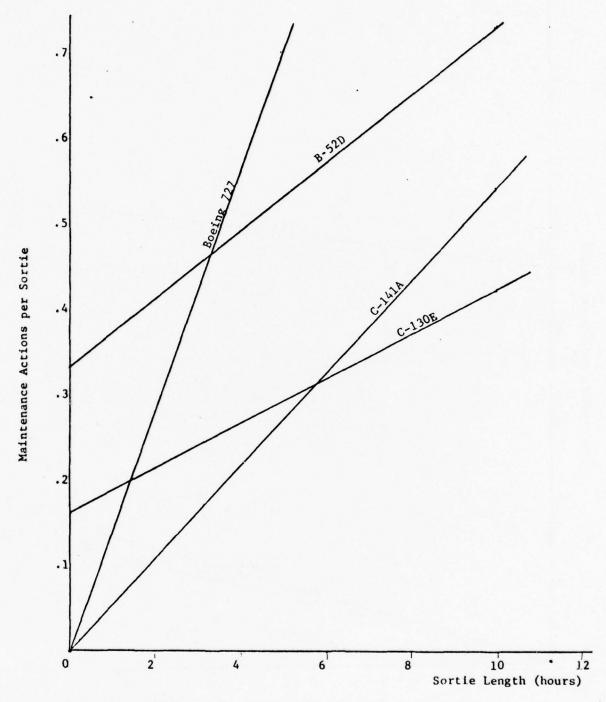


Figure C-2. System 12 - Fuselage Compartments

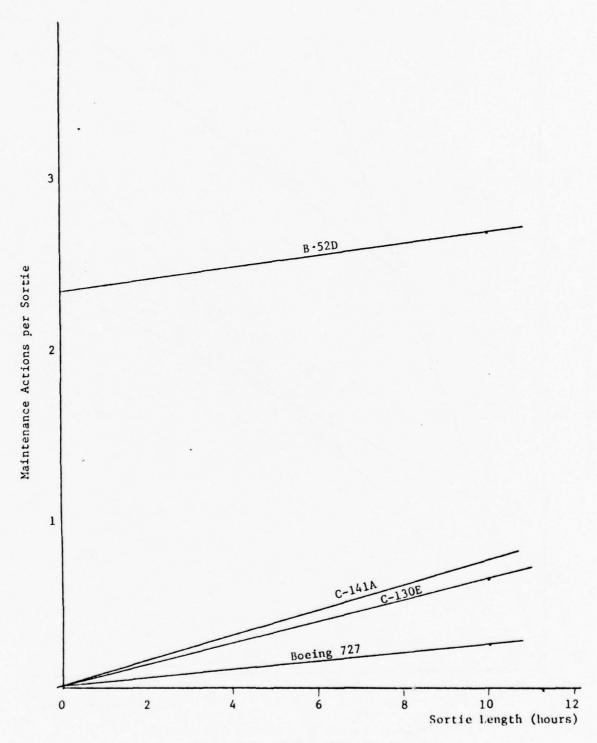


Figure C-3. System 13 - Landing Gear

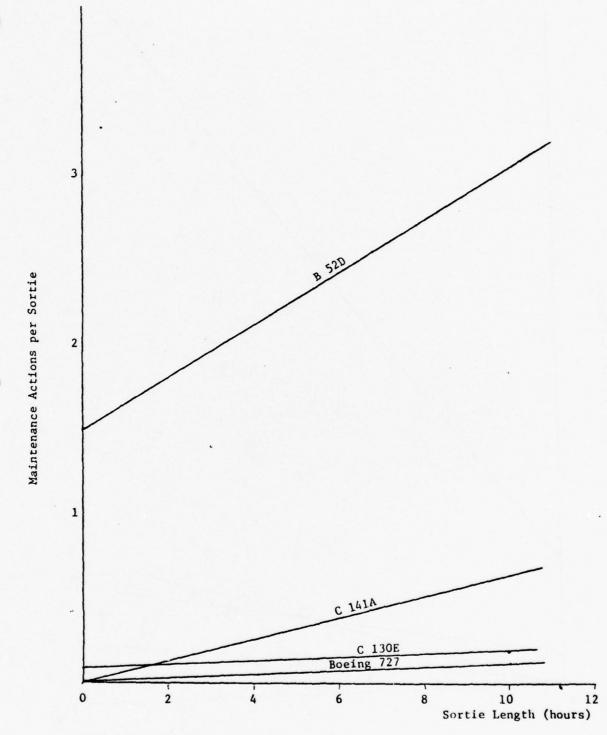


Figure C-4. System 14 - Flight Controls

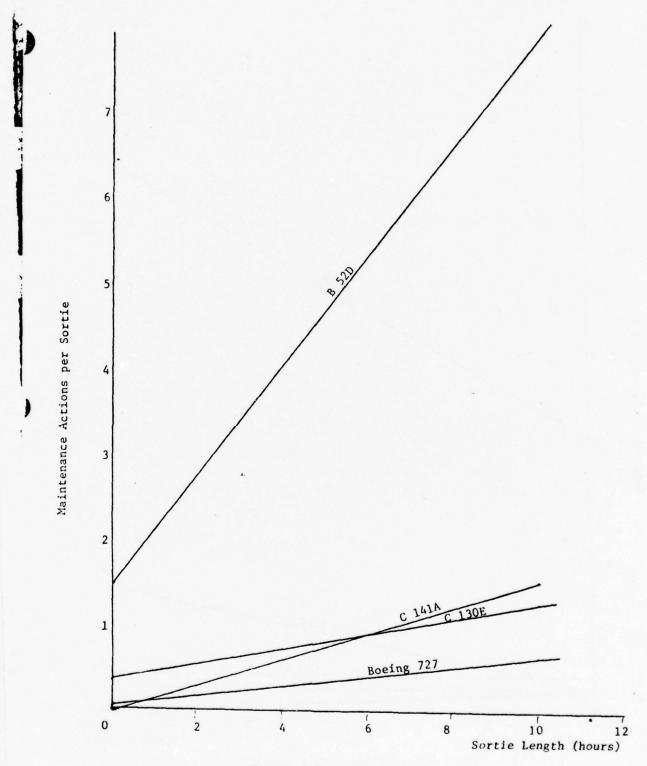


Figure C-5. System 22/23 - Engine

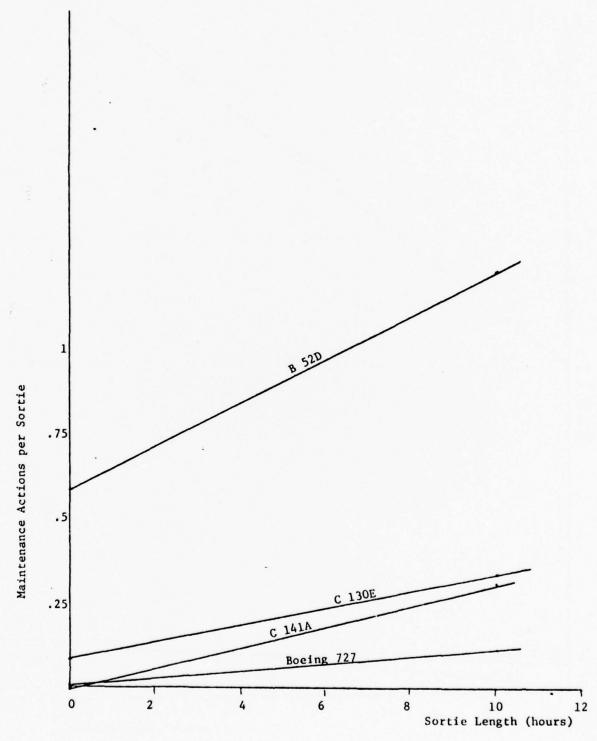


Figure C-6. System 41 - Air Cond

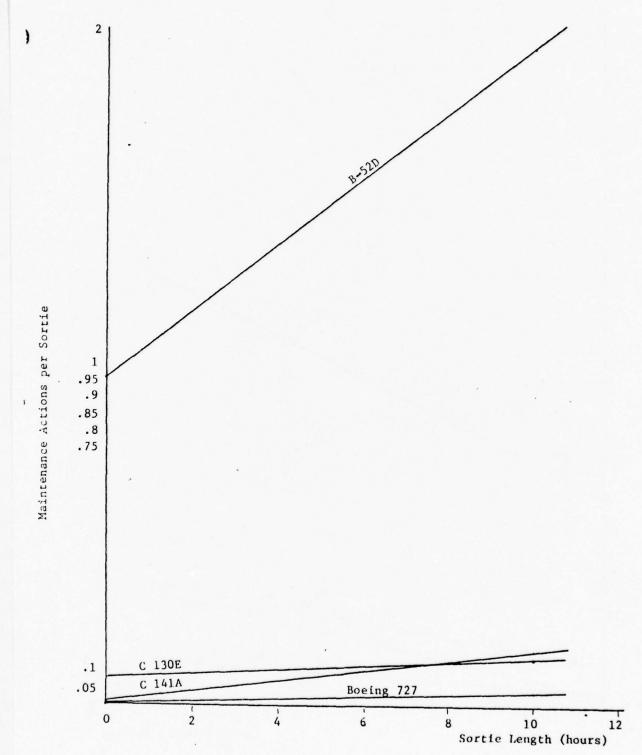


Figure C-7. System 42 - Electrical Power Supply

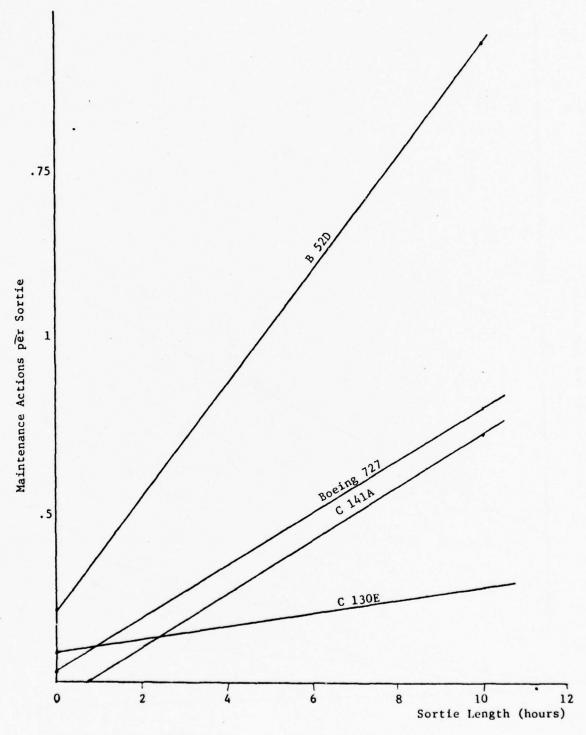


Figure C-8. System 44 - Lighting Systems

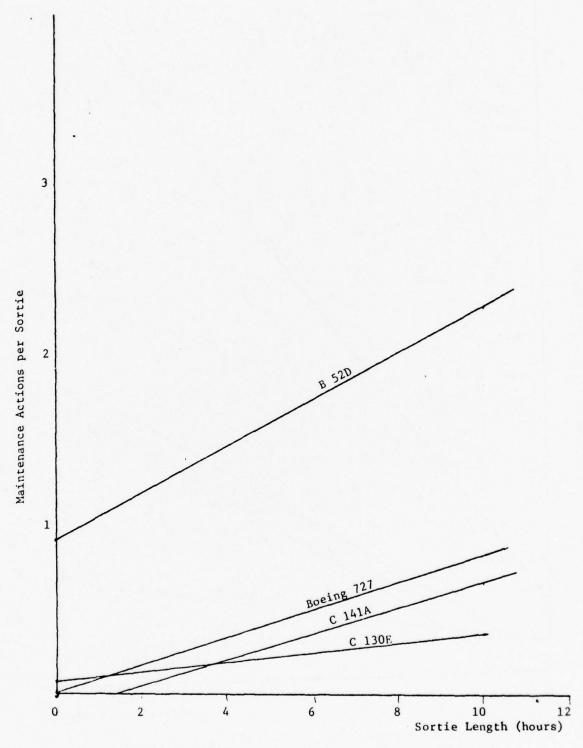


Figure C-9. System 45 - Hydraulic & Pneumatic Sys

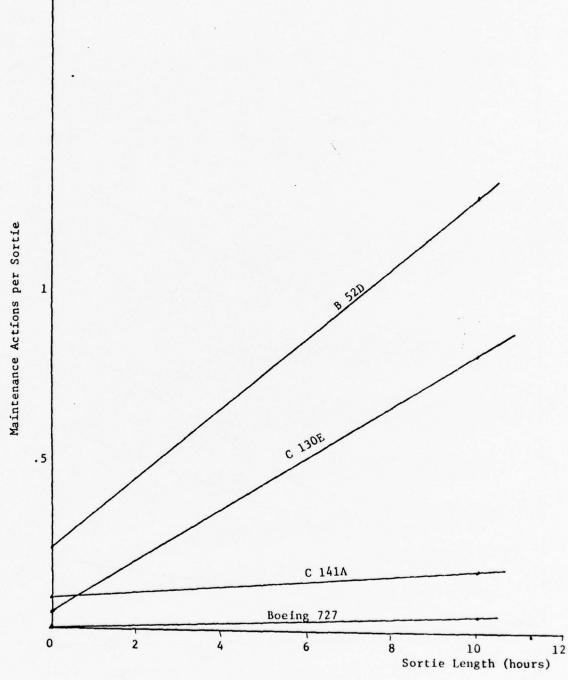


Figure C-10. System 46 - Fuel Systems

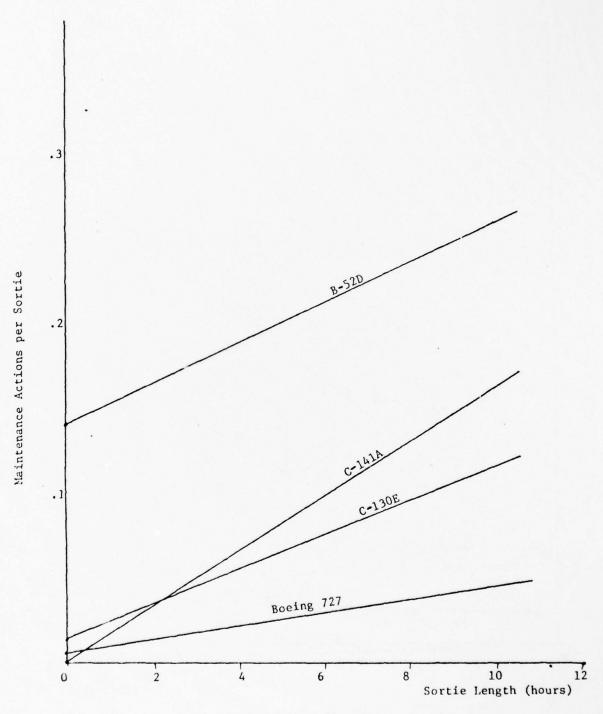


Figure C-11. System 47 - Oxygen Supply

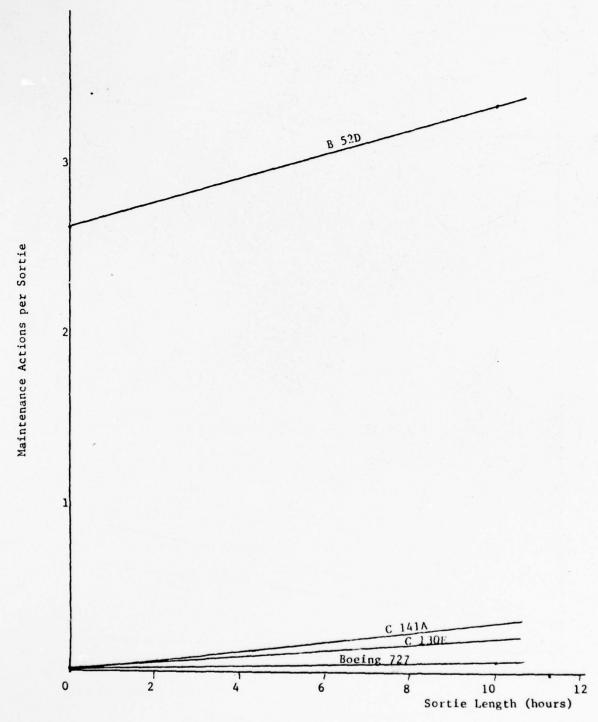


Figure C-12. System 51 - Instruments

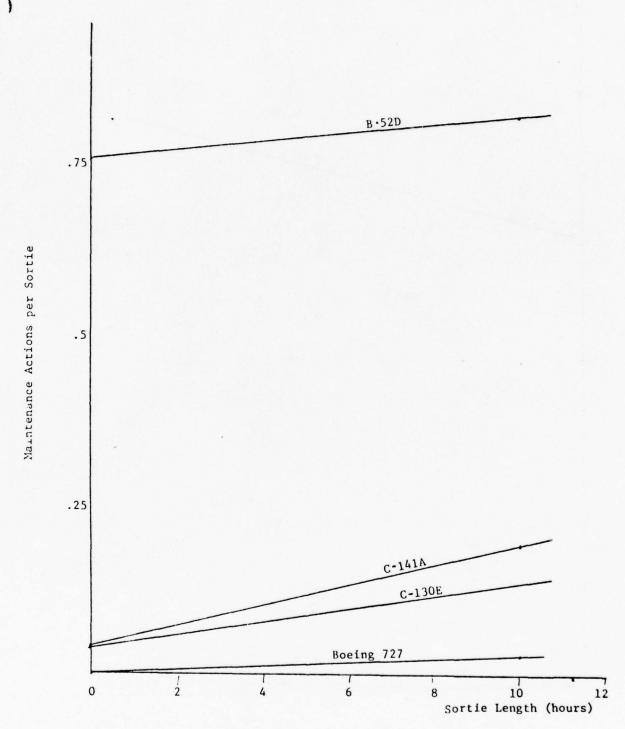


Figure C-13. System 52 - Autopilot

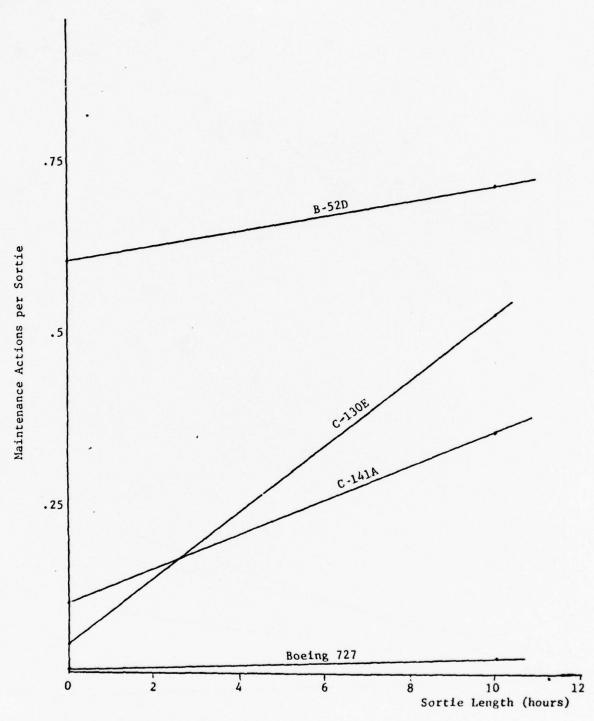
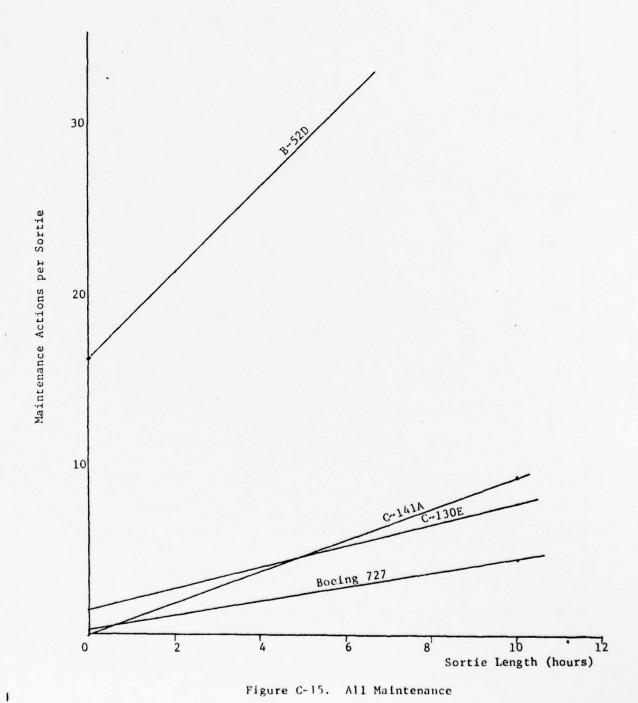


Figure C-14. System 61/64 - Communications



## APPENDIX D

C-130E and C-141A Maintenance Data and Flight Statistics

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SUNDAY,	HRS	0.692	443.9	465.3	538.2	246.5	747.7	453.3	461.8	296.9	508.9	660.1	492.7	701.5	9./49	510.2	793.6	639.1	754.5	100.8	462.6	416.5	298.7	733.6	386.8	390.8	NDAY,	HRS		611.4	521.4	604.7	628.9	534.7	638.2	0.610	434.0	390.1	662.8	550.1	475.1	420.0	2.064	631.7	623.2	742.1	544.8	9.019	0.649	4.869	604.6	291.0
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PLANE	7893	1894	7895	9682	7879	8684	7899	9810	9811	9812	9813	9814	9815	9816	9817
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